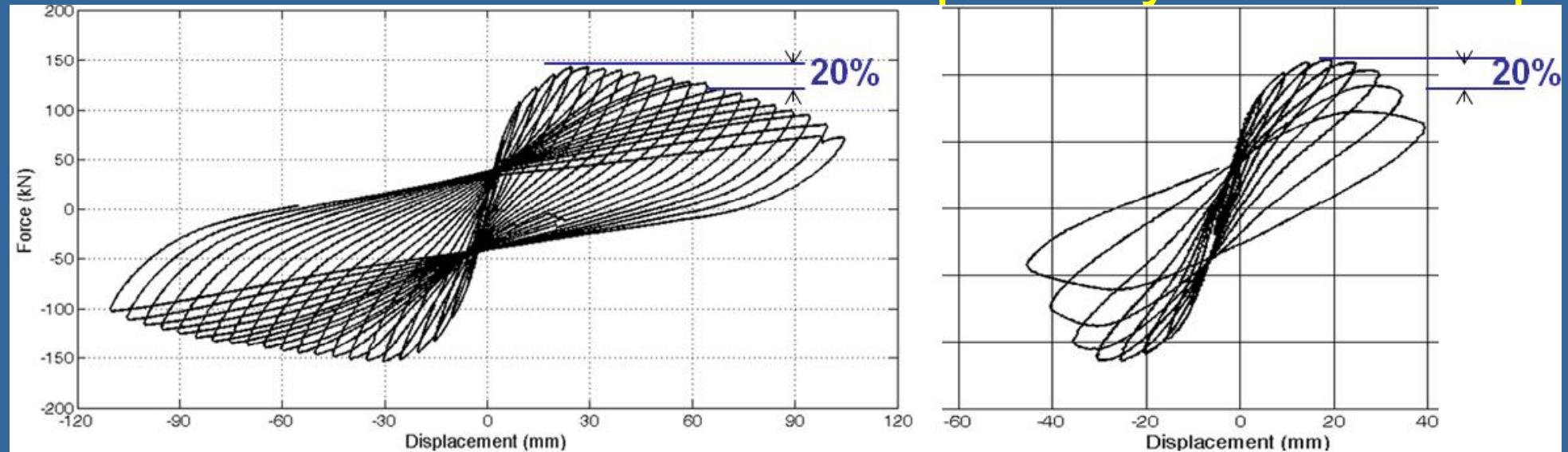


MODELLING THE BEHAVIOR OF CONCRETE MEMBERS: DEVELOPMENTS SINCE THE COMPLETION OF EN 1998-3:2005

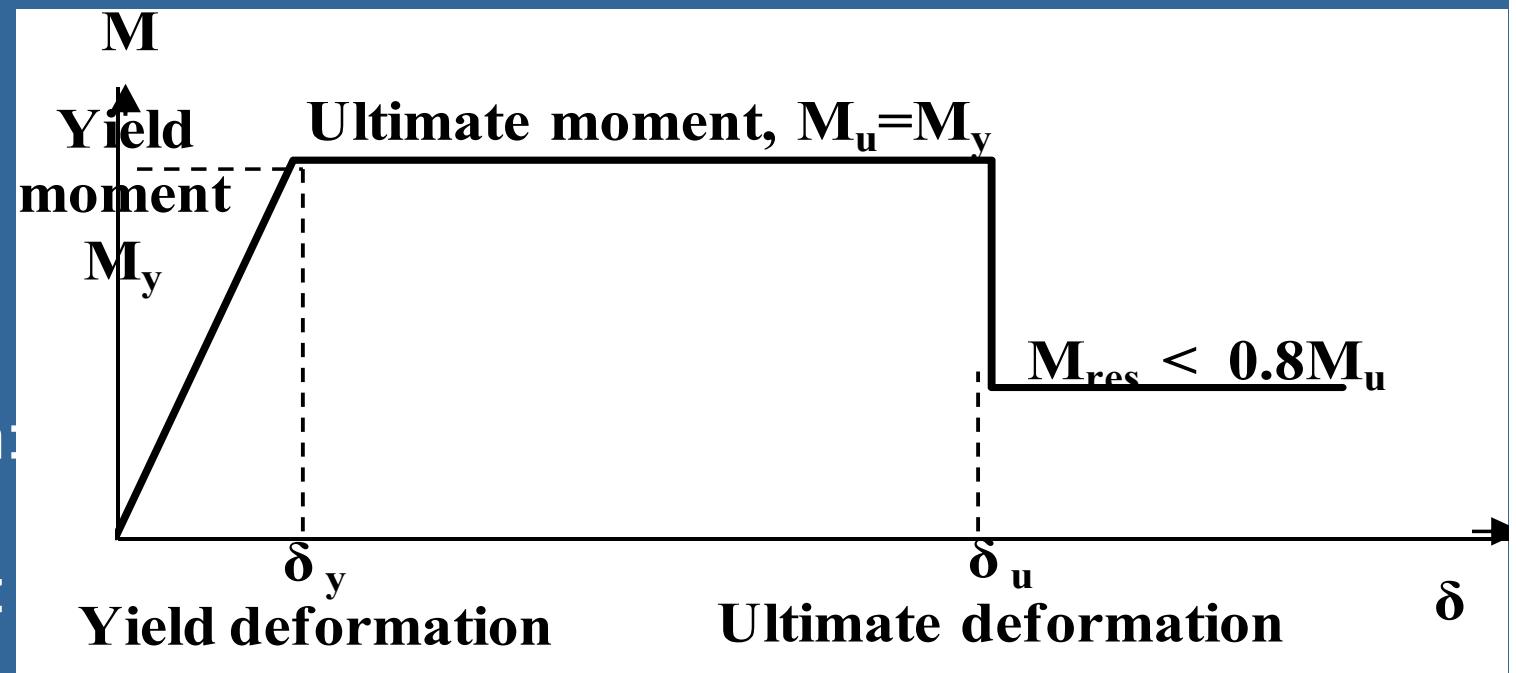
**Michael N. Fardis
University of Patras**

Idealized skeleton curve - envelope to hysteresis loops



Effective elastic
stiffness:
secant-to-yielding

member deformation:
chord-rotation
section deformation:
curvature



**Practical expressions
for the yield & failure properties
developed for EN 1998-3:2005
- their advancement since 2005**

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Experimental Database

1. Range/mean of parameters in tests for calibration of expressions for the member chord rotation and secant stiffness at yielding

Parameter	1653 rectangular beams/columns		214 rectangular walls		229 members of non-rectangular section		307 circular columns	
	min/max	mean	min/max	mean	min/max	mean	min/max	mean
section depth or diameter, h (m)	0.1 / 2.4	0.31	0.4 / 3.0	1.19	0.2 / 3.96	1.4	0.15 / 1.83	0.43
shear-span-to-depth ratio, L_s/h	1 / 13.3	3.78	0.45 / 5.53	1.92	0.45 / 8.33	2.03	1.1 / 10	3.77
section aspect ratio, h/b_w	0.2 / 4	1.3	4 / 30	10.9	2.5 / 57	16.7	1.0	1.0
f_c (MPa)	9.6 / 175	37.7	13.5 / 109	35.8	13.5 / 101.8	40.4	16.7 / 90	36.7
axial-load-ratio, N/Af_c	-0.05 / 0.9	0.126	0 / 0.86	0.10	0 / 0.50	0.06	-0.1 / 0.7	0.137
transverse steel ratio, ρ_w (%)	0 / 3.54	0.62	0.05 / 2.18	0.54	0.04 / 2.44	0.7	0 / 8.83	0.86
total longitudinal steel ratio ρ_{tot} (%)	0.2 / 8.55	1.97	0.07 / 4.27	1.5	0.205 / 6.2	1.23	0.53 / 5.8	2.34
diagonal steel ratio, ρ_d (%)	0 / 1.68	0.027	0 / 0.25	0.005	-	-	-	-
transverse steel yield stress f_{yw} , MPa	118 / 2050	468	220 / 1375	443	160 / 1375	504	200 / 1728	454.2
longitudinal steel yield stress f_y , MPa	247 / 1200	440.2	276 / 1273	470	209 / 900	453	240 / 648	414.3

Experimental Database

2. Range/mean of parameters in tests for calibration of expressions for ultimate curvature in monotonic or cyclic loading

Parameter	415 rectangular beams/columns					59 rectangular walls			
	254 monotonic		160 cyclic		13 monotonic		46 cyclic		
	min/max	mean	min/max	mean	min/max	mean	min/max	mean	
section depth or diameter, h (m)	0.12 / 0.8	0.31	0.22 / 2.4	0.41	2.39 / 2.41	2.4	1.7 / 2.0	1.99	
section aspect ratio, h/b_w	0.225 / 3.73	1.54	0.5 / 2	1.21	21.2 / 23.4	22.4	13.3 / 28.3	13.7	
f_c (MPa)	19.7 / 99.4	34.8	17.7 / 102.2	38.6	34.5 / 40.8	35.3	26.2 / 45.6	40.6	
axial-load-ratio, N/Af_c	0 / 0.78	0.08	0 / 0.77	0.236	0.063 / 0.077	0.07	0.05 / 0.11	0.07	
transverse steel ratio, ρ_w (%)	0 / 2.38	0.345	0.04 / 2.96	0.656	0.41 / 0.82	0.66	0.11 / 0.25	0.24	
total longitudinal steel ratio ρ_{tot} (%)	0 / 3.68	1.4	0.37 / 4.19	1.82	0.88 / 1.76	1.35	0.07 / 0.77	0.63	
transverse steel yield stress f_{yw} , MPa	0 / 596	419	255 / 1402	477.2	440 / 483	443.3	465 / 562	502	
longitudinal steel yield stress f_y , MPa	277 / 596	490	341 / 573	493.3	444 / 510	466.6	523 / 580	552	

Experimental Database

3. Range/mean of parameters in tests for calibration of expressions for the member ultimate cyclic chord rotation

Parameter	1159 rectangular beams/columns		95 rectangular walls		53 members of non-rectangular section		143 circular columns	
	min/max	mean	min/max	mean	min/max	mean	min/max	mean
section depth or diameter, h (m)	0.1 / 2.4	0.33	0.4 / 2.75	1.15	0.2 / 3.4	1.21	0.2 / 1.83	0.45
shear-span-to-depth ratio, L_s/h	1 / 13.3	3.7	0.5 / 5.53	2.15	0.65 / 8.33	2.85	1.77 / 10	4.22
section aspect ratio, h/b_w	0.2 / 6	1.18	2.5 / 28.3	9.75	2.5 / 36	9.95	1.0	1.0
f_c (MPa)	12.2 / 175	43.7	13.5 / 109	35.9	20.8 / 83.6	38.5	23.1 / 90	38.0
axial-load-ratio, N/A_cf_c	-0.1 / 0.9	0.165	0 / 0.86	0.116	0 / 0.30	0.07	-0.09 / 0.7	0.15
transverse steel ratio, ρ_w (%)	0.015 / 3.37	0.82	0.05 / 2.18	0.63	0.04 / 2.09	0.59	0.1 / 8.83	0.94
total longitudinal steel ratio ρ_{tot} (%)	0 / 6.29	2.08	0.07 / 4.27	1.37	0.2 / 6.19	1.32	0.75 / 5.5	2.05
diagonal steel ratio, ρ_d (%)	0 / 1.68	0.028	0 / 0.25	0.004	-	-	-	-
transverse steel yield stress f_{yw} , MPa	118 / 1497	497	220 / 1375	435.3	178 / 1375	513	200 / 1569	482.5
longitudinal steel yield stress f_y , MPa	281 / 1275	467.5	276 / 1273	471.2	331 / 596	438.7	240 / 648	428

Yield & failure properties of RC sections

$M-\phi$ at yielding of section w/ rectangular compression zone (width b , effective depth d) – section analysis

- Yield moment (from moment-equilibrium & elastic σ - ϵ laws):

$$\frac{M_y}{bd^3} = \varphi_y \left\{ E_c \frac{\xi_y^2}{2} \left(0.5(1 + \delta_1) - \frac{\xi_y}{3} \right) + \frac{E_s}{2} \left[(1 - \xi_y)\rho_1 + (\xi_y - \delta_1)\rho_2 + \frac{\rho_v}{6}(1 - \delta_1) \right] (1 - \delta_1) \right\}$$

- ρ_1, ρ_2 : tension & compression reinforcement ratios, ρ_v : “web” reinforcement ratio, ~uniformly distributed between ρ_1, ρ_2 : (all normalized to bd); $\delta_1 = d_1/d$.

- Curvature at yielding of tension steel:

- from axial force-equilibrium & elastic σ - ϵ laws ($\alpha = E_s/E_c$):

$$\varphi_y = \frac{f_y}{E_s(1 - \xi_y)d}$$

$$\xi_y = (\alpha^2 A^2 + 2\alpha B)^{1/2} - \alpha A$$

$$A = \rho_1 + \rho_2 + \rho_v + \frac{N}{bdf_y}, \quad B = \rho_1 + \rho_2\delta_1 + 0.5\rho_v(1 + \delta_1) + \frac{N}{bdf_y}$$

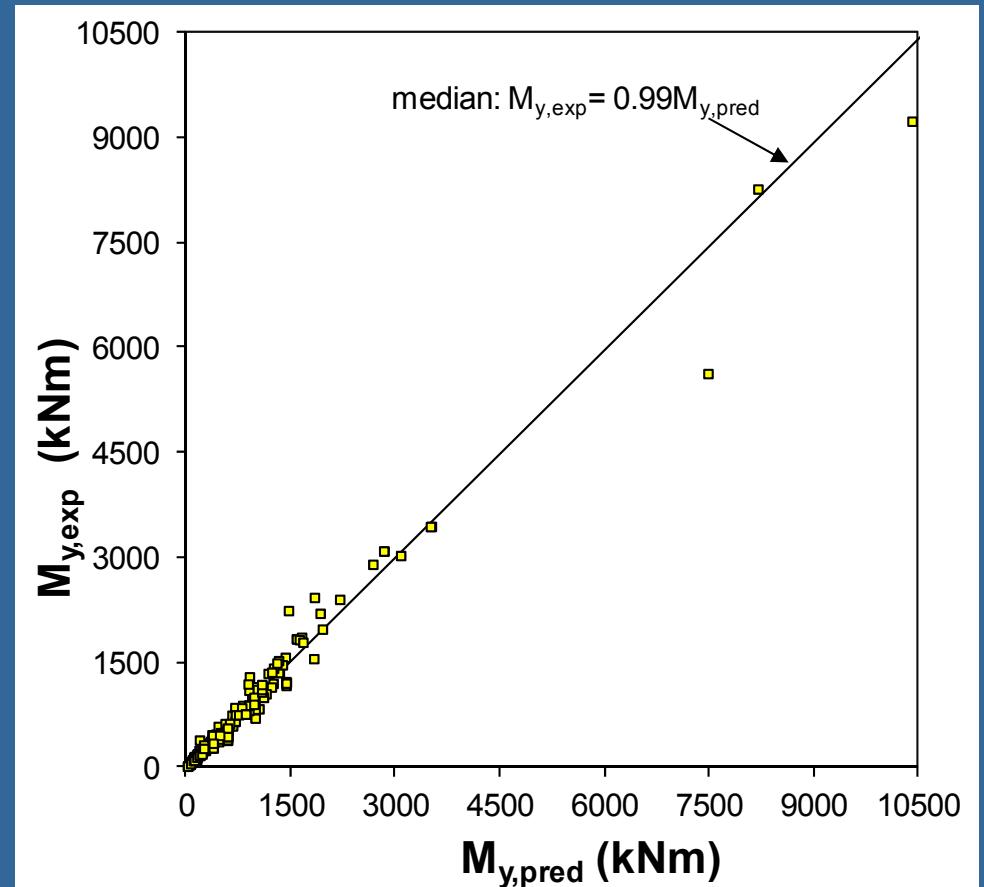
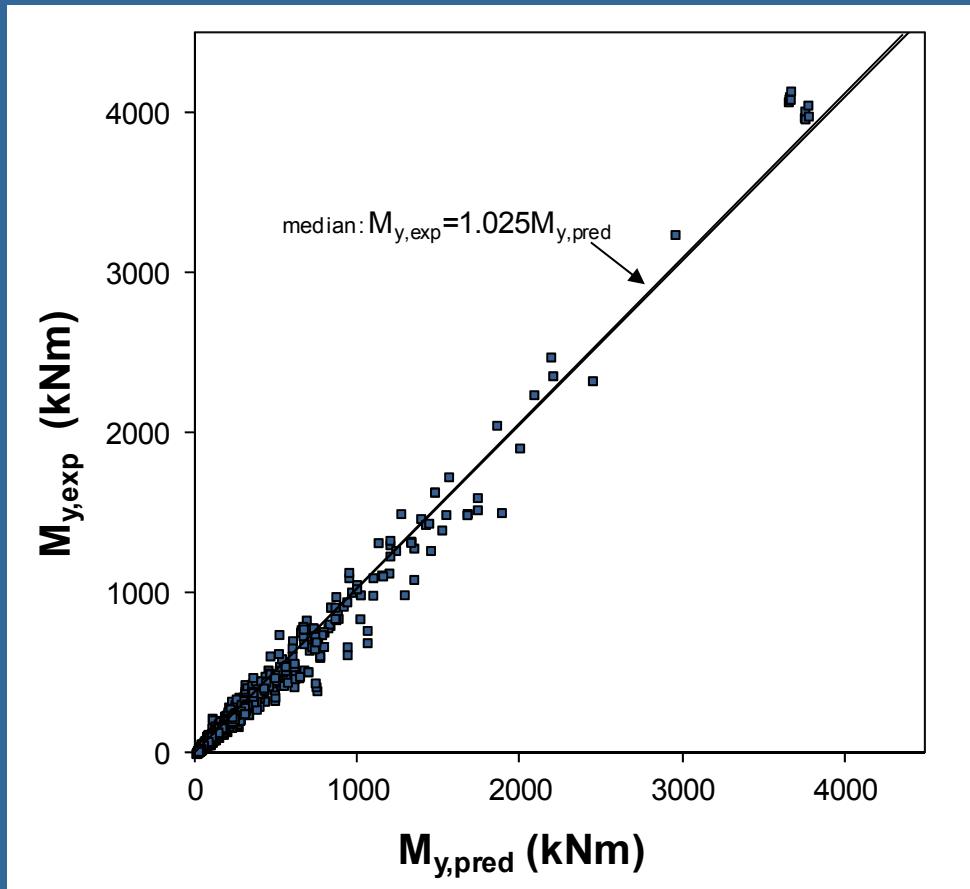
- Curvature at ~onset of nonlinearity of concrete:

$$\varphi_y = \frac{\varepsilon_c}{\xi_y d} \approx \frac{1.8 f_c}{E_c \xi_y d}$$

$$A = \rho_1 + \rho_2 + \rho_v - \frac{N}{\varepsilon_c E_s bd} \approx \rho_1 + \rho_2 + \rho_v - \frac{N}{1.8 \alpha bdf_c}, \quad B = \rho_1 + \rho_2\delta_1 + 0.5\rho_v(1 + \delta_1)$$

Moment at corner of bilinear envelope to experimental moment-deformation curve vs yield moment from section analysis

Left: 2085 beams/columns, CoV:16.3%; Right: 224 rect. walls, CoV:16.9%



Bias by +2.5% or -1%, because corner of bilinear envelope of the experimental moment-deformation curve \neq 1st yielding in section. Same bias considered to apply to predicted yield curvature.

Empirical formulas for yield curvature - section w/ rectangular compression zone

for beams or columns:

$$\varphi_y \approx \frac{1.54 f_y}{E_s d} \quad \varphi_y \approx \frac{1.75 f_y}{E_s h}$$

for walls:

$$\varphi_y \approx \frac{1.37 f_y}{E_s d} \quad \varphi_y \approx \frac{1.47 f_y}{E_s h}$$

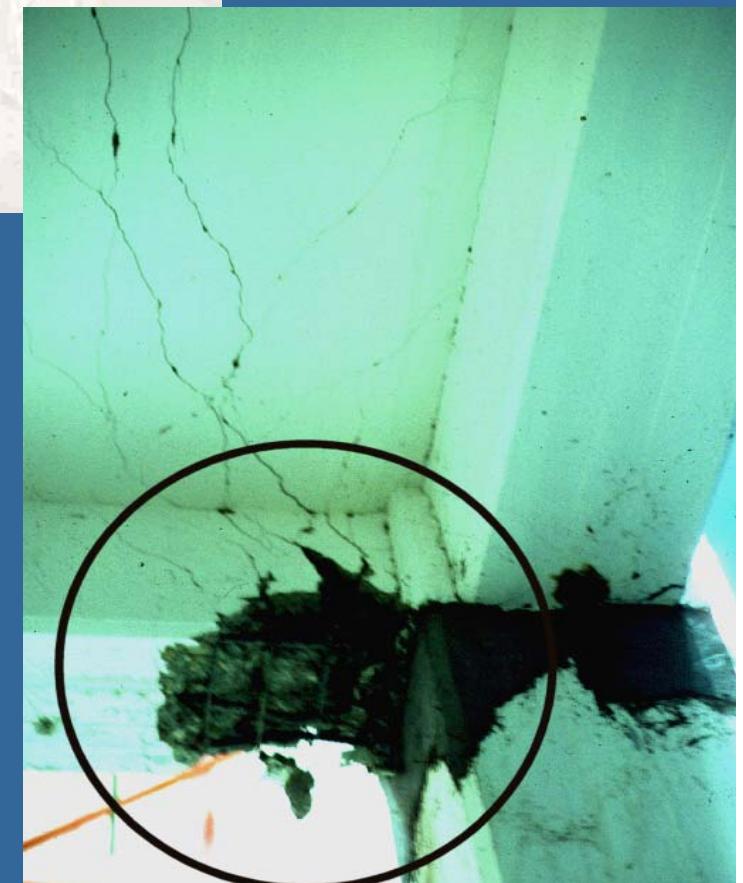
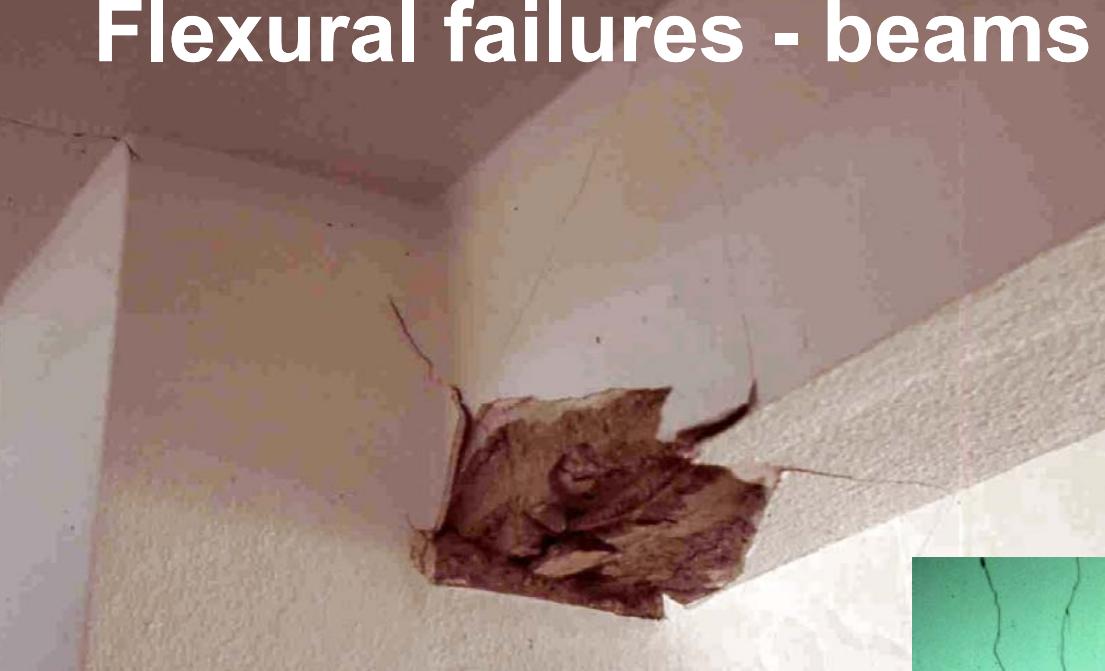
Empirical expressions don't have a bias w.r.to experimental yield moment; but scatter is larger:

In ~2100 test beams, columns or walls: CoV: ~18%

Flexural failures - columns

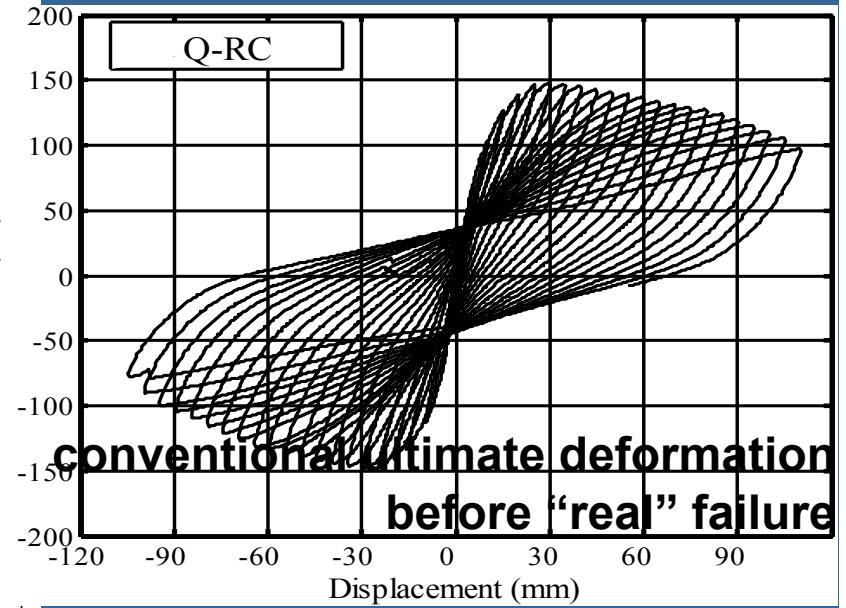
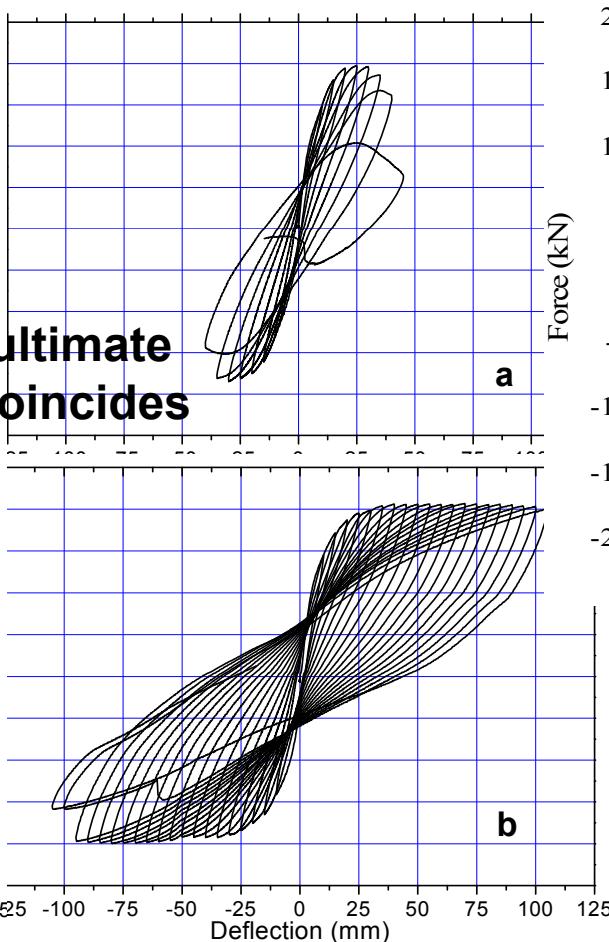
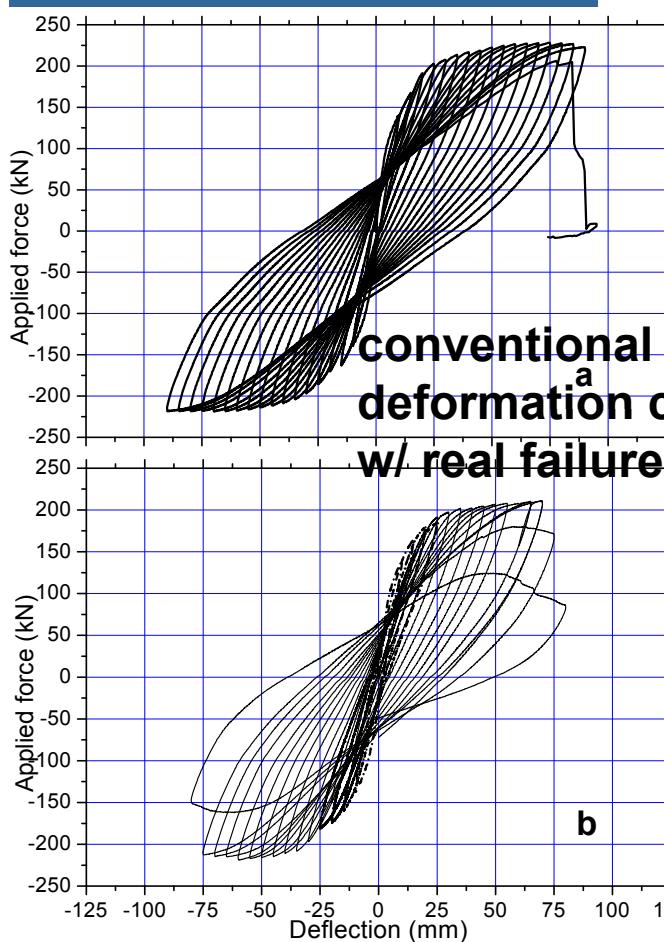


Flexural failures - beams



Conventional definition of ultimate deformation

The value beyond which, any increase in deformation cannot increase the resistance above 80% of the maximum previous (ultimate) resistance.



Ultimate curvature of section with rectangular compression zone, from section analysis

- Concrete σ - ϵ law:
 - Parabolic up to f_c , ϵ_{co} ,
 - constant stress (rectangular) for $\epsilon_{co} < \epsilon < \epsilon_{cu}$
- Steel σ - ϵ law:
 - Elastic-perfectly plastic, if steel strain rather low and concrete fails first;
 - Elastic-linearly strain-hardening, if steel strains are relatively high and steel breaks at stress and strain f_t , ϵ_{su} .

Possibilities for ultimate curvature:

1. Section fails by rupture of tension steel, $\varepsilon_{s1} = \varepsilon_{su}$, before extreme compression fibers reach their ultimate strain (spalling), $\varepsilon_c < \varepsilon_{cu} \rightarrow$
Ultimate curvature occurs in unspalled section, due to steel rupture: $\varphi_{su} = \frac{\varepsilon_{su}}{(1 - \xi_{su})d}$
2. Compression fibres reach their ultimate strain (spalling): $\varepsilon_c = \varepsilon_{cu} \rightarrow$
the confined concrete core becomes now the member section.
Two possibilities:
 - i. The moment capacity of the spalled section, M_{Ro} , never increases above 80% of the moment at spalling, M_{Rc} : $M_{Ro} < 0.8M_{Rc} \rightarrow$
Ultimate curvature occurs in unspalled section, due to the concrete: $\varphi_{cu} = \frac{\varepsilon_{cu}}{\xi_{cu}d}$
 - ii. Moment capacity of spalled section increases above 80% of the moment at spalling: $M_{Ro} > 0.8M_{Rc} \rightarrow$

The confined concrete core is now the member section and Cases 1 and 2(i) - applied for the confined core - are the two possibilities for attainment of the ultimate curvature $\rightarrow \varphi_{su}, \varphi_{cu}$ calculated as above but for the confined core; the minimum of the two is the ultimate curvature.

$$M_{Rc} = bd^2 f_c \left\{ \frac{(1 - \delta_1)(\omega_1 + \omega_2)}{2} + \frac{\omega_v}{1 - \delta_1} \left[(\xi - \delta_1)(1 - \xi) - \frac{1}{3} \left(\frac{\xi f_y}{E_s \varepsilon_{cu}} \right)^2 \right] + \xi \left[\frac{1 - \xi}{2} - \frac{\varepsilon_{co}}{3\varepsilon_{cu}} \left(\frac{1}{2} - \xi + \frac{\varepsilon_{co}}{4\varepsilon_{cu}} \xi \right) \right] \right\}$$

Ultimate curvature of section w/ rectangular compression zone for steel rupture:

$$\varphi_{su} = \frac{\varepsilon_{su}}{(1 - \xi_{su})d}$$

- Steel ruptures before concrete crushes, after compression steel yields, if v :

$$\frac{\delta_1 \varepsilon_{su} + \varepsilon_y - (1 - \delta_1) \frac{\varepsilon_{co}}{3}}{\varepsilon_{su} + \varepsilon_y} + \omega_2 - \omega_1 \frac{f_t}{f_y} - \omega_v \left(1 + \frac{f_t}{f_y} \right) \frac{\varepsilon_{su} - \varepsilon_y}{2(\varepsilon_{su} + \varepsilon_y)} \leq v \leq \frac{\varepsilon_{cu} - \frac{\varepsilon_{co}}{3}}{\varepsilon_{cu} + \varepsilon_{su}} + \omega_2 - \omega_1 \frac{f_t}{f_y} - \omega_v \left(1 + \frac{f_t}{f_y} \right) \frac{\varepsilon_{su}(1 + \delta_1) - \varepsilon_{cu}(1 - \delta_1)}{(1 - \delta_1)(\varepsilon_{su} + \varepsilon_{cu})}$$

- ξ_{su} calculated from axial force equilibrium for:

$$\xi_{su} = \frac{(1 - \delta_1) \left(v + \omega_1 \frac{f_t}{f_y} - \omega_2 + \frac{\varepsilon_{co}}{3\varepsilon_{su}} \right) + \left(\frac{1 + \delta_1}{2} \right) \left(1 + \frac{f_t}{f_y} \right) \omega_v}{(1 - \delta_1) \left(1 + \frac{\varepsilon_{co}}{3\varepsilon_{su}} \right) + \left(1 + \frac{f_t}{f_y} \right) \omega_v}$$

$v = N/bdf_c$, ω_1 , ω_2 : tension & compression mech. reinforcement ratios, ω_v : "web" mech. reinforcement ratio ~uniform distribution between ω_1 , ω_2 ($\omega = \rho f_y/f_c$); $\delta_1 = d_1/d$.

- Steel ruptures before concrete crushes or compression steel yields, if v :

$$v \leq \frac{\delta_1 \varepsilon_{su} + \varepsilon_y - (1 - \delta_1) \frac{\varepsilon_{co}}{3}}{\varepsilon_{su} + \varepsilon_y} + \omega_2 - \omega_1 \frac{f_t}{f_y} - \omega_v \left(1 + \frac{f_t}{f_y} \right) \frac{\varepsilon_{su} - \varepsilon_y}{2(\varepsilon_{su} + \varepsilon_y)}$$

- ξ_{su} from: $\left[1 + \frac{\varepsilon_{co}}{3\varepsilon_{su}} + \frac{\omega_v}{2(1 - \delta_1)} \left(1 + \frac{f_t}{f_y} - \frac{f_t}{E_s \varepsilon_{su}} - \frac{\varepsilon_{su}}{\varepsilon_y} \right) \right] \xi^2 -$

$$\left[1 + v + \frac{2\varepsilon_{co}}{3\varepsilon_{su}} + \omega_1 \frac{f_t}{f_y} + \omega_2 \frac{\varepsilon_{su}}{\varepsilon_y} + \frac{\omega_v}{(1 - \delta_1)} \left(1 + \frac{f_t}{f_y} - \frac{f_t}{E_s \varepsilon_{su}} - \delta_1 \frac{\varepsilon_{su}}{\varepsilon_y} \right) \right] \xi + \left[v + \frac{\varepsilon_{co}}{3\varepsilon_{su}} + \omega_1 \frac{f_t}{f_y} + \omega_2 \delta_1 \frac{\varepsilon_{su}}{\varepsilon_y} + \frac{\omega_v}{2(1 - \delta_1)} \left(1 + \frac{f_t}{f_y} - \frac{f_t}{E_s \varepsilon_{su}} - \delta_1^2 \frac{\varepsilon_{su}}{\varepsilon_y} \right) \right] = 0$$

Ultimate curvature of section w/ rectangular compression zone for concrete crushing: $\varphi_{cu} = \frac{\varepsilon_{cu}}{\xi_{cu} d}$

- Concrete crushes after tension steel yields, w/o compression steel yielding, if v :

$$v \leq \omega_2 - \omega_1 + \frac{\omega_v}{1 - \delta_1} \left(\delta_1 \frac{\varepsilon_{cu} + \varepsilon_y}{\varepsilon_{cu} - \varepsilon_y} - 1 \right) + \delta_1 \frac{\varepsilon_{cu} - \frac{\varepsilon_{co}}{3}}{\varepsilon_{cu} - \varepsilon_y}$$

- ξ_{cu} from axial force equilibrium, for:

$$\left[1 - \frac{\varepsilon_{co}}{3\varepsilon_{cu}} + \frac{\omega_v}{2(1 - \delta_1)} \frac{(\varepsilon_{cu} + \varepsilon_y)^2}{\varepsilon_{cu} \varepsilon_y} \right] \xi^2 - \left[v + \omega_1 - \omega_2 \frac{\varepsilon_{cu}}{\varepsilon_y} + \frac{\omega_v}{1 - \delta_1} \left(1 + \frac{\varepsilon_{cu} \delta_1}{\varepsilon_y} \right) \right] \xi - \left[\omega_2 - \frac{\omega_v \delta_1}{2(1 - \delta_1)} \right] \frac{\varepsilon_{cu} \delta_1}{\varepsilon_y} = 0$$

- Concrete crushes w/ tension & compression steel yielding, if v :

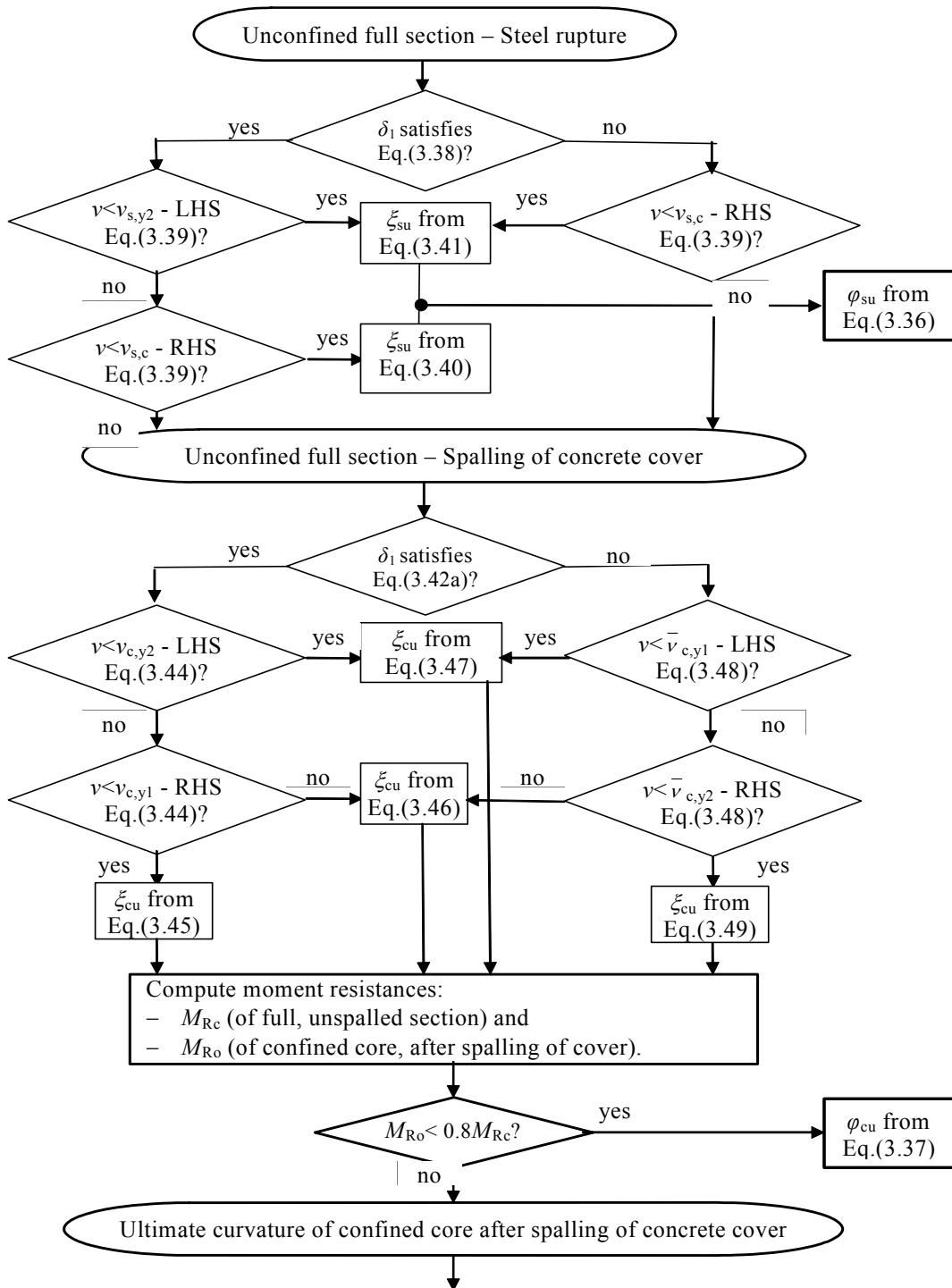
$$\omega_2 - \omega_1 + \frac{\omega_v}{1 - \delta_1} \left(\delta_1 \frac{\varepsilon_{cu} + \varepsilon_y}{\varepsilon_{cu} - \varepsilon_y} - 1 \right) + \delta_1 \frac{\varepsilon_{cu} - \frac{\varepsilon_{co}}{3}}{\varepsilon_{cu} - \varepsilon_y} \leq v < \omega_2 - \omega_1 + \frac{\omega_v}{1 - \delta_1} \left(\frac{\varepsilon_{cu} - \varepsilon_y}{\varepsilon_{cu} + \varepsilon_y} - \delta_1 \right) + \frac{\varepsilon_{cu} - \frac{\varepsilon_{co}}{3}}{\varepsilon_{cu} + \varepsilon_y}$$

- ξ_{cu} from:
$$\xi_{cu} = \frac{(1 - \delta_1)(v + \omega_1 - \omega_2) + (1 + \delta_1)\omega_v}{(1 - \delta_1) \left(1 - \frac{\varepsilon_{co}}{3\varepsilon_{cu}} \right) + 2\omega_v}$$

- Concrete crushes after compression steel yields, w/o tension steel yielding, if v :

- ξ_{cu} from:
$$\omega_2 - \omega_1 + \frac{\omega_v}{1 - \delta_1} \left(\frac{\varepsilon_{cu} - \varepsilon_y}{\varepsilon_{cu} + \varepsilon_y} - \delta_1 \right) + \frac{\varepsilon_{cu} - \frac{\varepsilon_{co}}{3}}{\varepsilon_{cu} + \varepsilon_y} \leq v$$

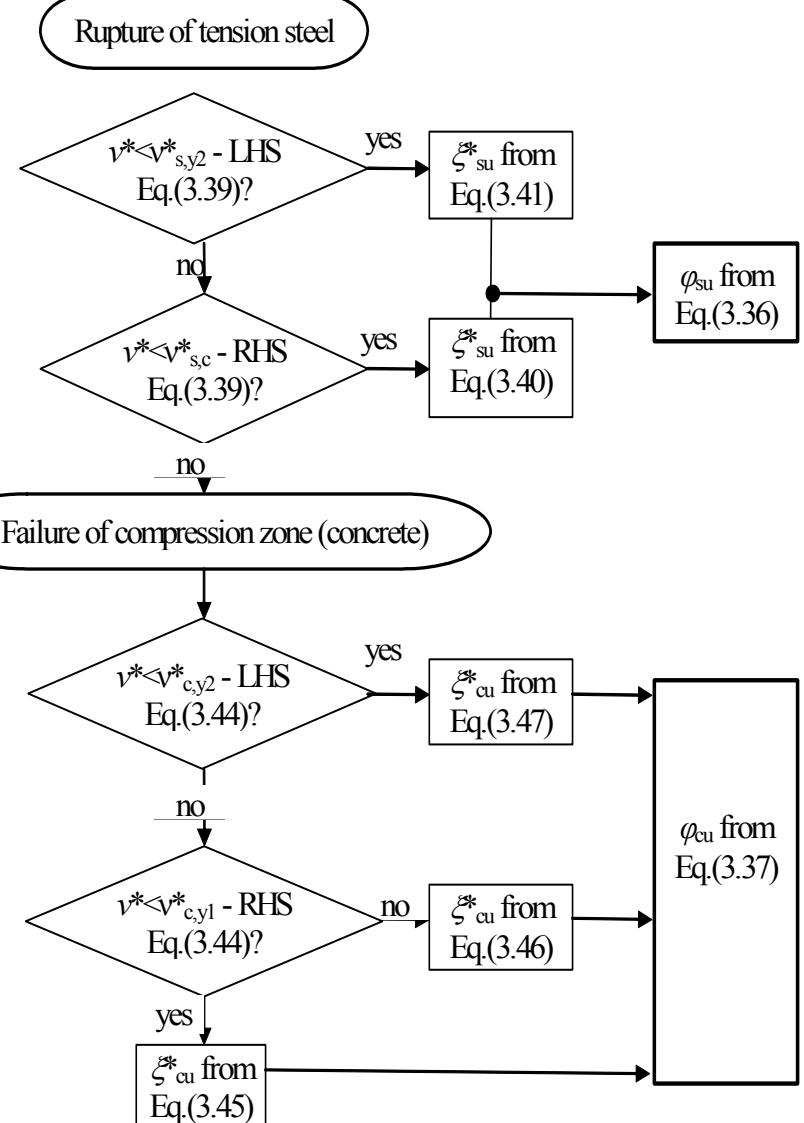
$$\left[1 - \frac{\varepsilon_{co}}{3\varepsilon_{cu}} - \frac{\omega_v}{2(1 - \delta_1)} \frac{(\varepsilon_{cu} - \varepsilon_y)^2}{\varepsilon_{cu} \varepsilon_y} \right] \xi^2 + \left[\omega_2 + \omega_1 \frac{\varepsilon_{cu}}{\varepsilon_y} - v + \frac{\omega_v}{1 - \delta_1} \left(\frac{\varepsilon_{cu}}{\varepsilon_y} - \delta_1 \right) \right] \xi - \left[\omega_1 + \frac{\omega_v}{2(1 - \delta_1)} \right] \frac{\varepsilon_{cu}}{\varepsilon_y} = 0$$



Confined core after spalling of concrete cover.

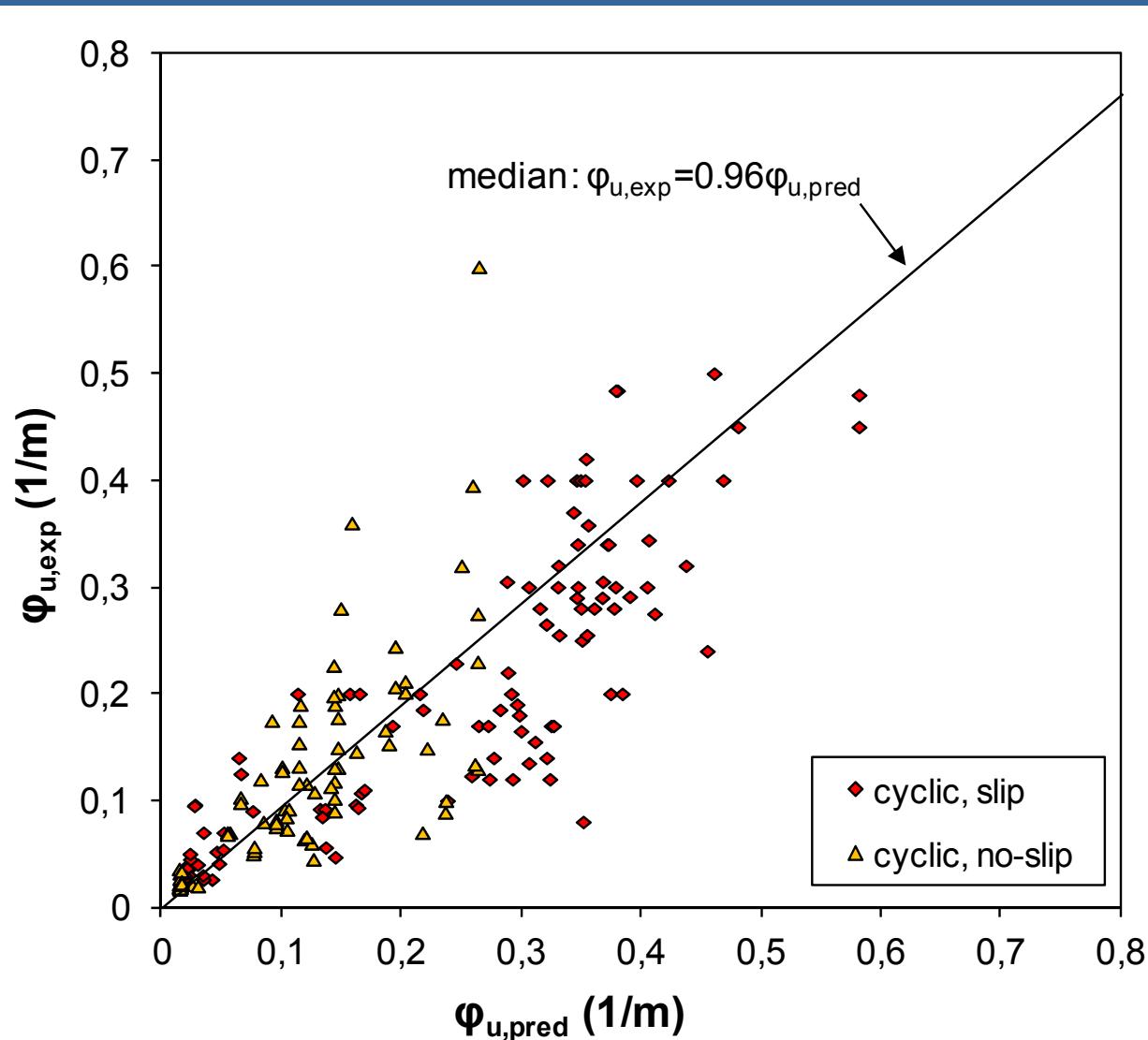
Parameters are denoted by an asterisk and computed with:

- b, d, d_1 replaced by geometric parameters of the core: b_c, d_c, d_{c1} ;
- N, ρ, ρ_2, ρ normalized to $b_c d_c$, instead of $b d$;
- $\sigma\text{-}\epsilon$ parameters of confined concrete, f_{cc}, ϵ_{cc} , used in lieu of f_c, ϵ_{cu}



Test results vs ultimate curvature w/ failure strains for cyclic flexure per EN 1998-3:2005 :

Steel: ε_{su} : 2.5%, 5%, 6% for steel class A, B, C per Eurocode 2



$$\varepsilon_{cu,c} = 0.004 + 0.5 \left(\alpha \rho_s f_{yw} / f_{cc} \right)$$

cyclic test results

Test results vs ultimate curvature w/ failure strains for cyclic flexure per Biskinis & Fardis 2010 (adopted in fib MC2010):

- **Monotonic flexure:**

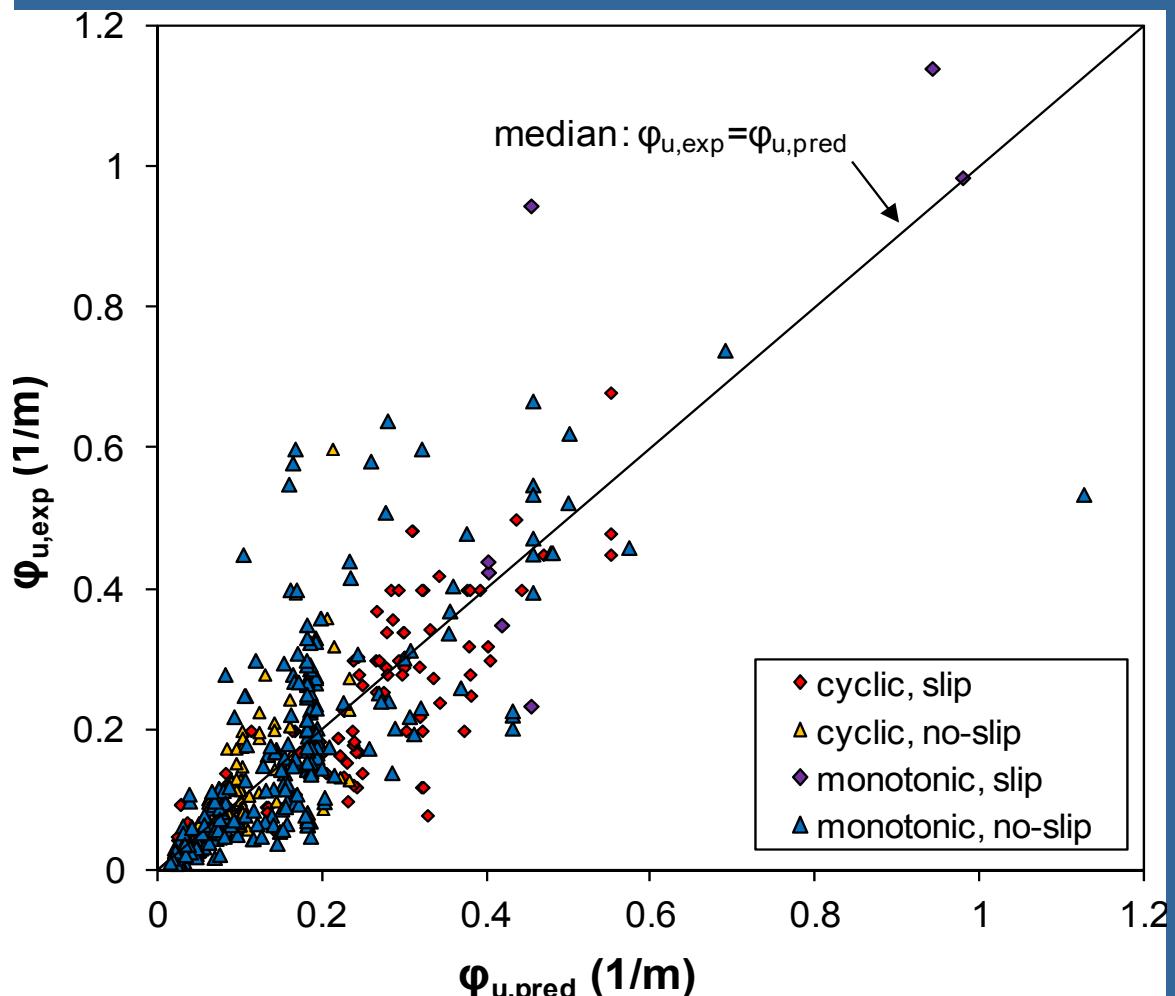
- max. available steel strain: $(7/12)\varepsilon_{su}$

- **Cyclic flexure:**

- max. available steel strain: $(3/8)\varepsilon_{su}$

$$\varepsilon_{cu}^* = 0.0035 + \left(\frac{10}{h_c(\text{mm})} \right)^2 + 0.285 \frac{\alpha \omega_w}{1+K}$$

$$\varepsilon_{cu}^* = 0.0035 + \left(\frac{10}{h_c(\text{mm})} \right)^2 + 0.2 \frac{\alpha \omega_w}{1+K}$$



monotonic & cyclic data, no. 474, median=1.00, CoV=49.7%

cyclic test data:

	Cyclic all	Concrete crushing	Steel rapture
Median	0.99	0.99	1.01
C.o.V.	44.2%	52.6%	34.2%
No.	205	97	108

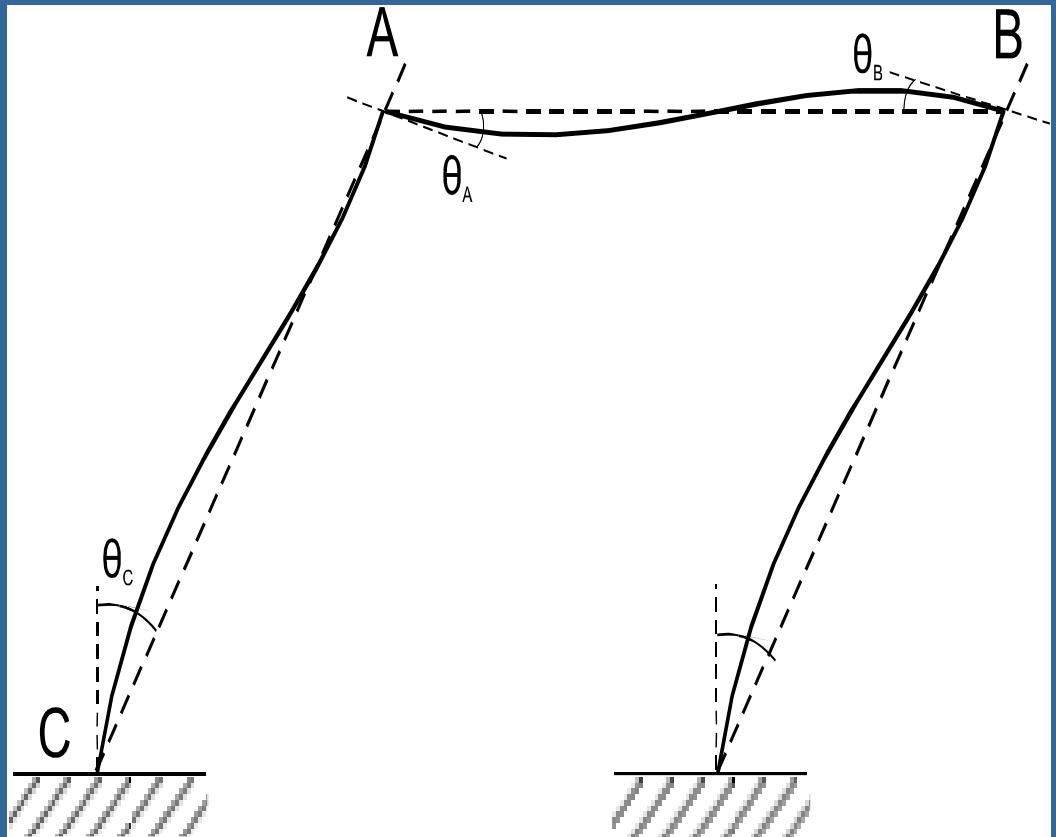
Yield & failure properties of RC members

Flexural behavior at member level (Moment-chord rotation)

Definition of chord rotations, θ , at member ends

$$\theta_A = \frac{1}{x_B - x_A} \int_{x_A}^{x_B} \varphi(x)(x_B - x) dx$$

$$\theta_B = \frac{1}{x_B - x_A} \int_{x_A}^{x_B} \varphi(x)(x_A - x) dx$$



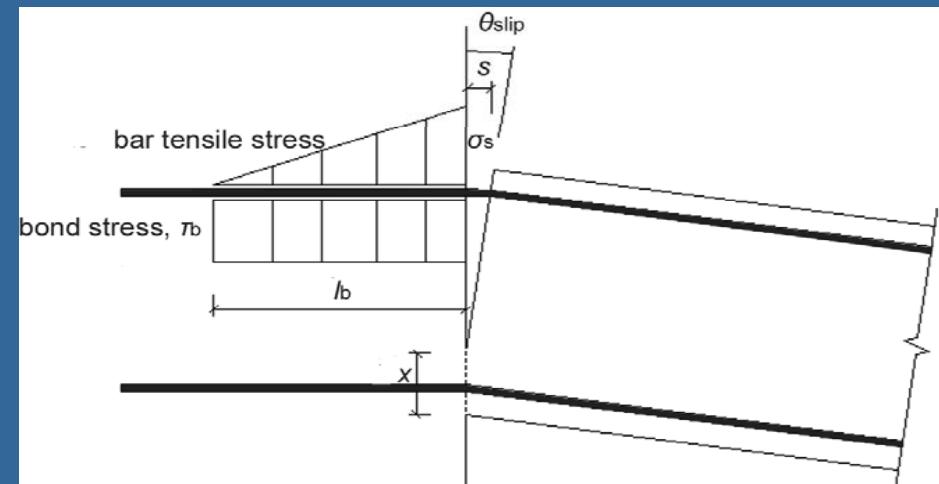
Elastic moments at ends A, B from chord rotations at A, B:

- $M_A = (2EI/L)(2\theta_A + \theta_B)$,
- $M_B = (2EI/L)(2\theta_B + \theta_A)$

Fixed-end rotation of member end due to bar slippage from their anchorage zone beyond member end

- Slippage of tension bars from region beyond end section (e.g. from joint or footing) → rigid-body rotation of entire shear span = fixed-end rotation, θ_{slip}
(included in measured chord-rotations of test specimen w.r. to base or joint; doesn't affect measured relative rotations between any two member sections).
- If s = slippage of tension bars from anchorage → $\theta_{\text{slip}} = s/(1-\xi)d$
- If bond stress uniform over straight length l_b of bar beyond section of maximum moment → bar stress decreases along l_b from σ_s ($=f_{yL}$ at yielding) at section of maximum M to zero at end of l_b → $s=\sigma_s l_b / (2E_s)$
- l_b = bond force demand per unit length ($=A_s \sigma_s / (\pi d_{bL}) = d_{bL} \sigma_s / 4$), divided by ~bond strength (assume $=\sqrt{f_c}$)
- $\varepsilon_s (= \sigma_s / E_s) / (1 - \xi) d = \varphi$
- At yielding of member end section

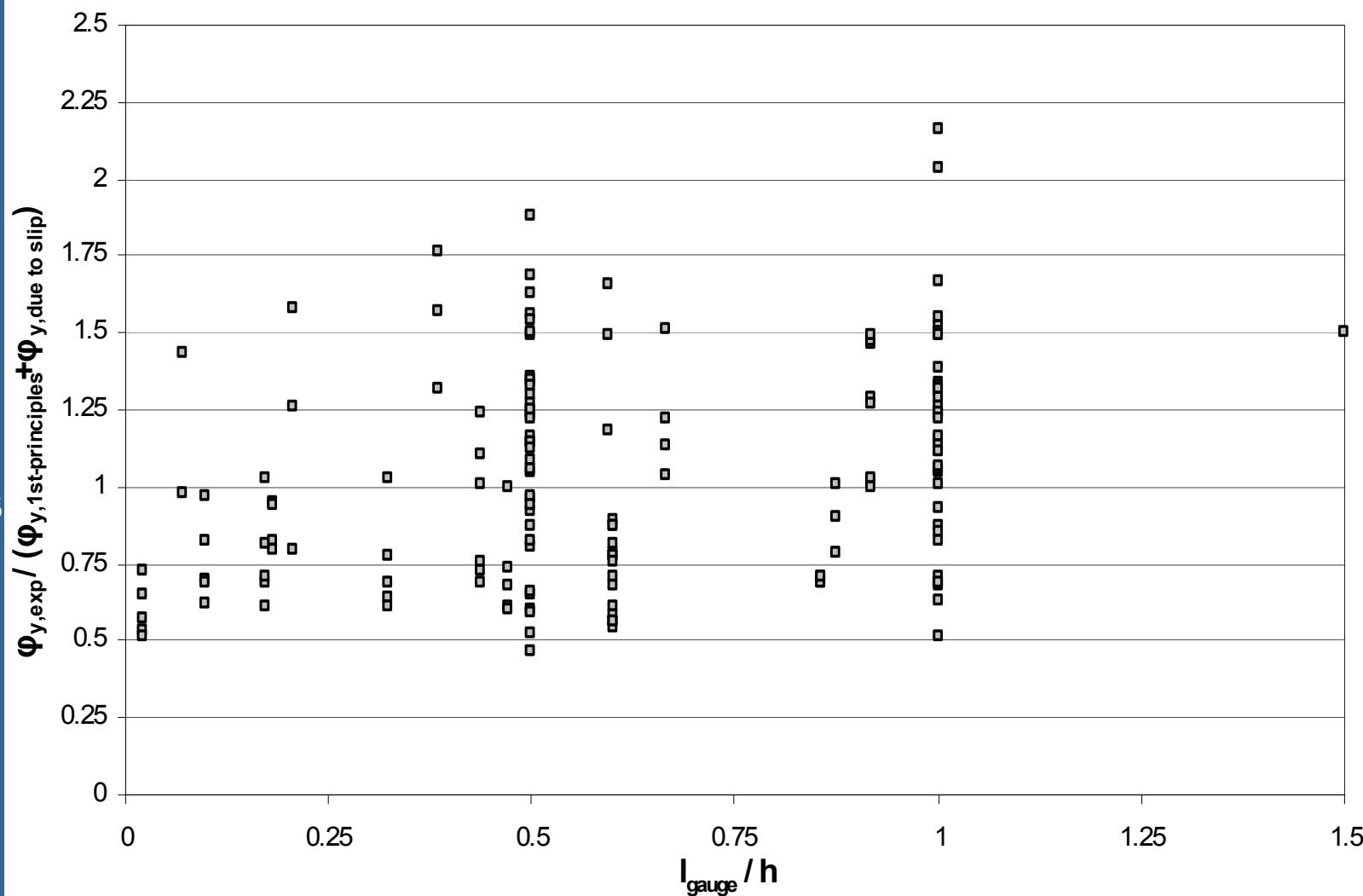
$$\theta_{y, \text{slip}} = \frac{\varphi_y d_{bL} f_y}{8\sqrt{f_c}} \quad (f_{yL}, f_c \text{ in MPa})$$



Fixed-end rotation of member end due to rebar pull-out from anchorage zone beyond member end, at member yielding

$\varphi_{y,\text{measured}} / (\varphi_{y,\text{predicted}} + \theta_{y,\text{slip}} / l_{\text{gauge}})$ no. 160 measurements w/ slip:
median = 1.0, C.o.V = 34%

Ratio:
experimental-to-
predicted yield
curvature (w/
correction for
fixed-end-
rotation) in terms
of gauge length



Chord rotation of shear span at yielding of end section per EN 1998-3:2005

- Rect. beams or columns:

$$\theta_y = \varphi_y \frac{L_s + a_V z}{3} + 0.0014 \left(1 + 1.5 \frac{h}{L_s} \right) + a_{sl} \theta_{y,slip}$$

- Rect. or non-rect. walls:

$$\theta_y = \varphi_y \frac{L_s + a_V z}{3} + 0.0013 + a_{sl} \theta_{y,slip}$$

“Shift rule”:

Diagonal cracking shifts value of force in tension reinforcement to a section at a distance from member end equal to z (internal lever arm)

- $z = d - d_1$ in beams, columns, or walls of barbelled or T-section,
- $z = 0.8h$ in rectangular walls.

— $a_v = 0$, if $V_{Rc} > M_y/L_s$;

— $a_v = 1$, if $V_{Rc} \leq M_y/L_s$.

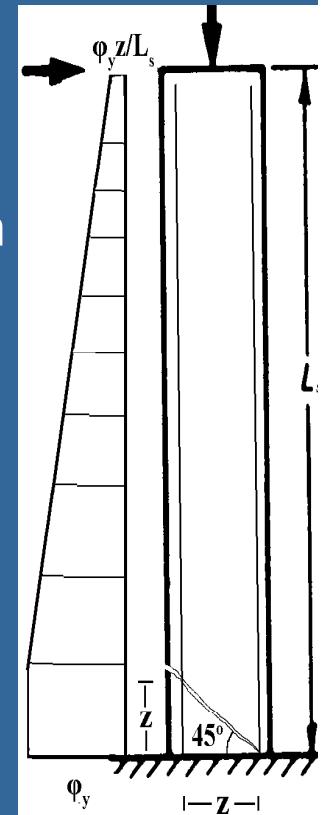
V_{Rc} = force at diagonal cracking per Eurocode 2

(in kN, dimensions in m, f_c in MPa):

$$V_{R,c} = \left\{ \max \left[180(100\rho_1)^{1/3}, 35 \sqrt{1 + \sqrt{\frac{0.2}{d}} f_c^{1/6}} \left(1 + \sqrt{\frac{0.2}{d}} \right) f_c^{1/3} + 0.15 \frac{N}{A_c} \right] b_w d \right\}$$

— $a_{sl} = 0$, if no slip from anchorage zone beyond end section;

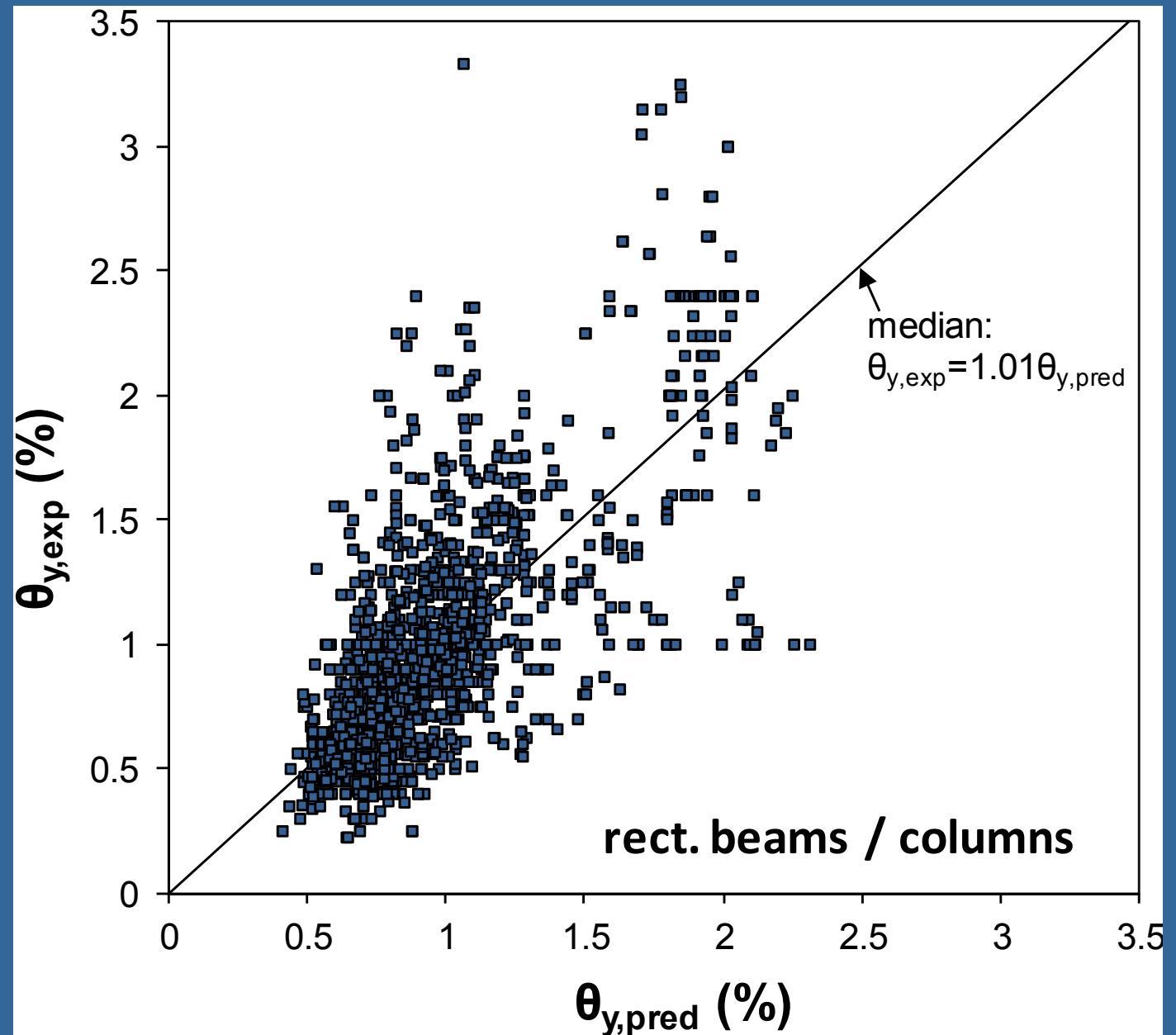
— $a_{sl} = 1$, if there is slip from anchorage zone beyond end section.



Test-model comparison – θ_y

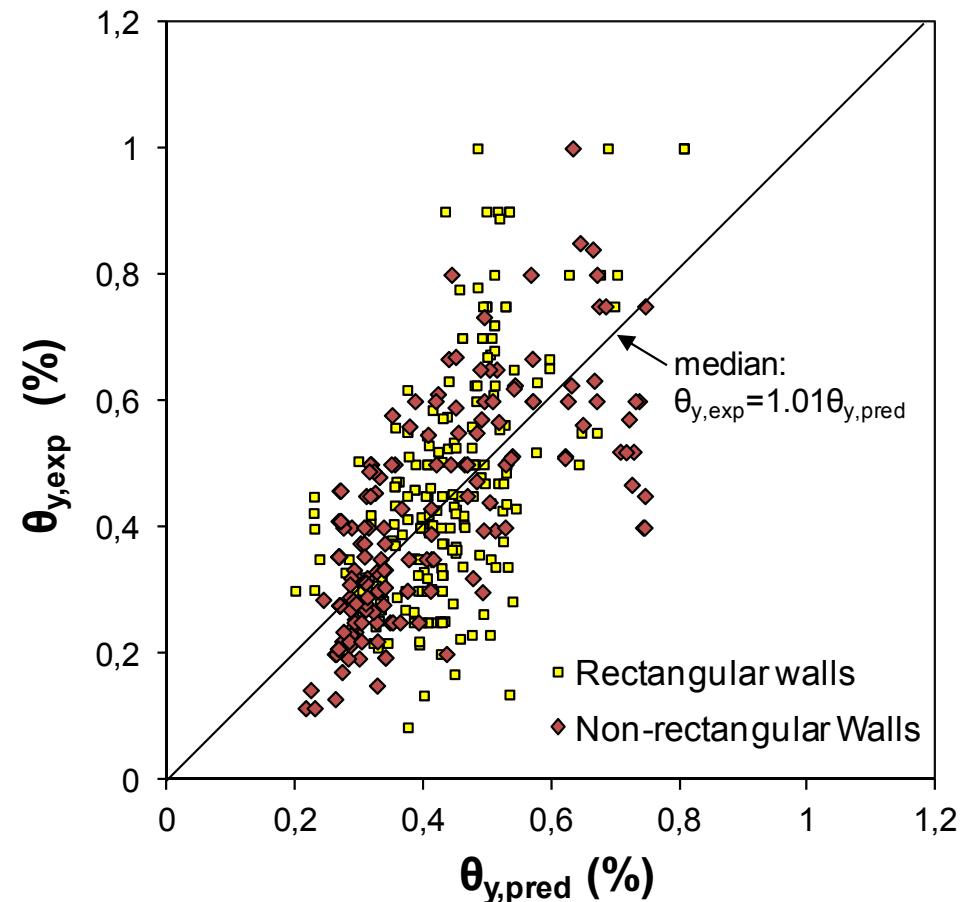
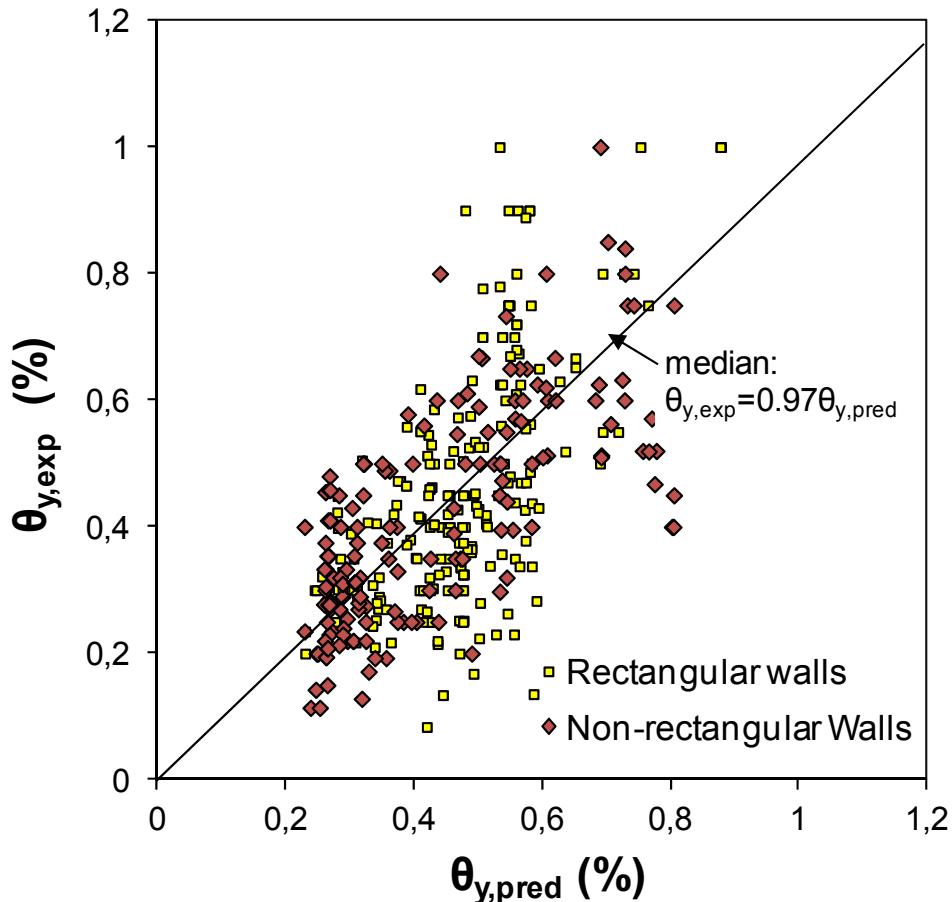
Beams/rect. columns, no. tests: 1653

median=1.01,
CoV=32.1%



Test-model comparison – θ_y

Walls, no. tests: 386



Expression in EN 1998-3:2005
 median=0.97, CoV=31.1%

Modified expression adopted in MC2010

$$\theta_y = \varphi_y \frac{L_s + a_V z}{3} + 0.00045 \left(1 + \frac{5h}{3L_s} \right) + a_{sl} \theta_{y,slip}$$

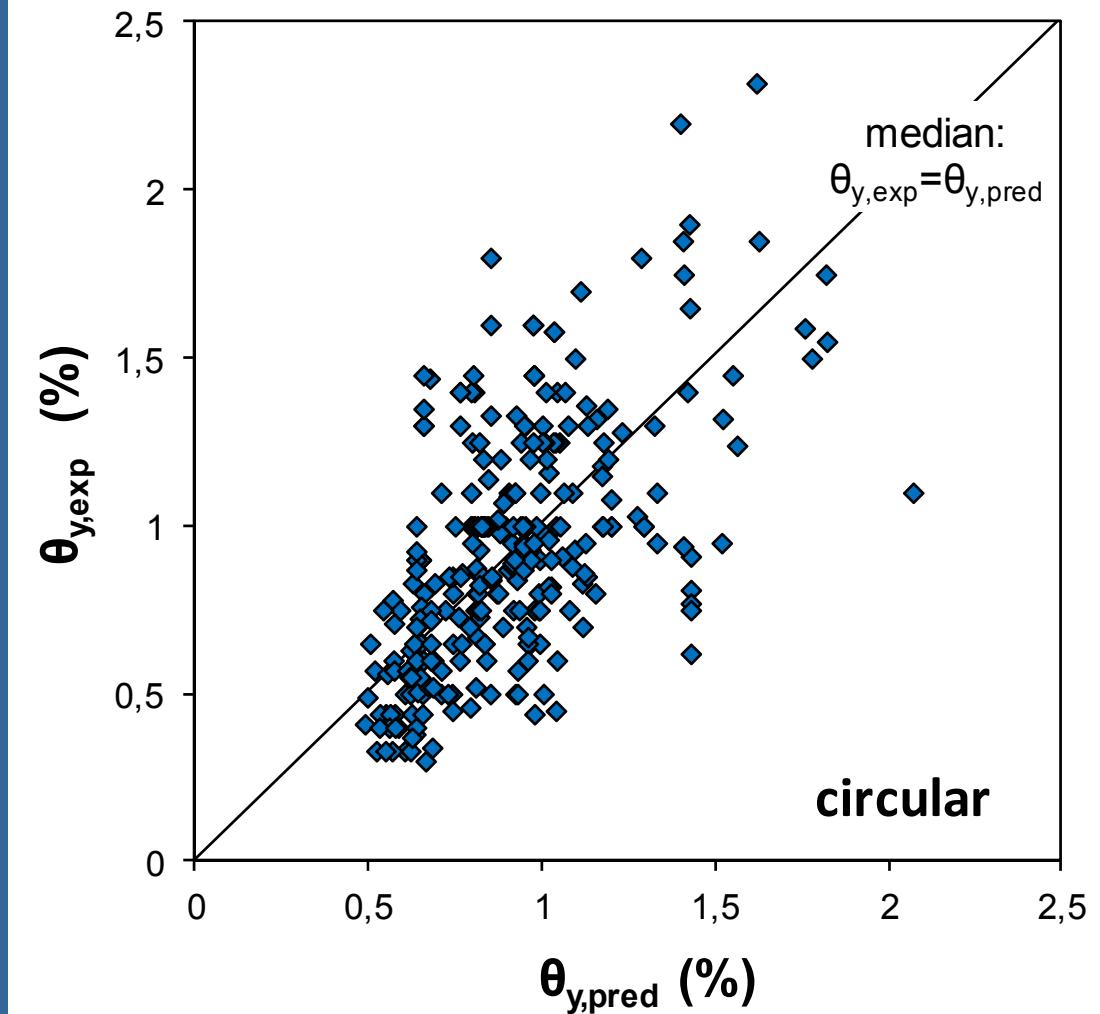
median=1.01, CoV=30.9%

Test-model comparisons - θ_y

Circular columns - not in EN 1998-3:2005: no. tests: 291

$$\theta_y = \varphi_y \frac{L_s + a_V z}{3} + 0.0027 \left(1 - \min \left(1; \frac{2L_s}{15D} \right) \right) + a_{sl} \theta_{y,slip}$$

median=1.00, CoV=31.7%



Effective elastic stiffness, EI_{eff} (for linear or nonlinear analysis)

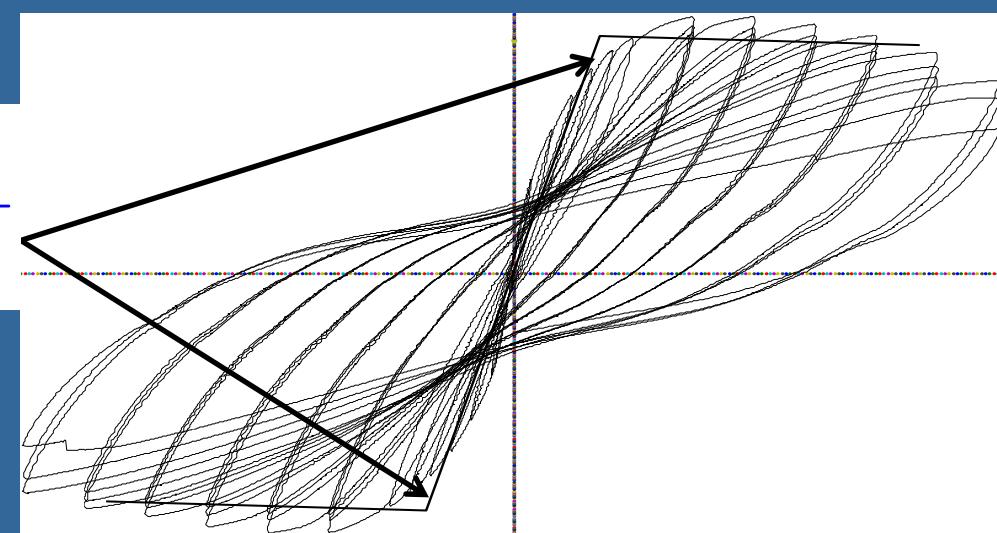
- Part 1 of EC8 (for design of new buildings):

- EI_{eff} : secant stiffness at yielding = 50% of uncracked gross-section stiffness.
- OK in force-based design of new buildings (safe-sided for forces);
- Unsafe in displacement-based assessment for displacement demands).

- More realistic:

$$EI_{\text{eff}} = \frac{M_y L_s}{3\theta_y}$$

- secant stiffness at yielding of end of shear span $L_s = M/V$

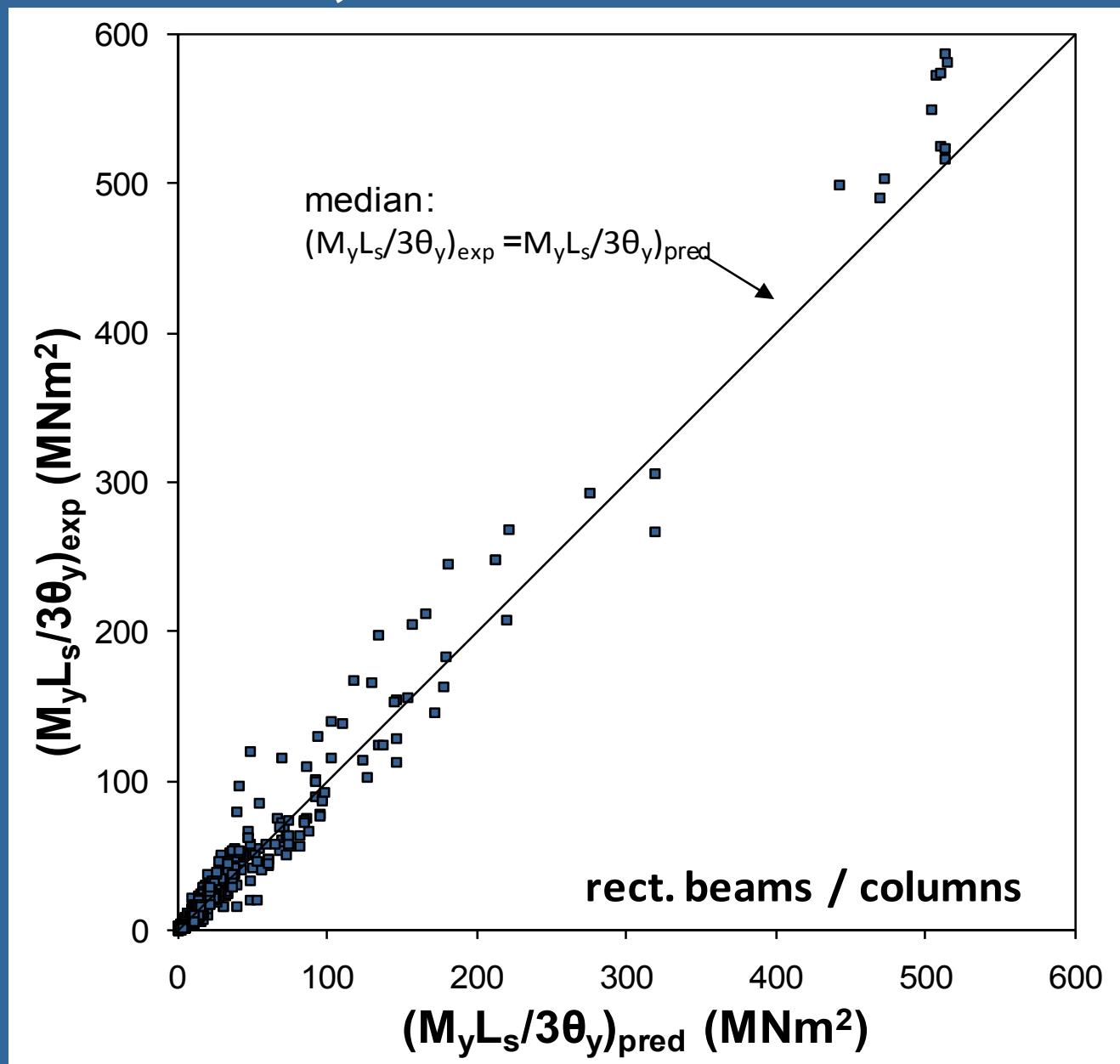


Test-model comparison – EI_{eff}

Beams/rect. columns, no. tests: 1616

median=1.00,
CoV=32.1%

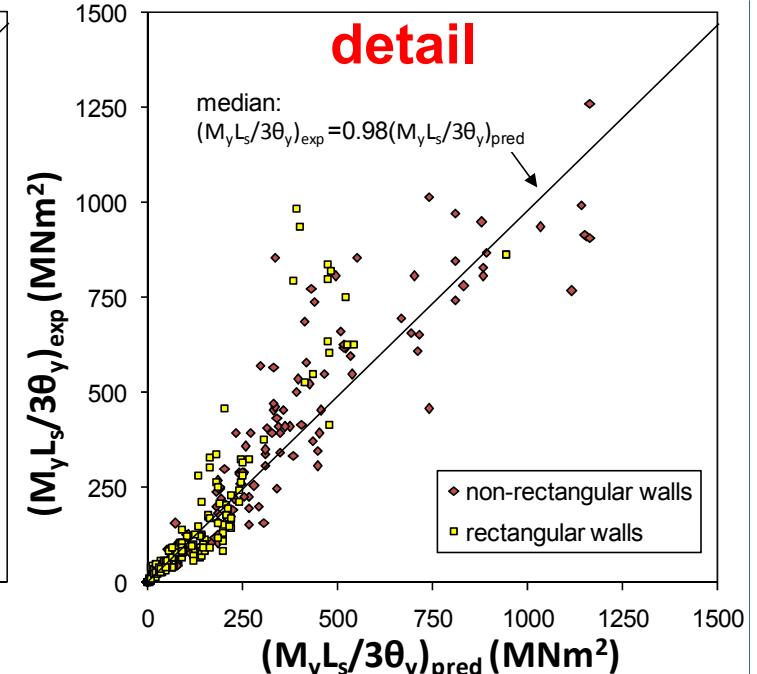
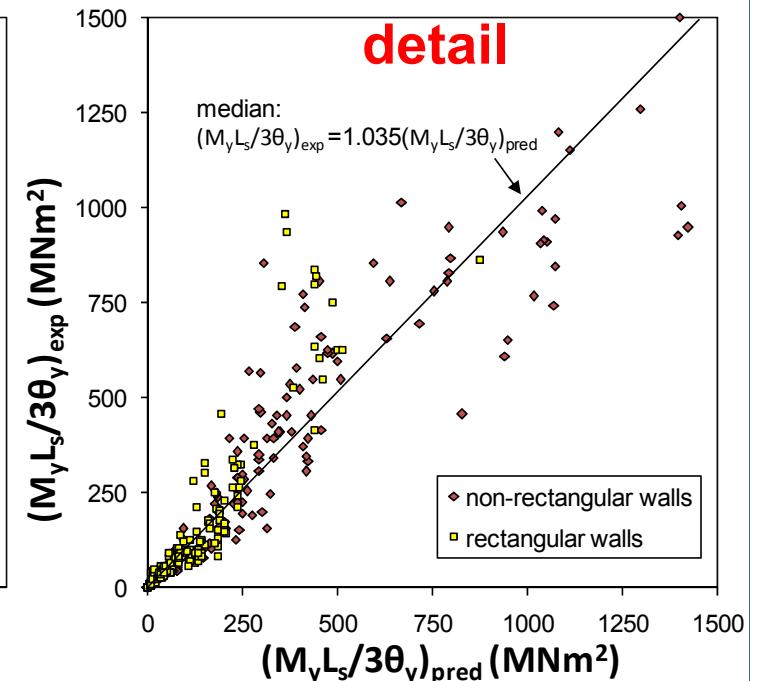
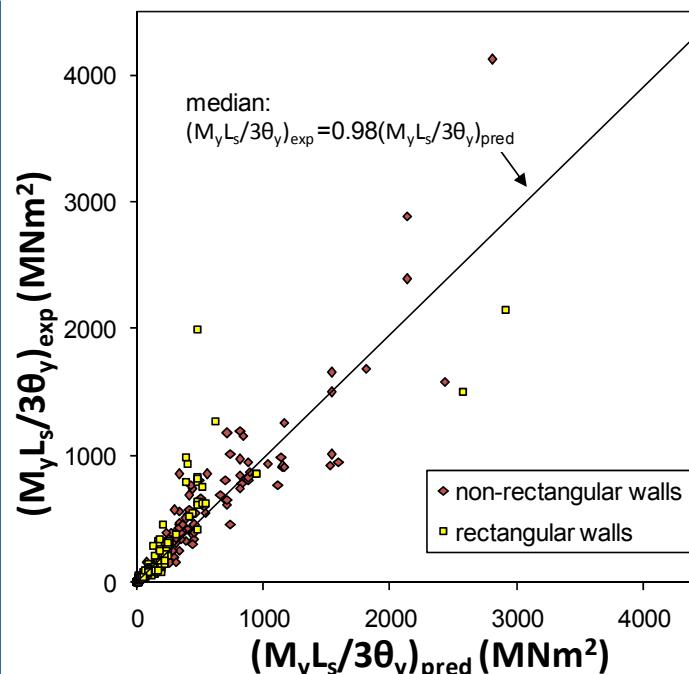
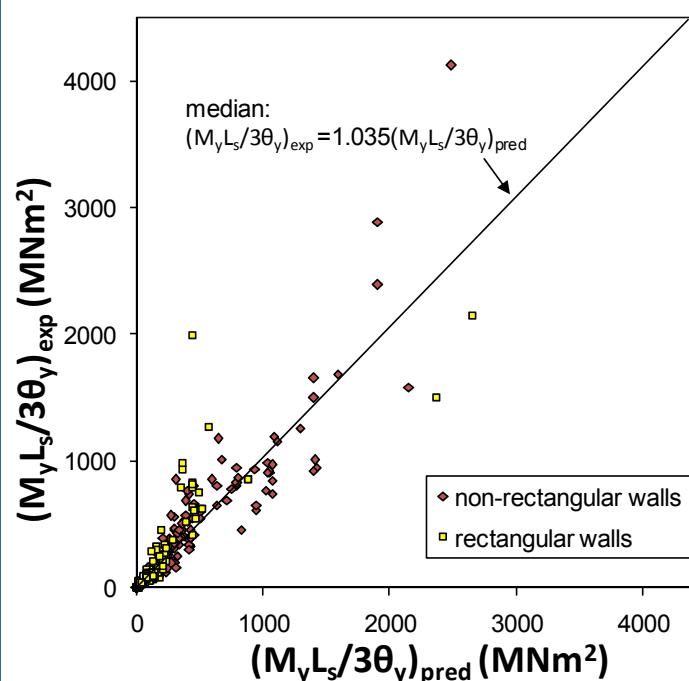
$$EI_{\text{eff}} = \frac{M_y L_s}{3\theta_y}$$



Test-model comparison – EI_{eff} Walls, no. tests: 386

$$EI_{\text{eff}} = \frac{M_y L_s}{3\theta_y}$$

With expression for θ_y
in EN 1998-3:2005
median=1.035
CoV=42.8%



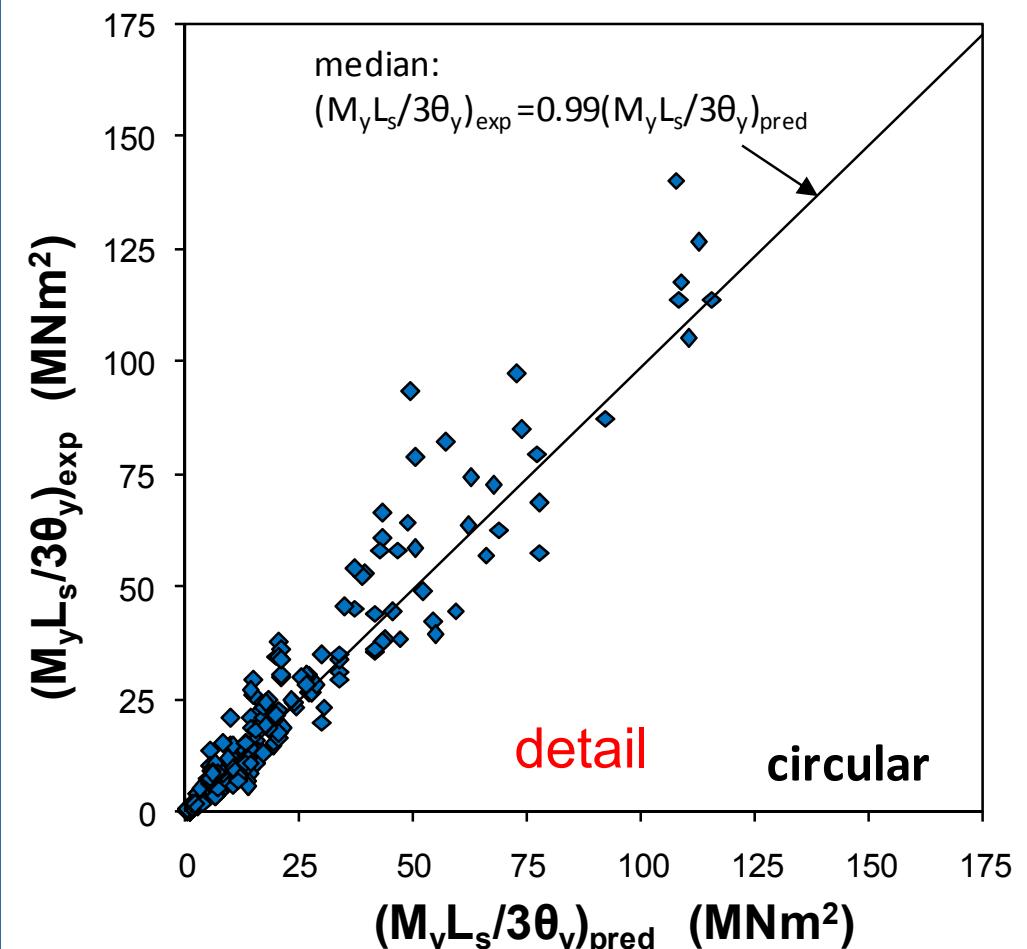
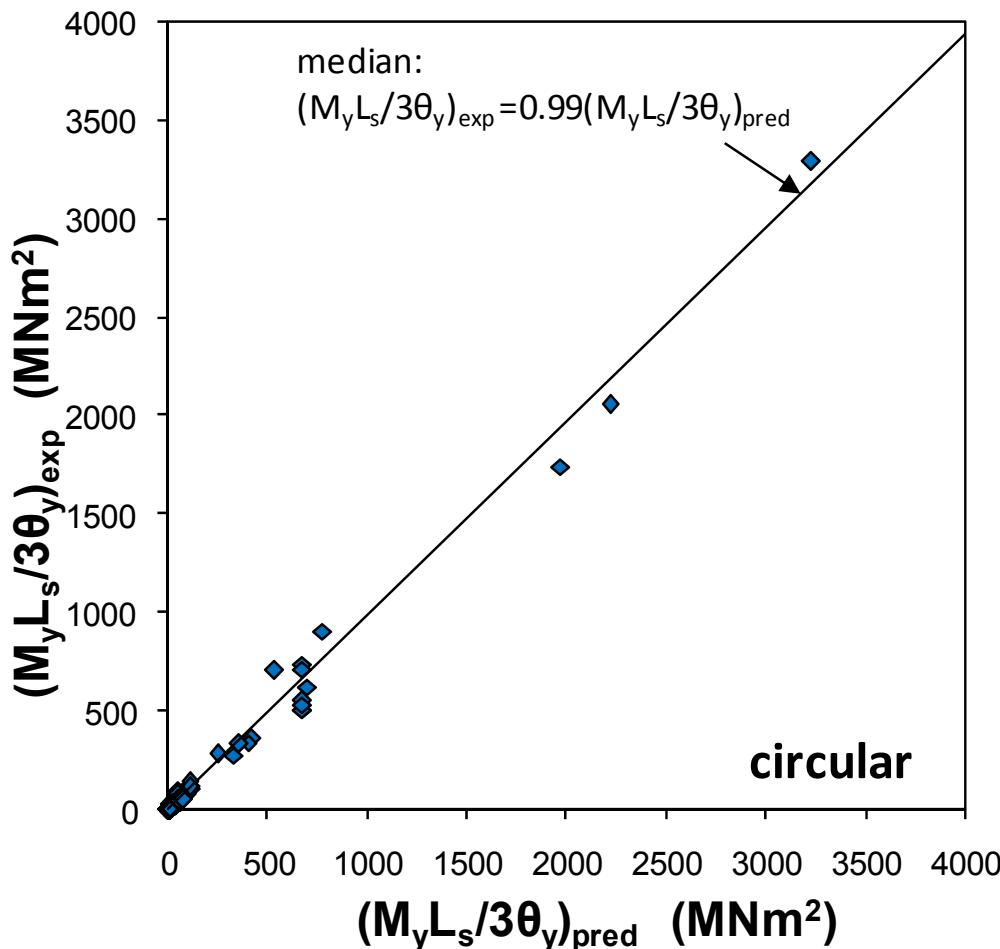
With new expression
for θ_y of walls
(adopted in MC2010)
median=0.98
CoV=41.1%

Test-model comparison – EI_{eff}

Circular columns - not in EN 1998-3:2005, no. tests: 273

$$EI_{\text{eff}} = \frac{M_y L_s}{3\theta_y}$$

median=0.99, CoV=31.2%



Empirical secant stiffness to yielding, EI_{eff} , independent of amount of reinforcement - not in EN 1998-3:2005

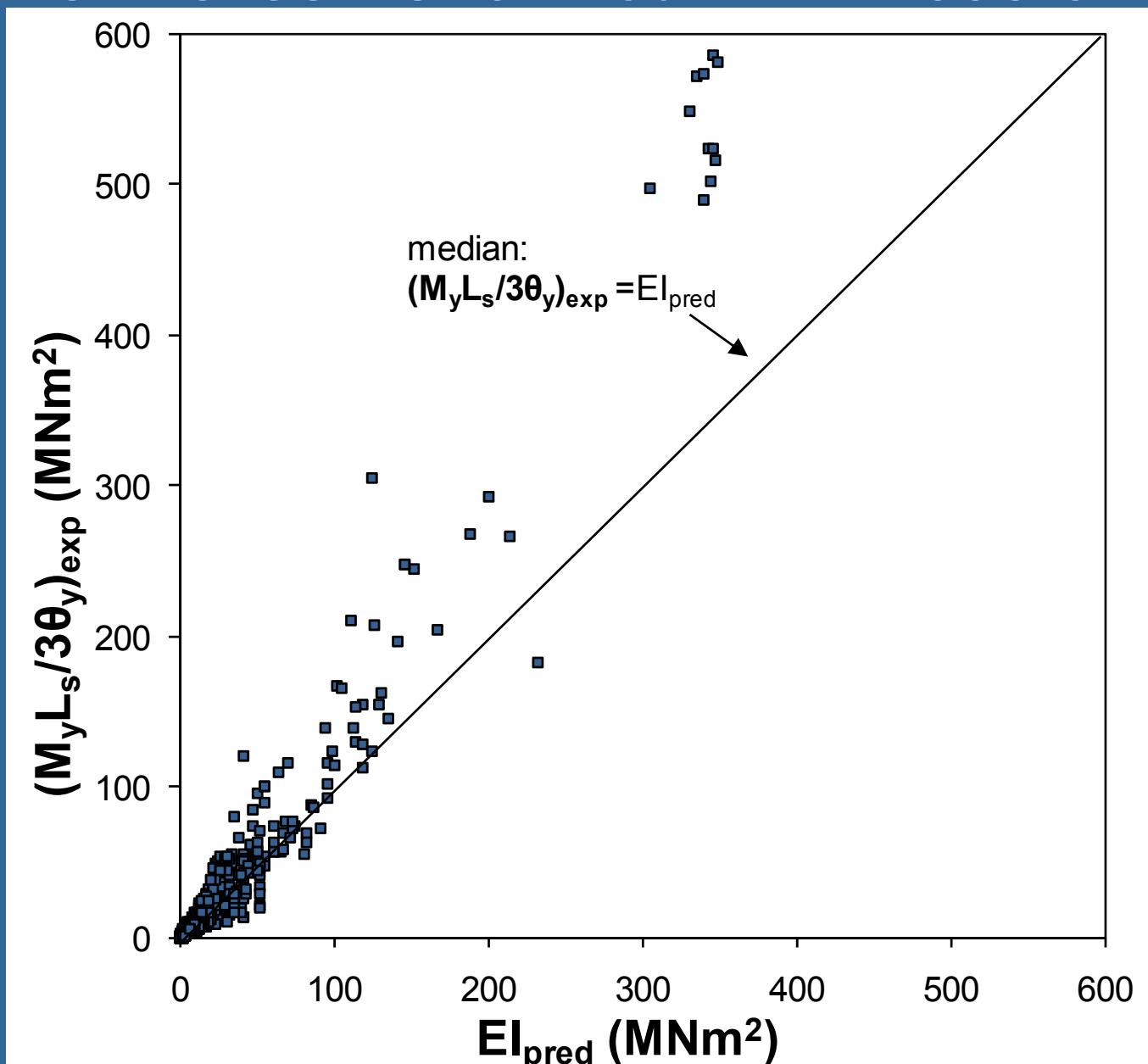
$$\frac{EI_{\text{eff}}}{E_c I_c} = a \left(0.8 + \ln \left(\max \left(\frac{L_s}{h}; 0.6 \right) \right) \right) \left(1 + 0.048 \min [50; \frac{N}{A_c} (\text{MPa})] \right)$$

(all variables known before dimensioning the longitudinal reinforcement)

- If there is slippage of longitudinal bars from their anchorage zone beyond the member end:
 - $a = 0.081$ for columns;
 - $a = 0.10$ for beams or non-rectangular walls (barbelled, T-, H-section);
 - $a = 0.115$ for rectangular walls;
 - $a = 0.12$ for members with circular section.
- If there is no slippage of longitudinal bars: effective stiffness $\times 4/3$

Test-model comparison – Empirical EI_{eff} , independent of amount of reinforcement - not in EN 1998-3:2005

Beams/columns
no. tests: 1616
median=1.00
CoV=36.1%



Test-model comparison – Empirical EI_{eff} independent of reinforcement , not in EN1998-3: 2005

Walls

no. tests: 386

median=1.00

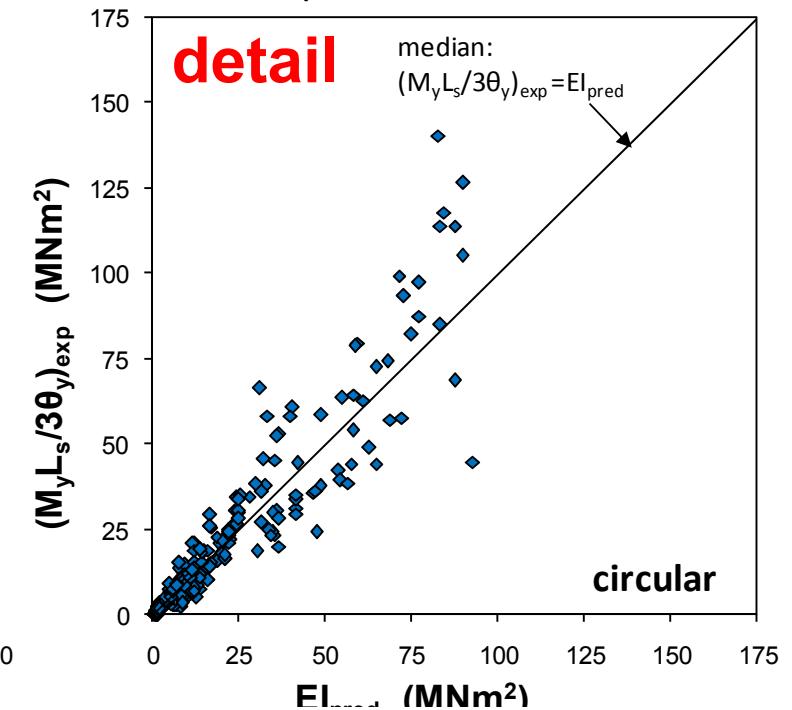
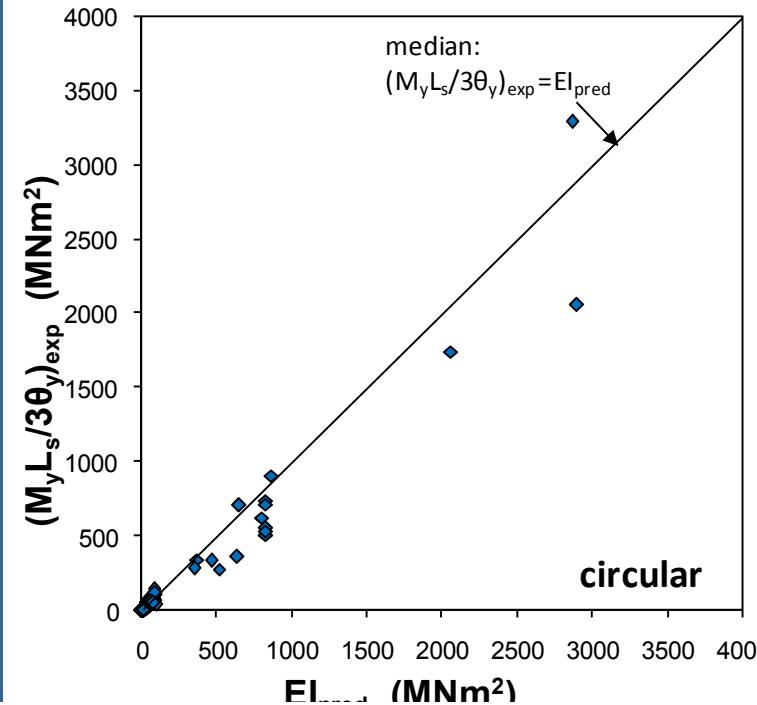
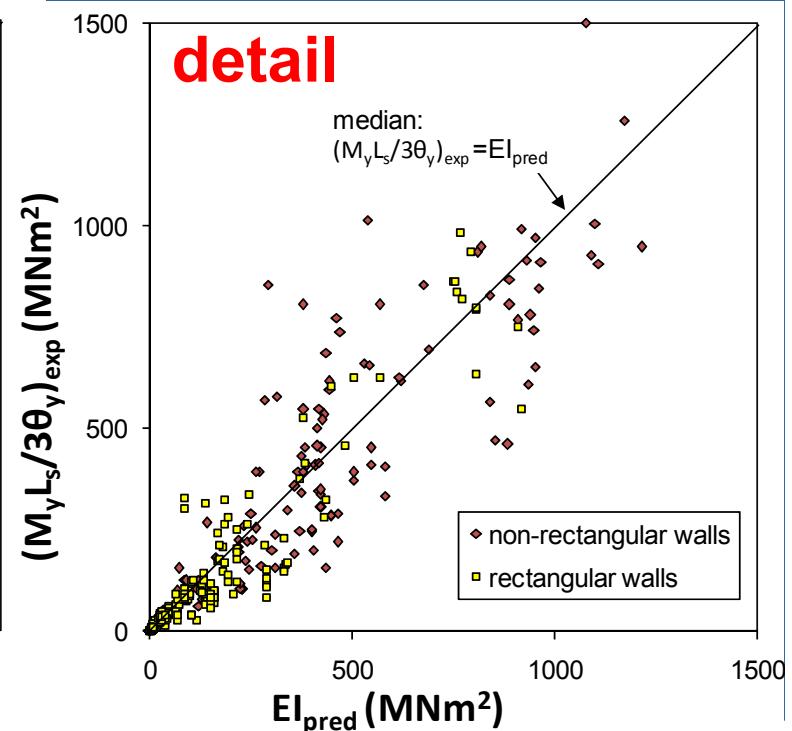
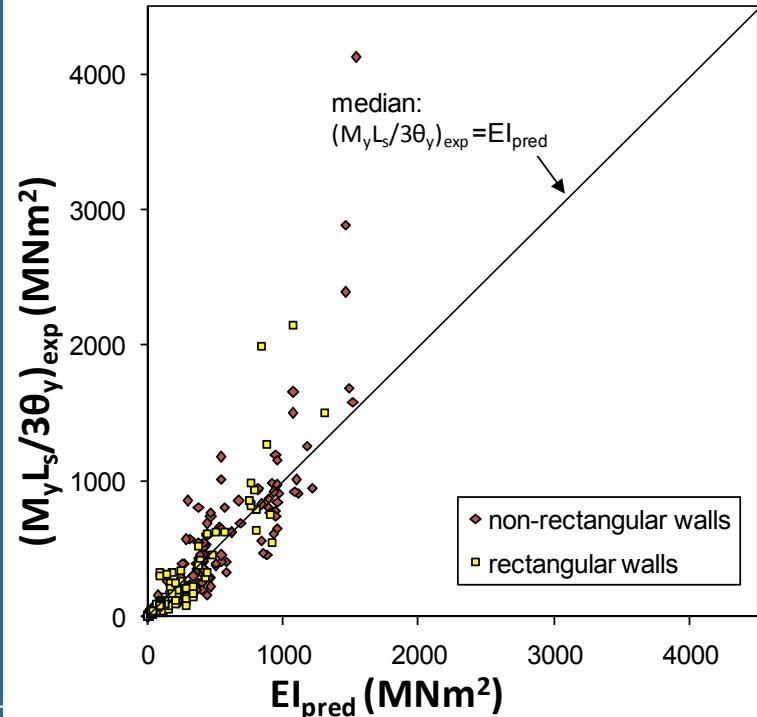
CoV=44.6%

Circular columns

no. tests: 273

median=0.995

CoV=31.4%



Flexure-controlled ultimate chord rotation from curvature & plastic hinge length per EN 1998-3:2005

$$\theta_u = \theta_y + \theta_{pl}^{pl} = \theta_y + (\varphi_u - \varphi_y) L_{pl} \left(1 - \frac{0.5 L_{pl}}{L_s} \right)$$

- ϕ_y : yield curvature (section analysis);

$$\phi_u = \min \left(\frac{\varepsilon_{cu,c}}{\xi_{cu,c} d_c}, \frac{\varepsilon_{su}}{(1 - \xi_{su}) d} \right)$$

Option 1: Confinement per Eurocode 2

$$\alpha \rho_{sx} f_{yw} \leq 0.05 f_c : \quad f_{cc} = f_c + 5 \alpha \rho_{sx} f_{yw}$$

$$\alpha \rho_{sx} f_{yw} > 0.05 f_c : \quad f_{cc} = 1.125 f_c + 2.5 \alpha \rho_{sx} f_{yw}$$

$$\varepsilon_{cu,c} = \varepsilon_{co} + 0.2 \left(\alpha \rho_{sx} f_{yw} / f_c \right)$$

$$L_{pl} = 0.1 L_s + 0.17 h + 0.24 d_b f_y / \sqrt{f_c}$$

index c: confined;

ρ_s : stirrup ratio;

L_s =M/V: shear span at member end;

h : section depth;

d_b : bar diameter;

f_y, f_c : MPa

Option 2: New, per EN 1998-3:2005

$$f_{cc} = f_c \left[1 + 3.7 \left(\alpha \rho_{sx} f_{yw} / f_c \right)^{0.86} \right]$$

$$\varepsilon_{cu,c} = 0.004 + 0.5 \left(\alpha \rho_{sx} f_{yw} / f_{cc} \right)$$

$$L_{pl} = L_s / 30 + 0.2 h + 0.11 d_b f_y / \sqrt{f_c}$$

α : confinement effectiveness:

– rectangular section:

– circular section & hoops:

s_h : centerline spacing of stirrups,

D_c, b_c, h_c : confined core dimensions to centerline of hoop;

b_i : centerline spacing along section perimeter of longitudinal bars (index: i) engaged by a stirrup corner or cross-tie.

$$\alpha = \left(1 - \frac{s_h}{2b_c} \right) \left(1 - \frac{s_h}{2h_c} \right) \left(1 - \frac{\sum b_i^2}{6b_c h_c} \right)$$

$$\alpha = \left(1 - \frac{s_h}{2D_c} \right)^2$$

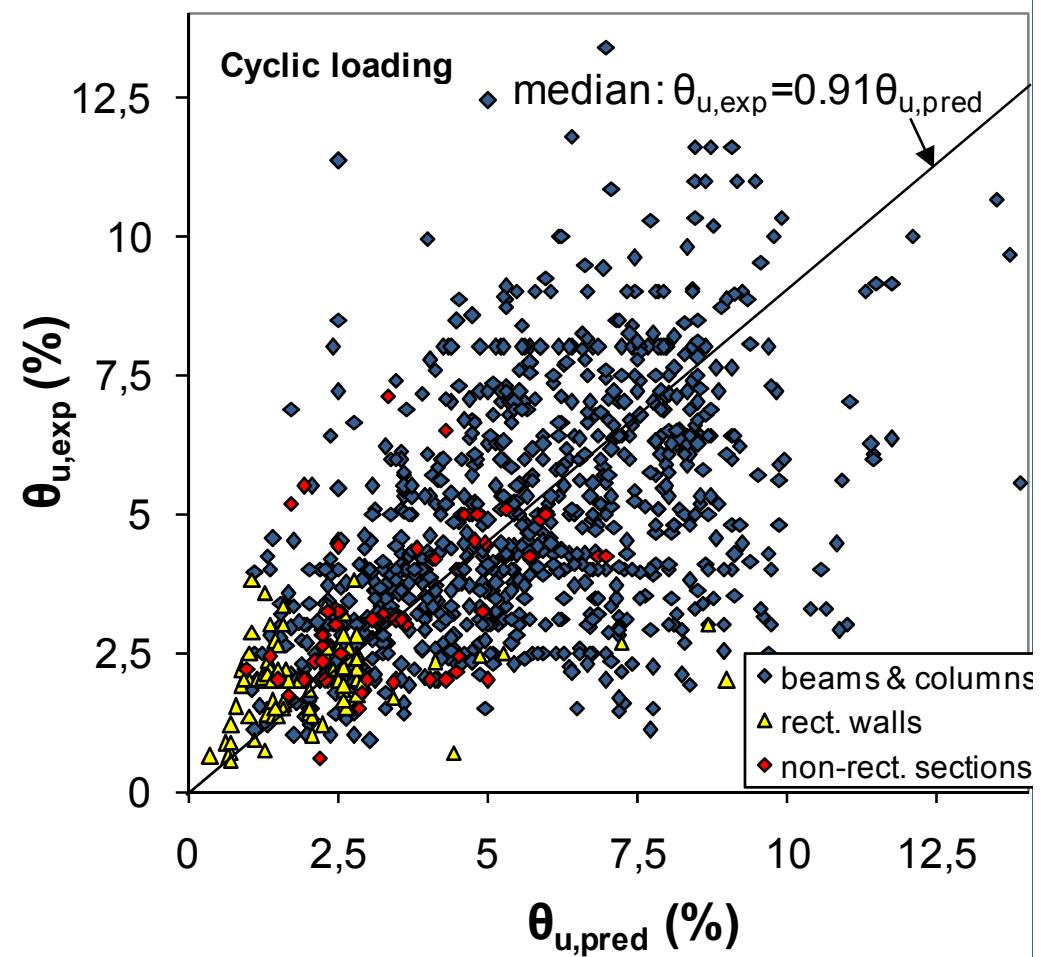
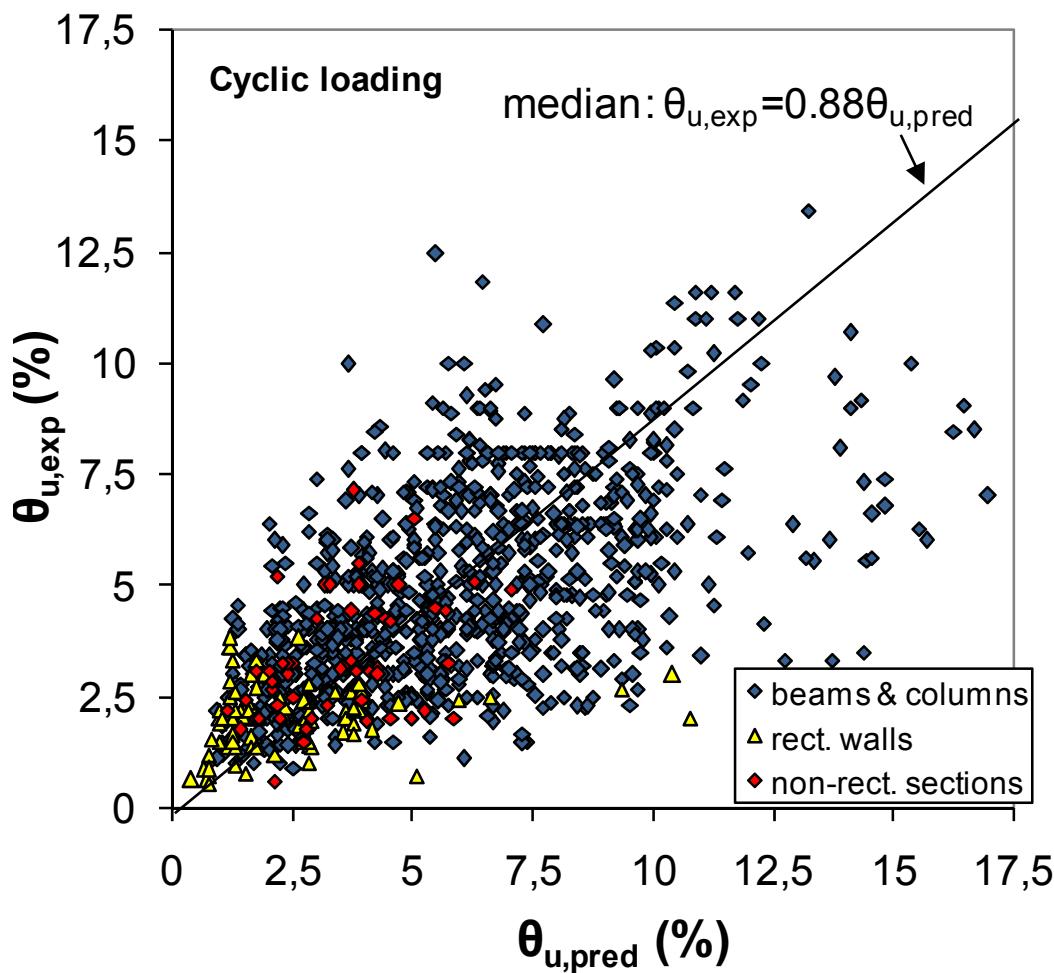
Test-model comparison – ultimate chord rotation from curvatures & plastic hinge length per EN 1998-3:2005, no. tests: 1100

Option 1: confinement per EC2

median=0.88, CoV=52.3%

Option 2: new confinement per EC8-3

median=0.91, CoV=52.2%



Flexure-controlled ultimate chord rotation of rect. beams, columns, walls, non-rect. walls & circular columns from curvatures & plastic hinge length by Biskinis & Fardis 2010, 2013 - adopted in fibMC2010

Flexure-controlled ultimate chord rotation, accounting separately for slippage in yield-penetration length, from yielding till ultimate deformation:

$$\theta_u = \theta_y + a_{sl} \Delta \theta_{u,slip} + (\varphi_u - \varphi_y) L_{pl} \left(1 - \frac{L_{pl}}{2L_s} \right) \quad \text{Confinement per fib MC2010: } f_{cc} = f_c \left[1 + 3.5 \left(\frac{\alpha \rho_w f_{yw}}{f_c} \right)^{3/4} \right]$$

- Monotonic loading - Rect. beams, columns, walls, non-rectangular walls:

$$\varepsilon_{cu,c} = 0.0035 + \left(\frac{10}{h_o \text{ (mm)}} \right)^2 + 0.57 \frac{\alpha \rho_w f_{yw}}{f_{cc}}, \quad \varepsilon_{su,mon} = \frac{7}{12} \varepsilon_{su,nominal}$$

$$L_{pl,mon} = h \left(1.1 + 0.04 \min \left(9; \frac{L_s}{h} \right) \right)$$

- Cyclic loading:

$$\varepsilon_{cu,c} = 0.0035 + \left(\frac{10}{h_o \text{ (mm)}} \right)^2 + 0.4 \frac{\alpha \rho_w f_{yw}}{f_{cc}}, \quad \varepsilon_{su,cy} = \frac{3}{8} \varepsilon_{su,nominal}$$

- **Rect. beams/columns/walls, non-rect. walls:** $L_{pl,cy} = 0.2h \left(1 + \frac{1}{3} \min \left(9; \frac{L_s}{h} \right) \right)$

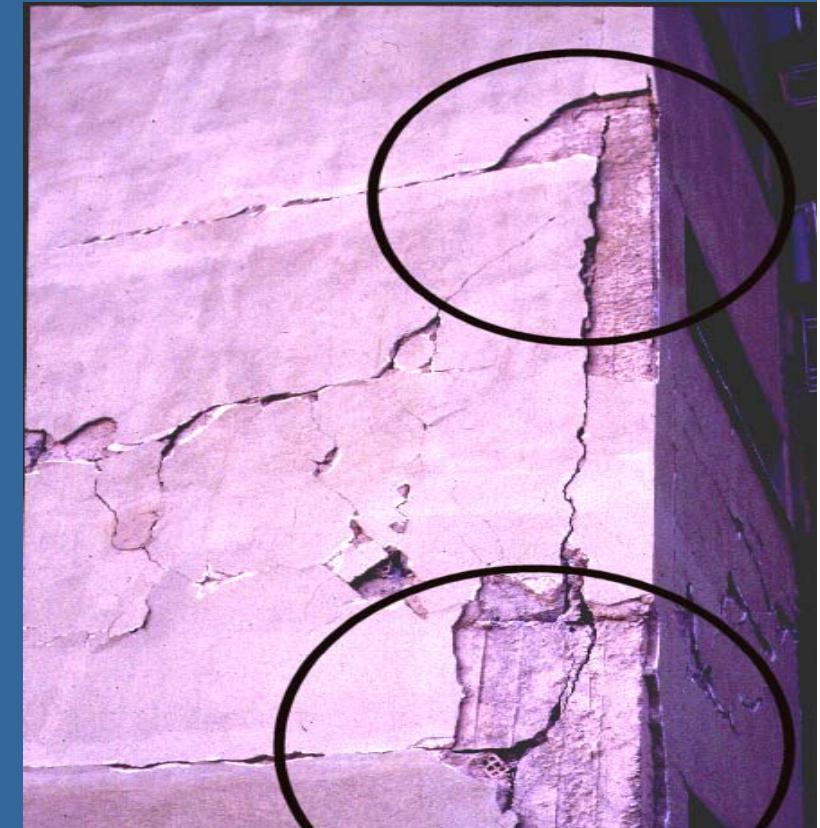
- **Circular columns:** $L_{pl,cy,cir} = 0.6D \left(1 + \frac{1}{6} \min \left(9; \frac{L_s}{D} \right) \right)$

Post-yield fixed-end rotation of member end due to bar slippage from yield penetration length beyond member end, from yielding till ultimate flexural deformation per Biskinis & Fardis 2010, 2013 - adopted in *fibMC2010* (not in EN 1998-3:2005)

- **Monotonic loading:** $\Delta\theta_{u,slip} = 9.5d_{bL}\varphi_u \quad or \quad = 16d_{bL}(\varphi_y + \varphi_u)/2$
- **Cyclic loading:** $\Delta\theta_{u,slip} = 5.5d_{bL}\varphi_u \quad or \quad = 10d_{bL}(\varphi_y + \varphi_u)/2$

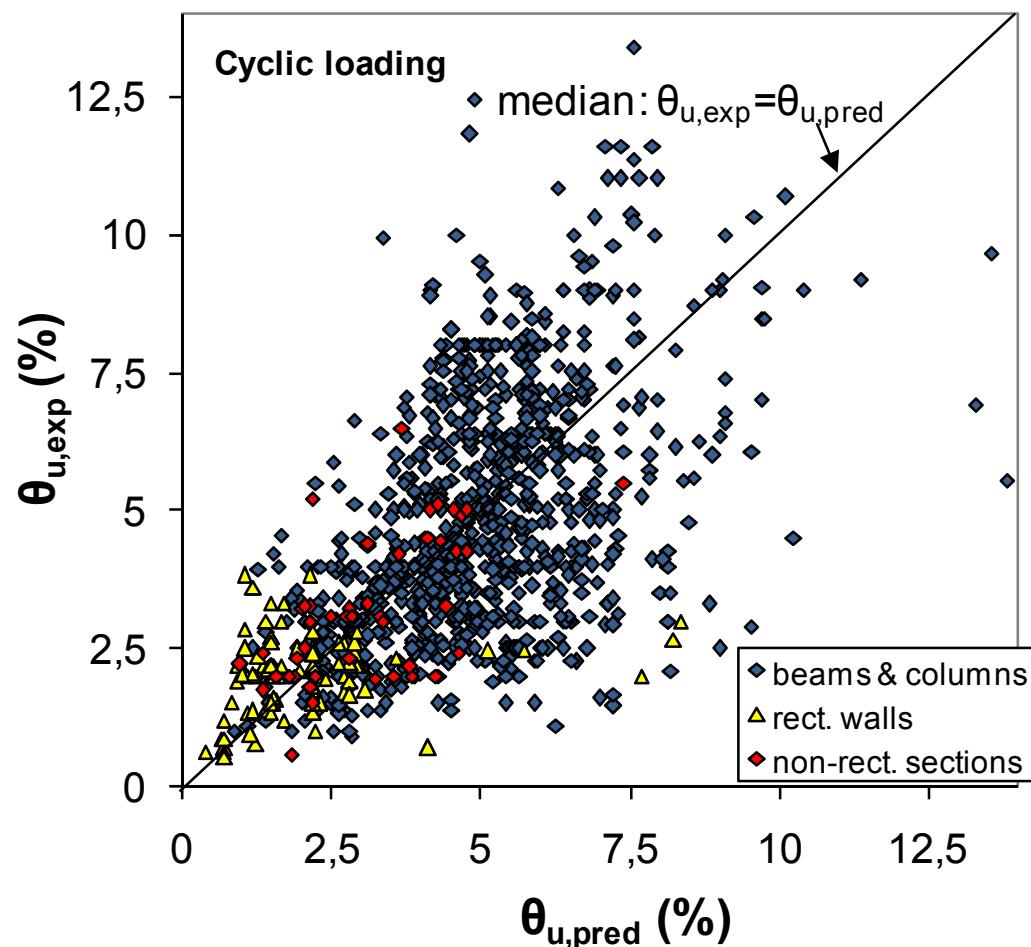


Complete pull-out of
beam bars, due to
short anchorage in
corner joint

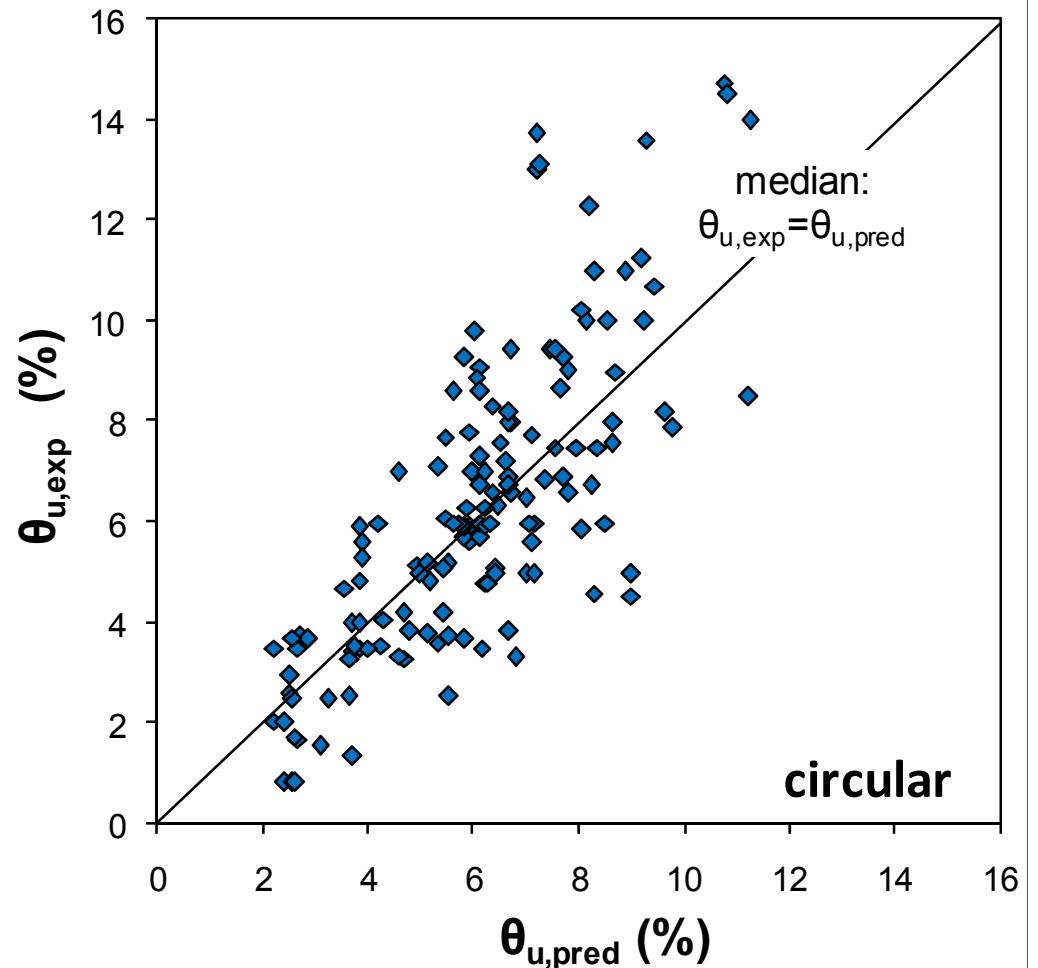


Test-model comparison – Cyclic ultimate chord rotation from curvatures & plastic hinge length per Biskinis & Fardis 2010, 2012

Rect. beams/columns/walls,
non-rect. walls - no. tests: 1100
median=1.00, CoV=43.2%



Circular columns – no. tests:
143, median=1.00, CoV=30.3%



Empirical cyclic flexure-controlled ultimate chord rotation of rect. beams/columns/walls, non-rect. walls in EN1998-3:2005 & fibMC2010

$$\theta_{\text{um}} = 0.016 \cdot (0.3^v) \left[\frac{\max(0.01; \omega')}{\max(0.01; \omega)} f_c \right]^{0.225} \left(\frac{L_s}{h} \right)^{0.35} 25^{\left(\alpha \rho_{sx} \frac{f_{yw}}{f_c} \right)} (1.25^{100 \rho_d})$$

or:

$$\theta_{\text{um}}^{\text{pl}} = \theta_{\text{um}} - \theta_y = 0.0145 \cdot (0.25^v) \left[\frac{\max(0.01; \omega')}{\max(0.01; \omega)} \right]^{0.3} f_c^{0.2} \left(\frac{L_s}{h} \right)^{0.35} 25^{\left(\alpha \rho_{sx} \frac{f_{yw}}{f_c} \right)} (1.275^{100 \rho_d})$$

ω, ω' : mechanical ratio of tension (including web) & compression steel;

v : N/bhf_c (b : width of compression zone; $N>0$ for compression);

L_s/h : M/Vh : shear span ratio;

α : confinement effectiveness factor :
$$\alpha = \left(1 - \frac{s_h}{2b} \right) \left(1 - \frac{s_h}{2h_c} \right) \left(1 - \frac{\sum b_i^2}{6b_c h_c} \right)$$

ρ_{sx} : $A_{sh}/b_w s_h$: transverse steel ratio // direction of loading;

ρ_d : ratio of diagonal reinforcement.

- Walls: 1st expression divided by 1.6; 2nd multiplied by 0.6
- Cold-worked brittle steel: 1st expression divided by 1.6; 2nd by 2.0

Non-seismically detailed members with continuous bars

- Plastic part, $\theta_{\text{um}}^{\text{pl}} = \theta_{\text{um}} - \theta_y$, of ultimate chord rotation: divided by 1.2.

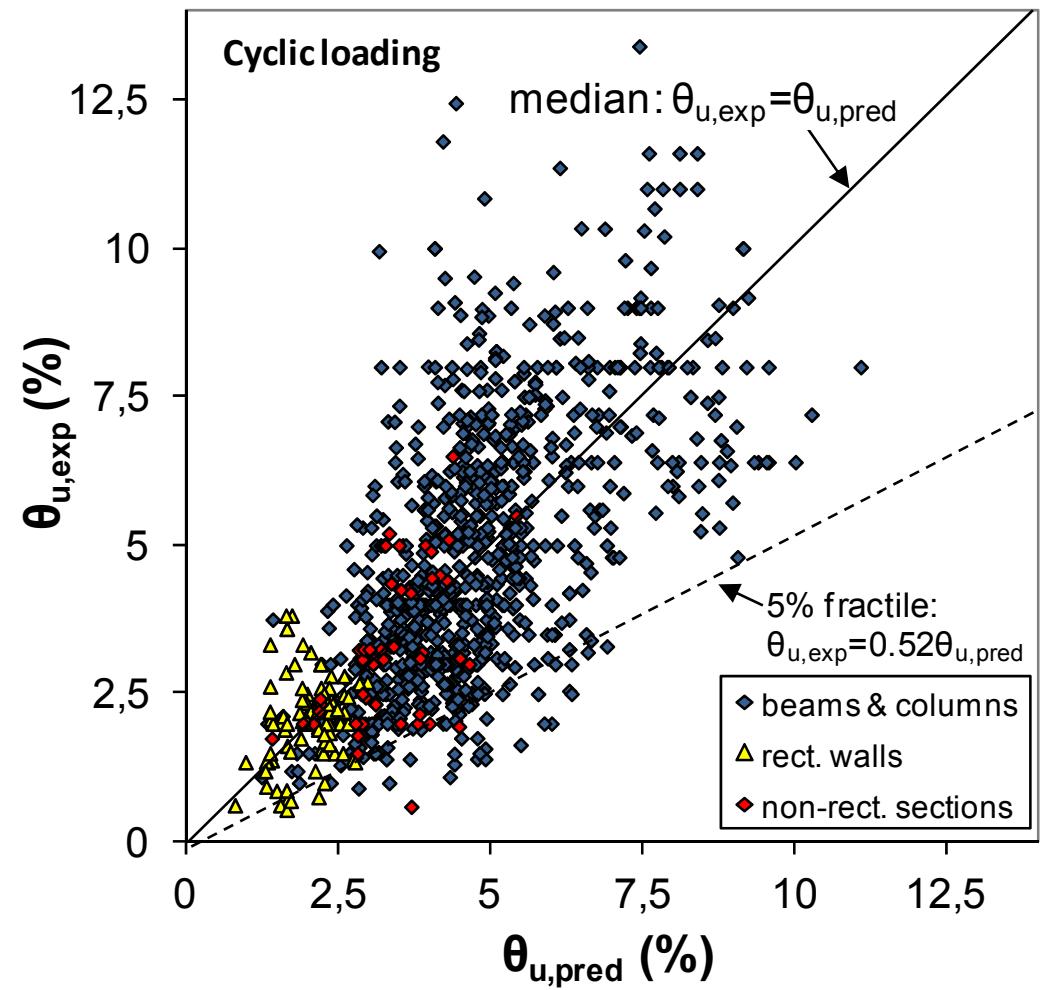
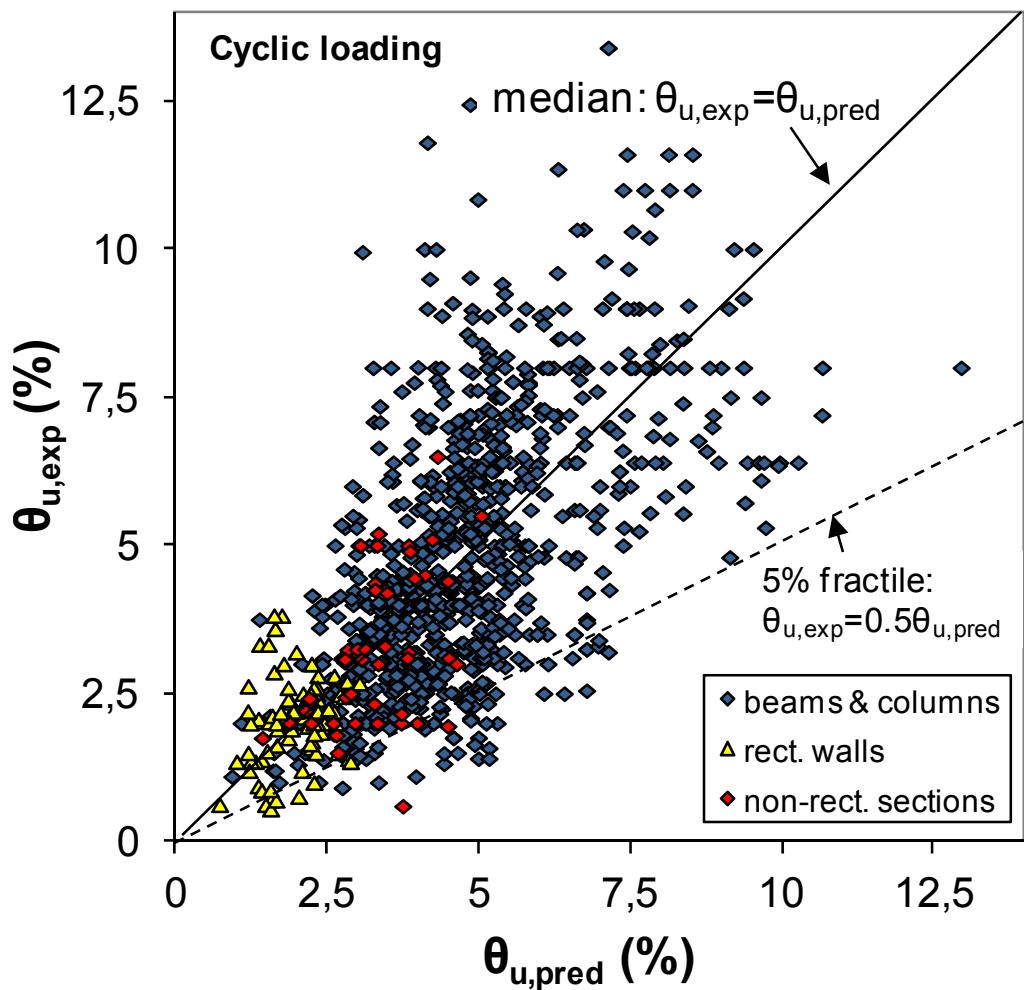
Test-model comparison – Empirical cyclic ultimate chord rotation of members with seismic detailing per EN 1998-3:2005 – no. tests 1100

Model for total $\theta_{u,m}$

median=1.00, CoV=37.8%

Model with $\theta_{u,m}=\theta_{y}+\theta_{pl}$

median=1.00, CoV=37.6%



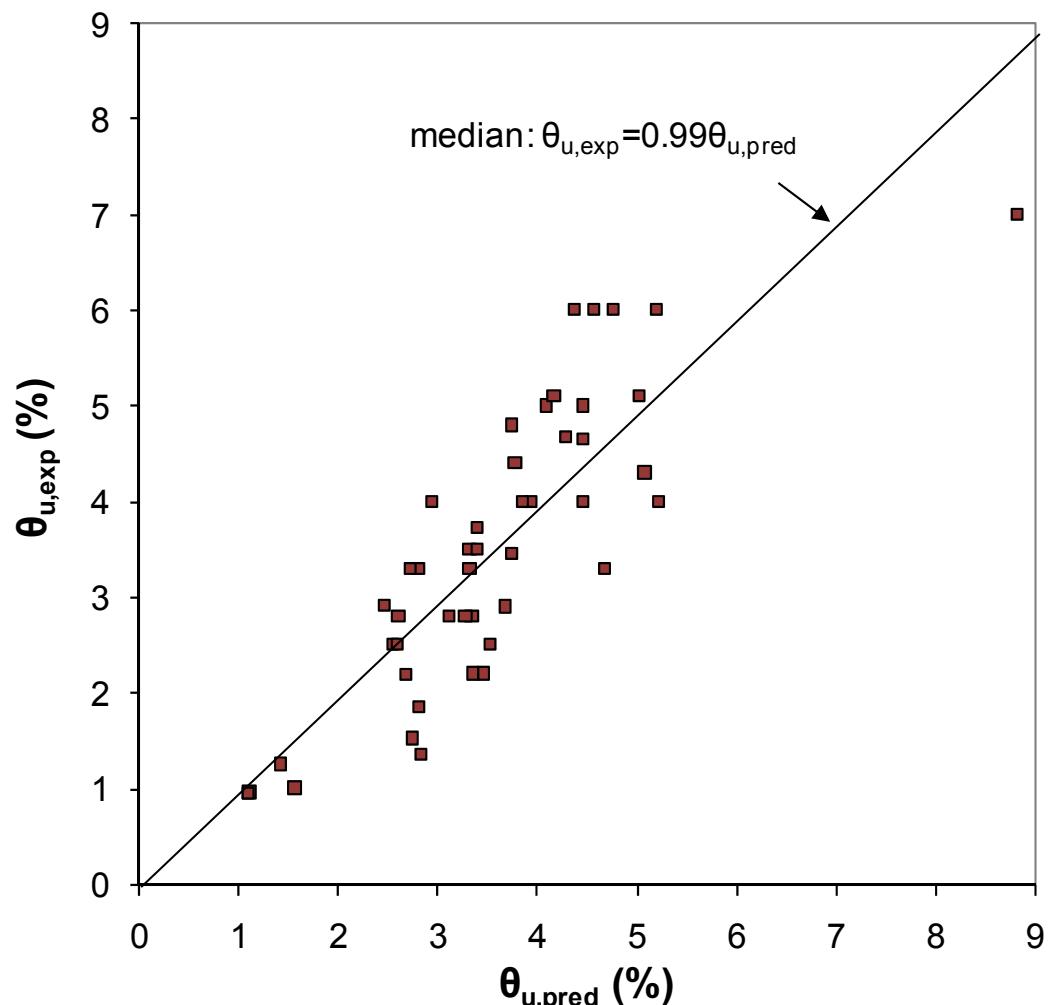
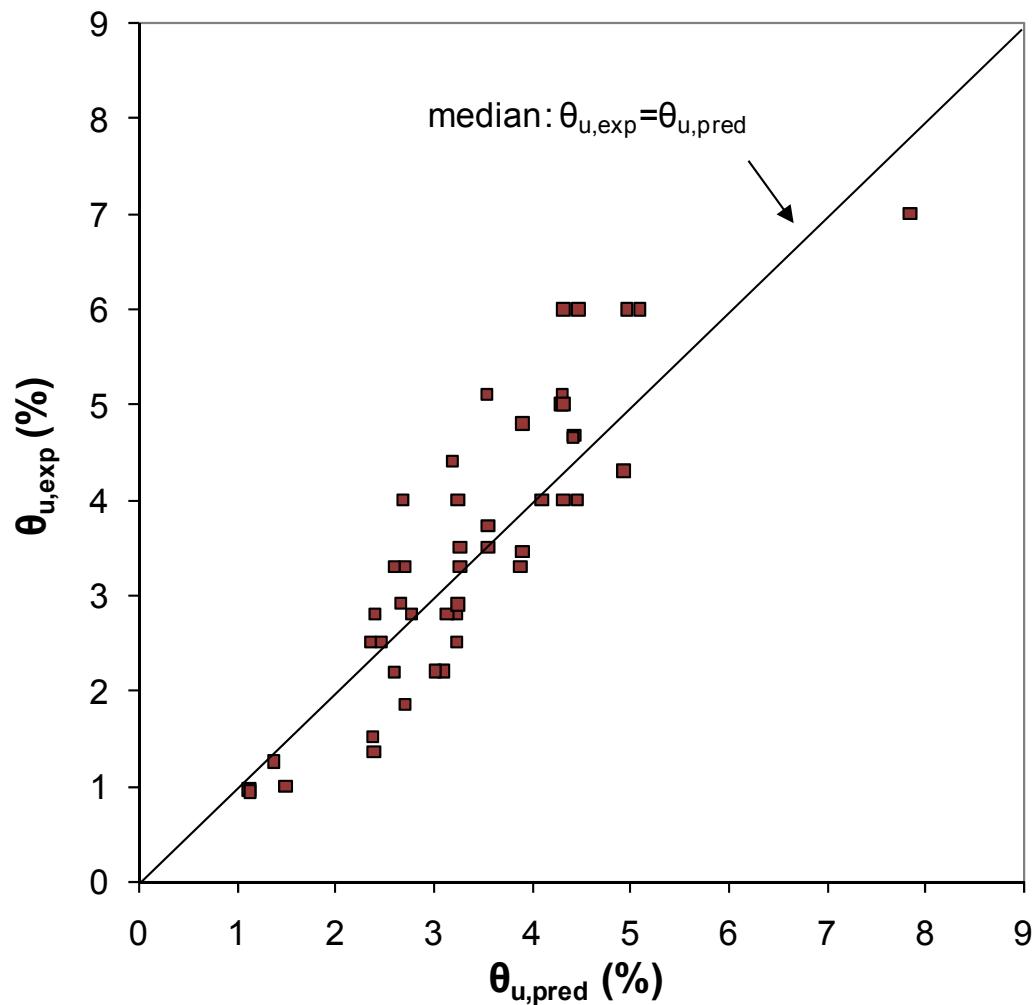
Test-model comparison – Empirical cyclic ultimate chord rotation of members without seismic detailing per EN 1998-3:2005 – no. tests 48

Model for total θ_{um}

median=1.00, CoV=30.6%

Model with $\theta_{um}=\theta_y+\theta_{pl}$

median=0.99, CoV=31.8%



General empirical ultimate chord rotation of rect. beams/columns/walls & non-rect. walls per Biskinis & Fardis 2010, adopted in fib MC2010 (not in EN1998-3:2005)

$$\theta_u^{pl} = \alpha_{st}^{hbw} (1 - 0.525 a_{cy}) (1 + 0.6 a_{sl}) \left(1 - 0.052 \max \left(1.5; \min \left(10; \frac{h}{b_w} \right) \right) \right) (0.2)^{\nu} \left(\frac{\max(0.01; \omega_2)}{\max(0.01; \omega_1)} \min \left(9; \frac{L_s}{h} \right) \right)^{\frac{1}{3}} f_c^{0.2} 25^{\left(\frac{\alpha \rho_w f_{yw}}{f_c} \right)} 1.225^{100 \rho_d}$$

α_{st}^{hbw} :

0.022 for hot-rolled or Tempcore bars;
0.0095 for brittle cold-worked bars;

α_{cy} :

= 1 for cyclic loading,
= 0 for monotonic;

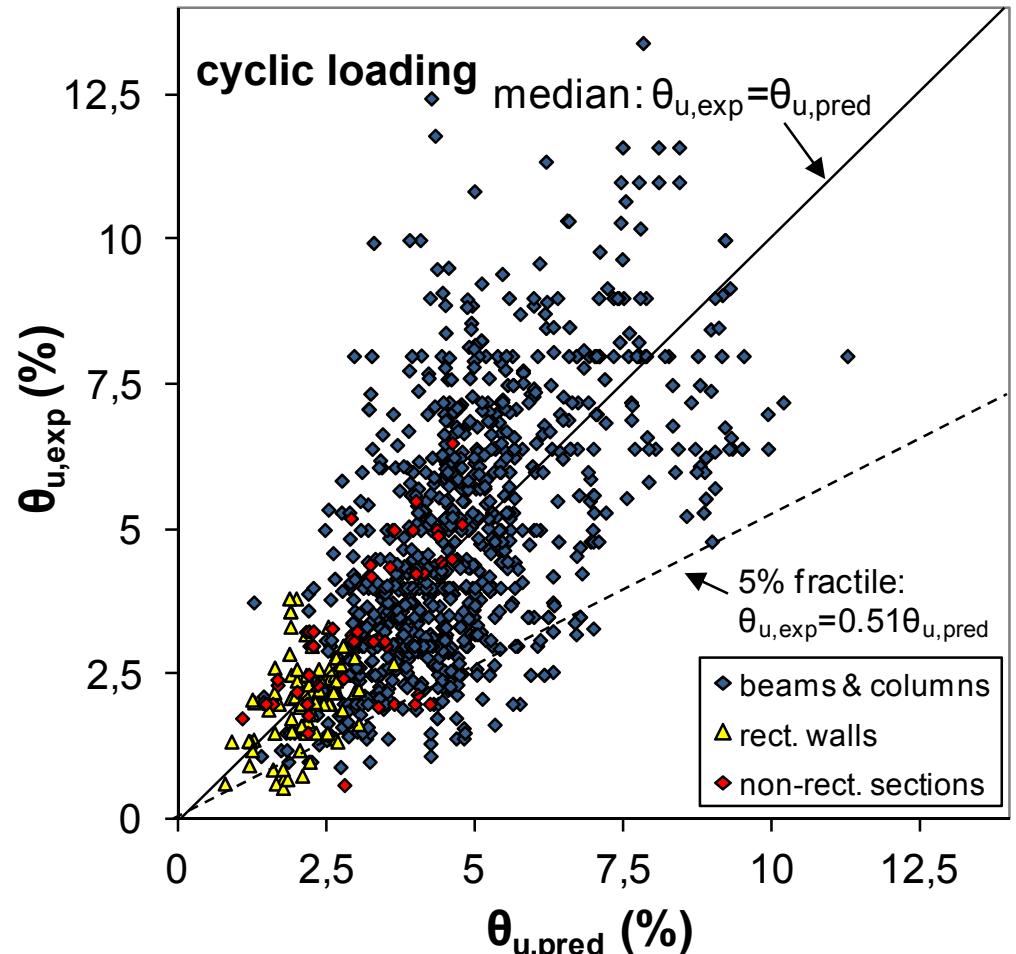
α_{sl} :

= 1 if slippage of long. bars from anchorage zone is possible,
= 0 otherwise;

b_w : width of (one) web

Non-seismically detailed members:

- $\theta_{um}^{pl} = \theta_{um} - \theta_y$ divided by 1.2.



no. tests: 1100, median=1.00, CoV=37.6%

**Effect of lap-splicing the column
bars in the plastic hinge region**

Members with ribbed bars lap-spliced over length l_o inside the plastic hinge region per EN 1998-3:2005 or other options

- **EN 1998-3:2005:**

1. Both bars in pair of lapped compression bars count as compression steel.
2. For the yield properties (M_y , ϕ_y , θ_y), the stress f_s of tension bars is:

$$f_s = f_y(l_o/l_{oy,min}), \text{ if } l_o \leq l_{oy,min} = (0.3f_y/\sqrt{f_c})d_b \quad (f_y, f_c \text{ in MPa})$$

3. Ultimate chord rotation

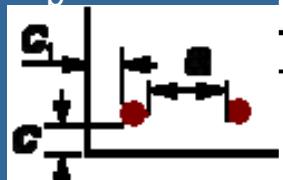
$$\theta_u = \theta_y + \theta_{pl}^u(l_o/l_{ou,min}), \text{ if } l_o \leq l_{ou,min} = d_b f_y / [(1.05 + 14.5\alpha_{rs}\omega_{sx})\sqrt{f_c}]$$

- f_y, f_c in MPa, $\omega_{sx} = p_{sx} f_{yw} / f_c$: mech. transverse steel ratio // loading,
- $\alpha_{rs} = (1 - s_h / 2b_o)(1 - s_h / 2b_o) n_{restr} / n_{tot}$ (n_{restr} / n_{tot} : restrained-to-total lap-spliced bars).

- Or, Eligehausen & Lettow 2007 for **fib MC2010**:

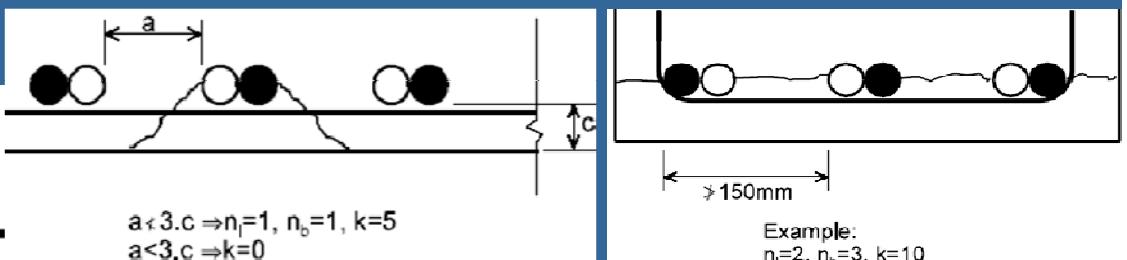
$$f_s = 51.2 \left(\frac{l_b}{d_b} \right)^{0.55} \left(\frac{f_c}{20} \right)^{0.25} \left(\frac{20}{\max(d_b; 20mm)} \right)^{0.2} \left[\left(\frac{c_d}{d_b} \right)^{1/3} \left(\frac{c_{max}}{c_d} \right)^{0.1} + k K_{tr} \right] \leq f_y, k K_{tr} = \frac{1}{n_b d_b} \left(\frac{k_s n_l A_{sh}}{s_h} \right)$$

- $c_d = \min[a/2; c_1; c] \geq d_b, c_d \leq 3d_b$



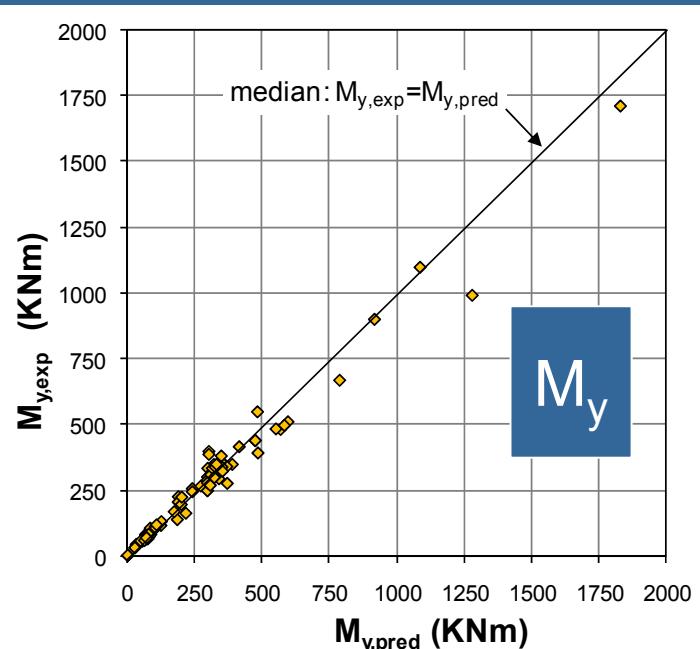
- $c_{max} = \max[a/2; c_1; c] \leq 5d_b$

- f_y, f_c in MPa

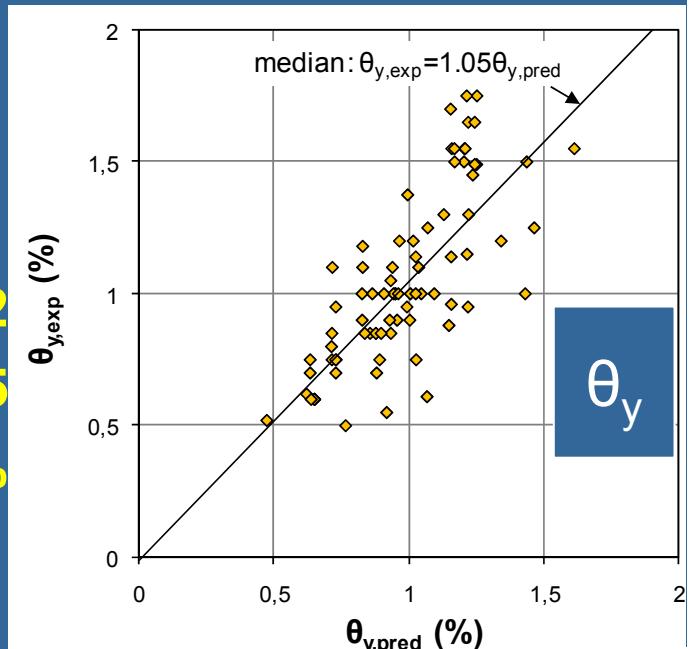


Test-model comparison: Yield moment & chord rotation, effective stiffness, ultimate cyclic chord rotation - lapped ribbed bars per EN 1998-3:2005

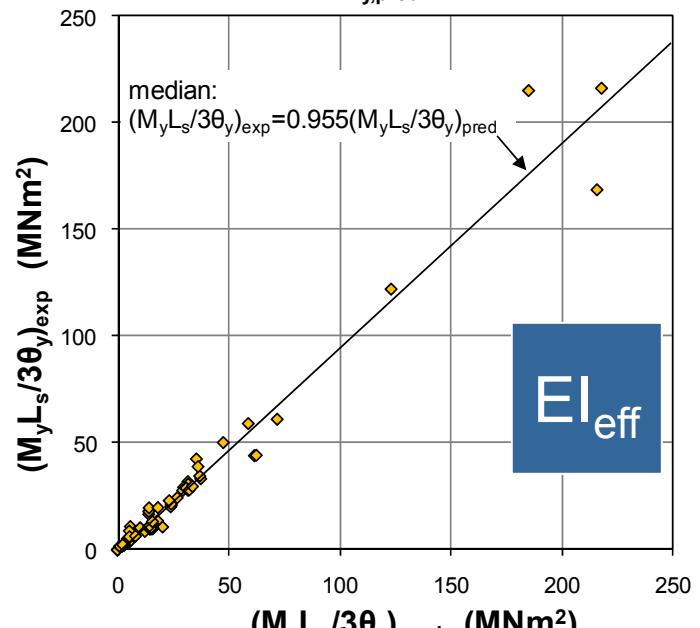
no.tests: 114
median=1.00
CoV=11.8%



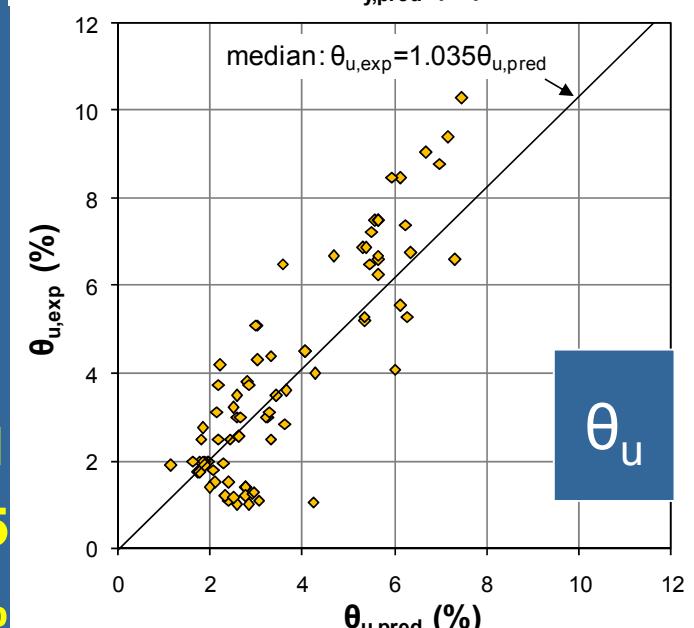
no.tests: 92
median=1.05
CoV=20%



no.tests: 92
median=0.955
CoV=24.8%

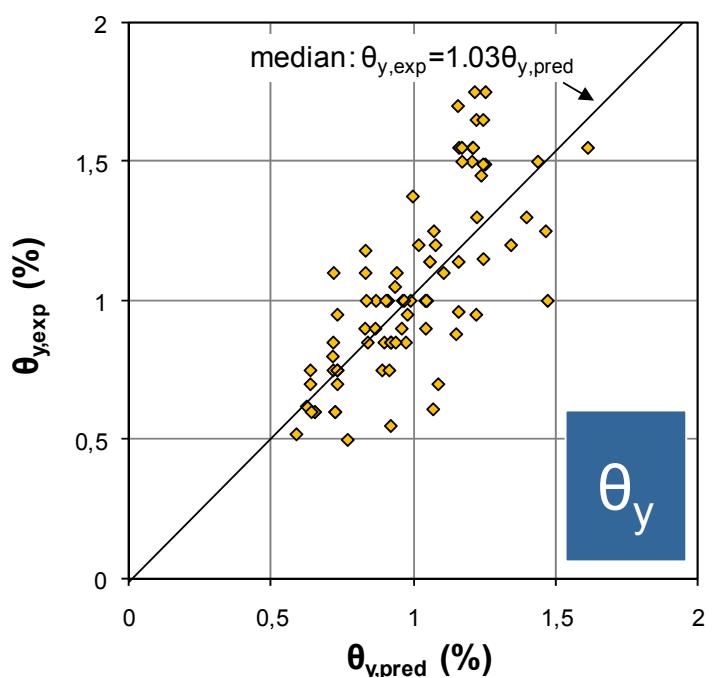


no.tests: 81
median=1.035
CoV=39.3%

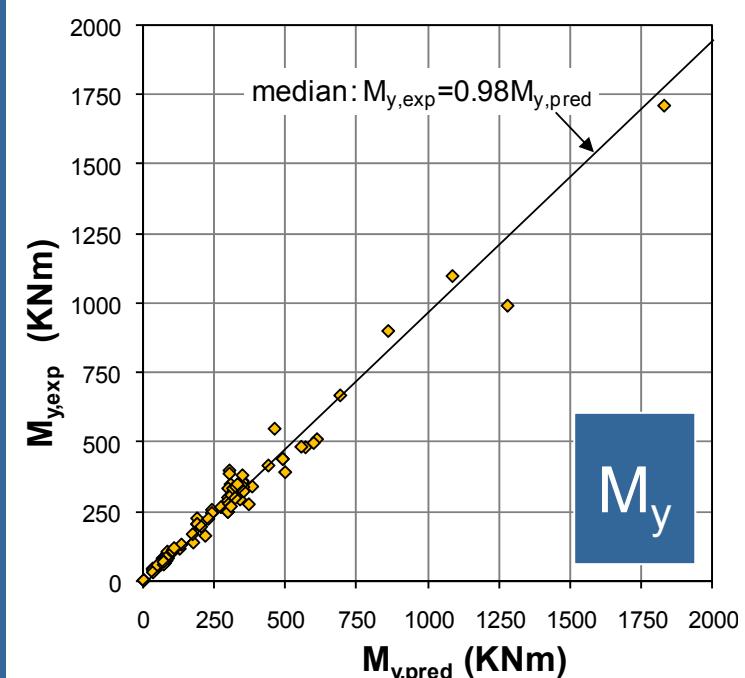


Test-model comparison: Yield moment & chord rotation, effective stiffness, - lapped bars, steel stress per Eligehausen & Letow 2007

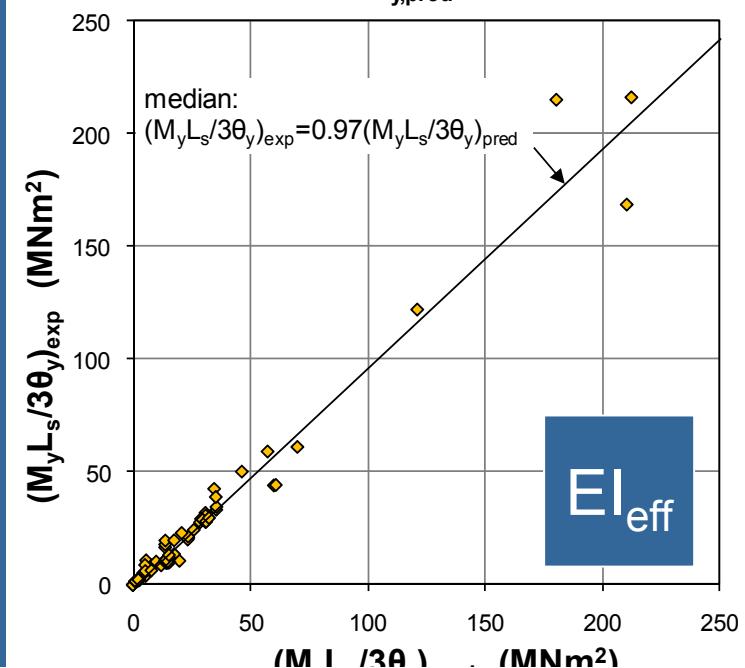
no.tests: 92
median=1.03
CoV=20.5%



no.tests: 114
Median=0.98
CoV=12%



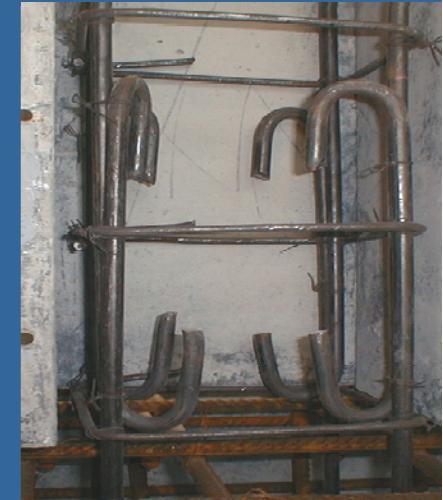
no.tests: 92
median=0.97
CoV=24.6%



Members with smooth hooked bars, with or without lap splice (of length l_o) in the plastic hinge per EN 1998-3:2005

Provided that lapping $l_o \geq 15d_b$:

1. Yield properties M_y , ϕ_y , θ_y :
 - As in members with continuous ribbed bars.
2. Ultimate cyclic chord rotation:
 - For continuous bars:
 - $0.8\theta_{um}$ (: ~0.95 for smooth bars /1.2 for no seismic detailing) or
 - $\theta_{um} = \theta_y + 0.75\theta_{um}^{pl}$ (: ~0.9 for smooth bars /1.2 for no seismic detailing)
 - For lapping $l_o \geq 15d_b$:
 - $\theta_{um} \times 0.019(10 + \min(40, l_o/d_b))$ or
 - $\theta_{um} = \theta_y + \theta_{um}^{pl} \times 0.019 \min(40, l_o/d_b)$
 - L_s for θ_{um}^{pl} not reduced by l_o



Test-model-ratio for θ_{um}

	no laps	laps
Median	1.015	1.03
C.o.V.	33.3%	33.4%
no. tests	34	11

(ultimate condition not necessarily controlled by region right above the lap)

Cyclic shear resistance of RC members

Cyclic shear resistance per EN 1998-3:2005

- Shear resistance in pl. hinge after flex. yielding, as controlled by stirrups
(linear decay of V_c & V_w with cyclic plastic rotation ductility ratio $\mu_\theta^{pl} = (\theta - \theta_y)/\theta_y > 0$)

$$V_R = \frac{h-x}{2L_s} \min(N; 0.55 A_c f_c) + (1 - 0.05 \min(5; \mu_\theta^{pl})) \left[0.16 \max(0.5; 100 \rho_{tot}) \left(1 - 0.16 \min\left(5; \frac{L_s}{h}\right) \right) \sqrt{f_c} A_c + V_w \right]$$

$V_w = \rho_w b_w z f_{yw}$, due to stirrups (b_w : web width, z : internal lever arm; ρ_w : shear reinf. ratio)

ρ_{tot} : total longitudinal reinforcement ratio

h : section depth

x : depth of compression zone at yielding

$$A_c = b_w d$$

- Shear resistance as controlled by web crushing (diagonal compression)

– **Walls**, before flexural yielding ($\mu_\theta^{pl} = 0$) or after (cyclic $\mu_\theta^{pl} > 0$):

$$V_R = 0.85 \left(1 - 0.06 \min(5, \mu_\Delta^{pl}) \right) \left(1 + 1.8 \min\left(0.15, \frac{N}{A_c f_c}\right) \right) \left(1 + 0.25 \max(1.75, 100 \rho_{tot}) \right) \left(1 - 0.2 \min\left(2, \frac{L_s}{h}\right) \right) \sqrt{f_c} b_w z$$

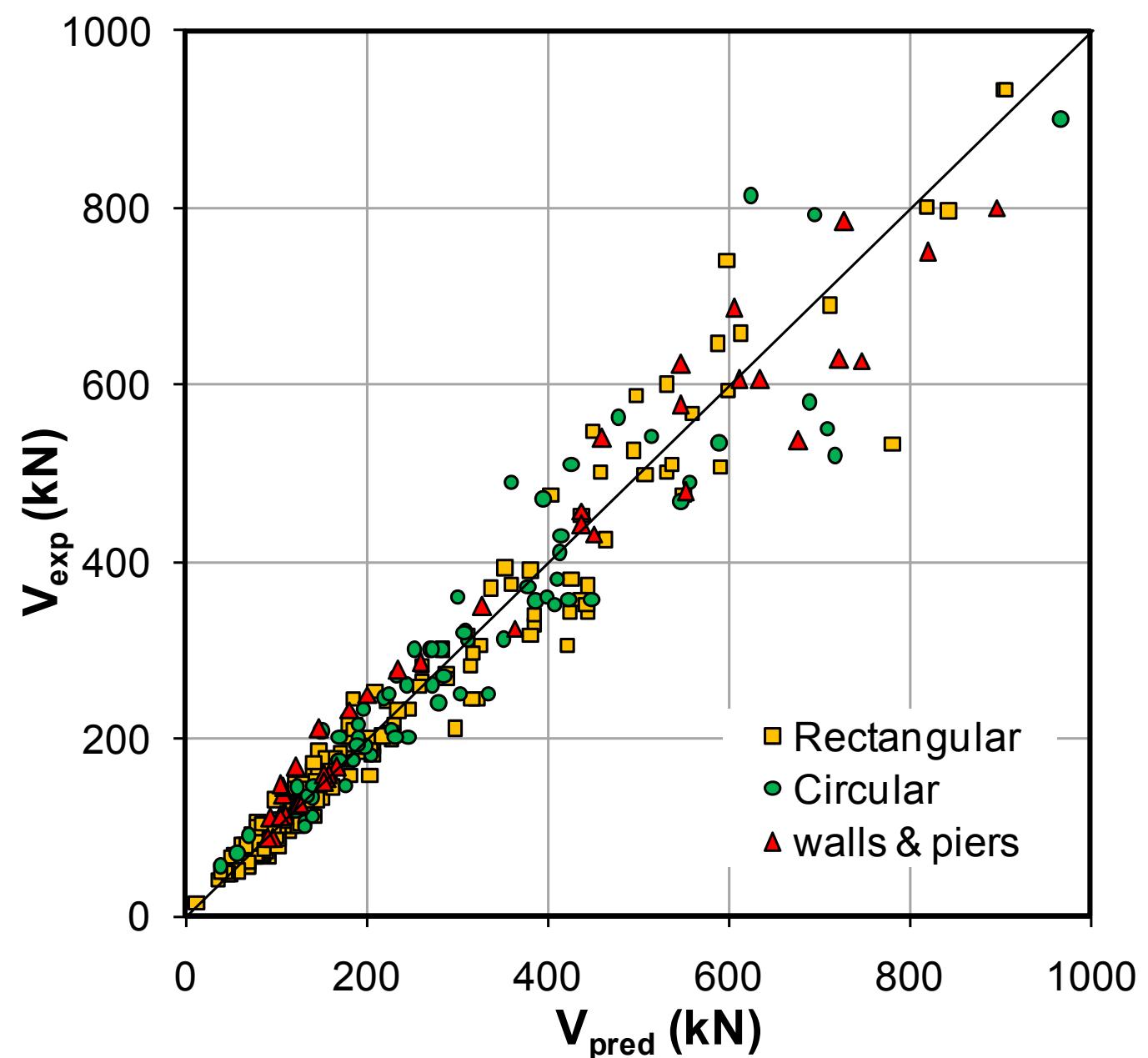
– **Squat columns** ($L_s/h \leq 2$) after flexural yielding (cyclic $\mu_\theta^{pl} > 0$):

$$V_R = \frac{4}{7} \left(1 - 0.02 \min(5, \mu_\Delta^{pl}) \right) \left(1 + 1.35 \frac{N}{A_c f_c} \right) \left(1 + 0.45 \cdot 100 \rho_{tot} \right) \sqrt{\min(f_c, 40)} b_w z \sin 2\delta$$

δ : angle between axis and diagonal of column ($\tan \delta = 0.5h/L_s$)

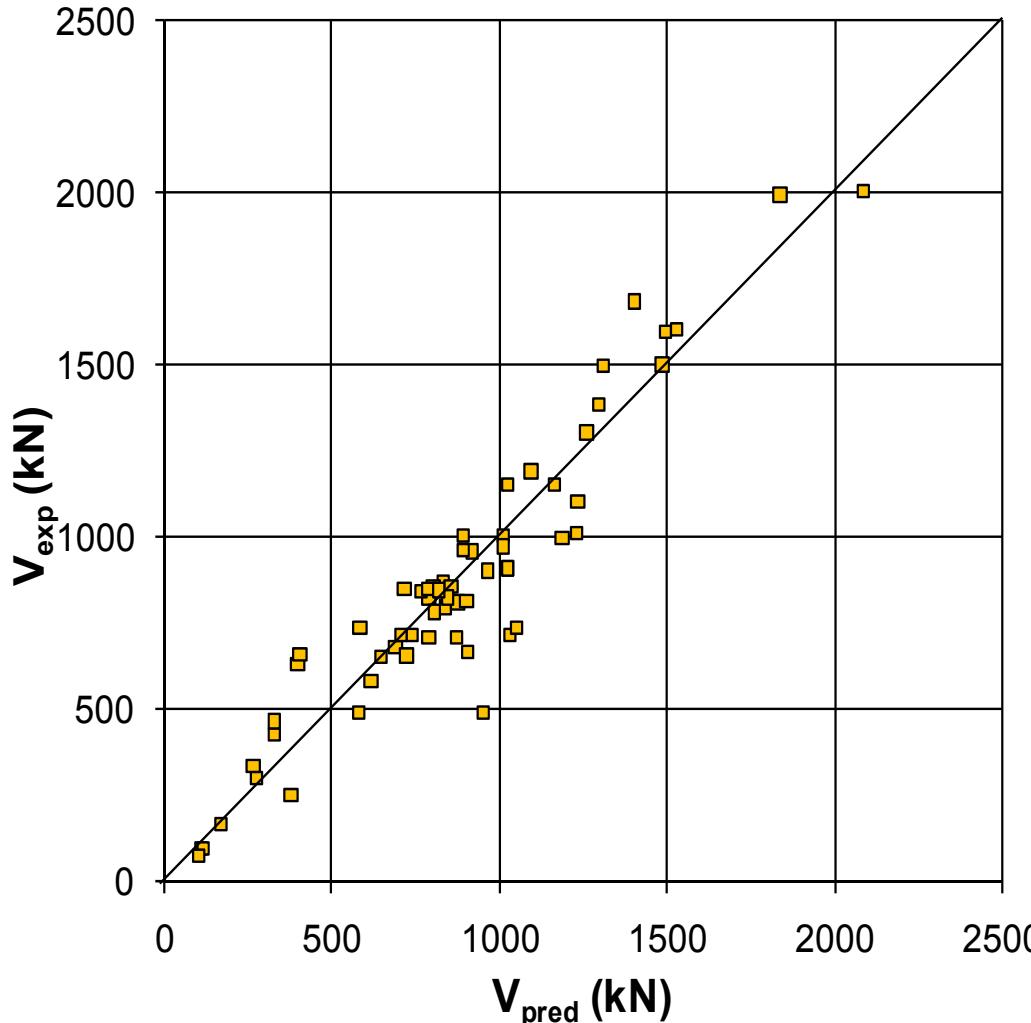
Test-model comparison: Cyclic shear resistance in plastic hinge (after flexural yielding) as controlled by stirrups, per EN 1998-3:2005

no. tests: 306
median=0.995
CoV=14.7%



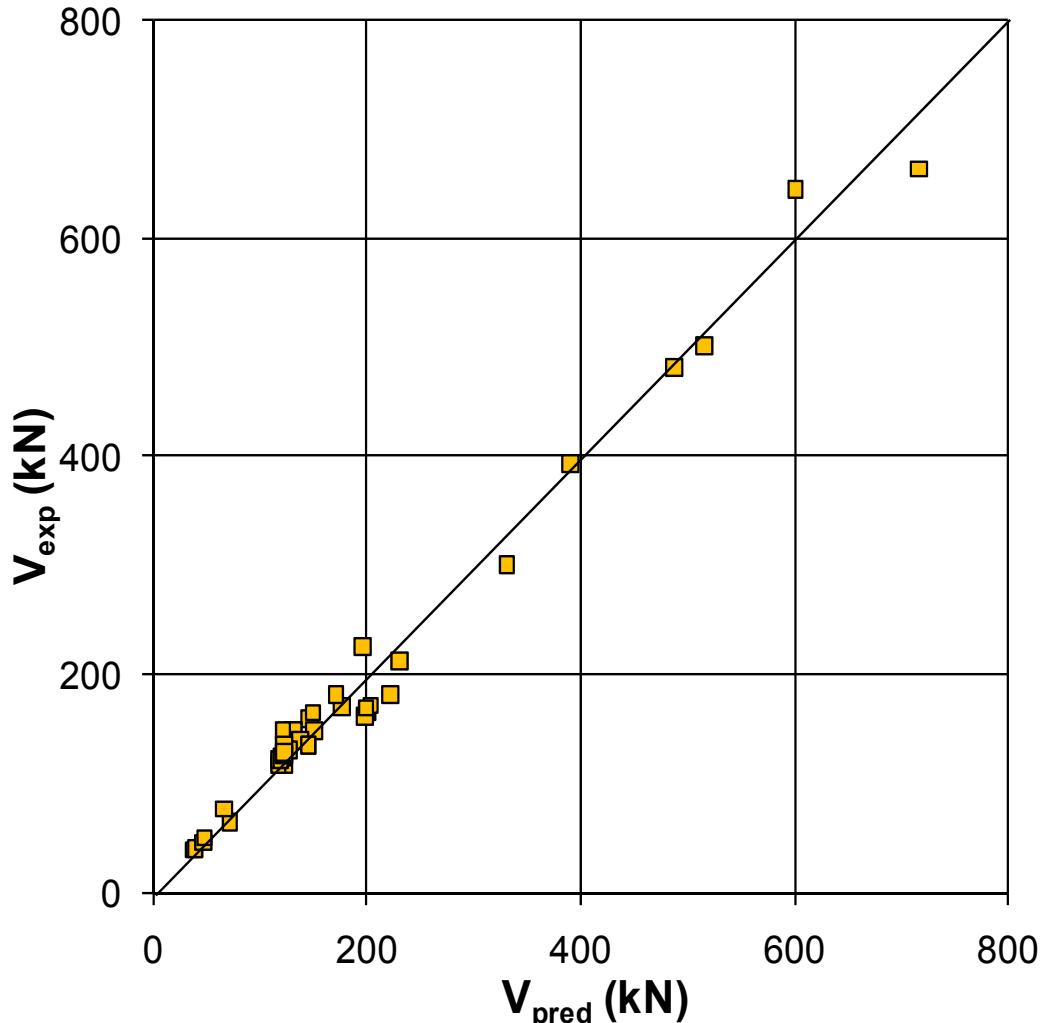
Test-model comparison: Cyclic shear resistance as controlled by web crushing (diagonal compression), per EN 1998-3:2005

Walls



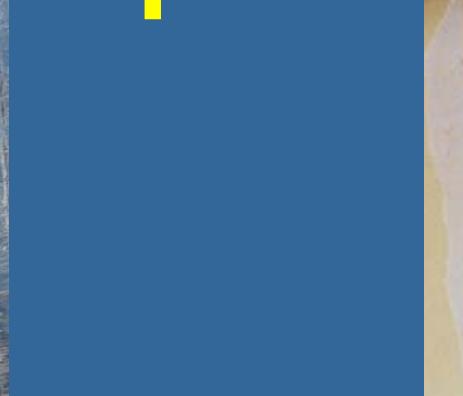
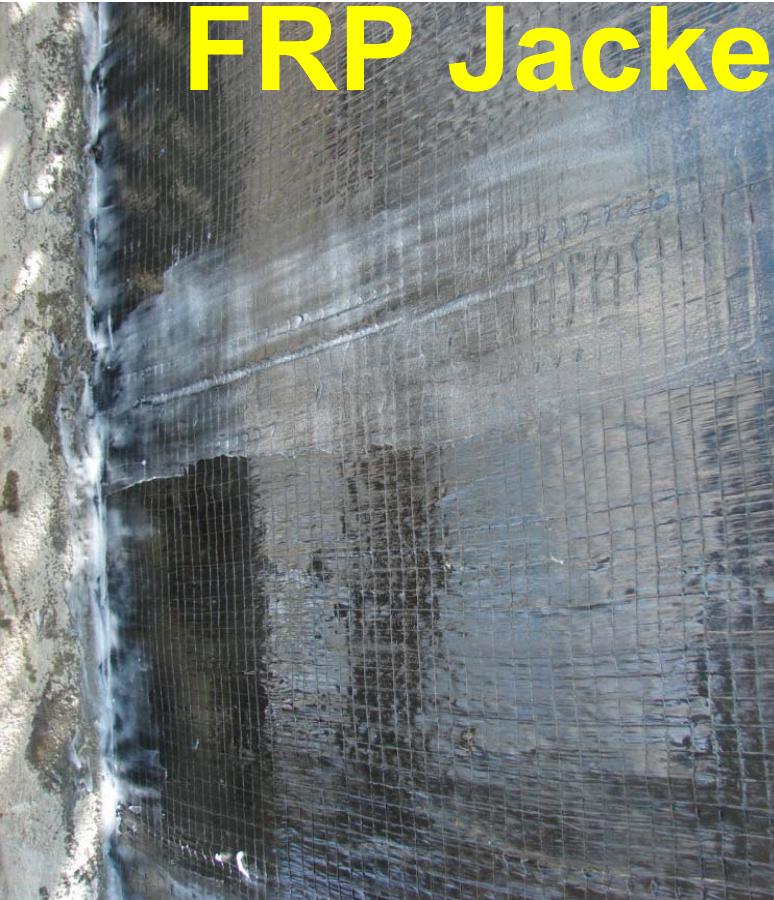
no. tests: 62, median=1.00,
CoV=19.3%

Squat columns



no. tests: 40, median=1.00,
CoV=9.6%

FRP Jackets per EN1998-3:2005



Experimental Database

4. Range and mean values of parameters in the tests of FRP-jacketed rectangular columns in the database

Parameter	219 columns with continuous bars (145 with CFRP, 24 with GFRP, 27 with AFRP and 23 with other composite)		45 columns with lap-spliced bars (42 with CFRP, 3 with GFRP)	
	min-max	mean	min-max	mean
effective depth, d (mm)	170-720	296	180-720	393
shear-span-to-depth ratio, L_s/h	1-7.4	3.55	2-6.6	4.5
concrete strength, f_c (MPa)	10.6-90	31.8	11.7-55	31
vertical bar yield stress, f_y (MPa)	295-816	431	331-617	492
stirrup yield stress, f_{yw} (MPa)	200-750	388	280-535	442
axial-load-ratio, $N/A_c f_c$	0-0.85	0.255	0-0.4	0.143
transverse steel ratio, ρ_w (%)	0-1.18	0.24	0-0.445	0.212
total vertical steel ratio, ρ_{tot} (%)	0.815-7.6	2.08	0.815-3.9	1.88
geometric ratio of FRP, ρ_f (%)	0.01-5.31	0.605	0.13-7.5	1.04
nominal FRP strength (MPa)	113-4830	2755	532-4430	2621
elastic modulus of FRP, E_f (GPa)	5.8-390	166	17.8-390	184
lapping-to-bar-diameter ratio, l_o/d_b	-	-	15-45	30.4

FRP-wrapping of plastic hinges in rectangular members with continuous bars per EN 1998-3:2005

- M_R, M_y : Enhanced by FRP jacket (by 9% w.r.to calculated w/o confinement)
 - EN 1998-3:2005: increase neglected.
- Effective (elastic) stiffness EI_{eff} : unaffected by FRP; pre-damage: 35% drop
- EN1998-3:2005: Flexure-controlled ultimate chord rotation, θ_u :
 - Confinement by FRP increases that due to the stirrups by $\alpha_f \rho_f f_{f,e} / f_c$, where:
 - $\rho_f = 2t_f/b_w$: FRP ratio // direction of loading;
 - $f_{f,e}$: FRP effective strength:

$$f_{f,e} = \min(f_{u,f}, \varepsilon_{u,f} E_f) \min\left(0.5; 1 - 0.7 \frac{\min(f_{u,f}, \varepsilon_{u,f} E_f) \rho_f}{f_c}\right)$$

$f_{u,f}, E_f$: FRP tensile strength & Modulus;

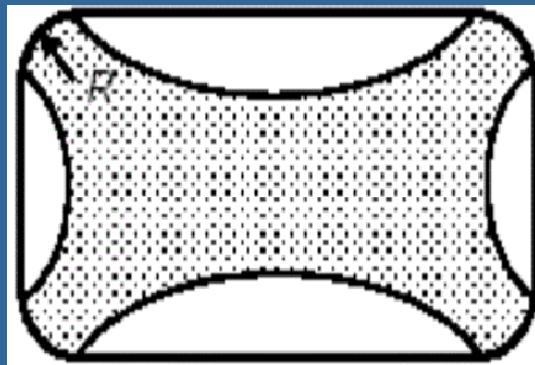
$\varepsilon_{u,f}$: FRP limit strain. CFRP/AFRP: $\varepsilon_{u,f}=1.5\%$; GFRP: $\varepsilon_{u,f}=2\%$

- FRP-confinement effectiveness:

$$\alpha_f = 1 - \frac{(h-2R)^2 + (b-2R)^2}{3bh}$$

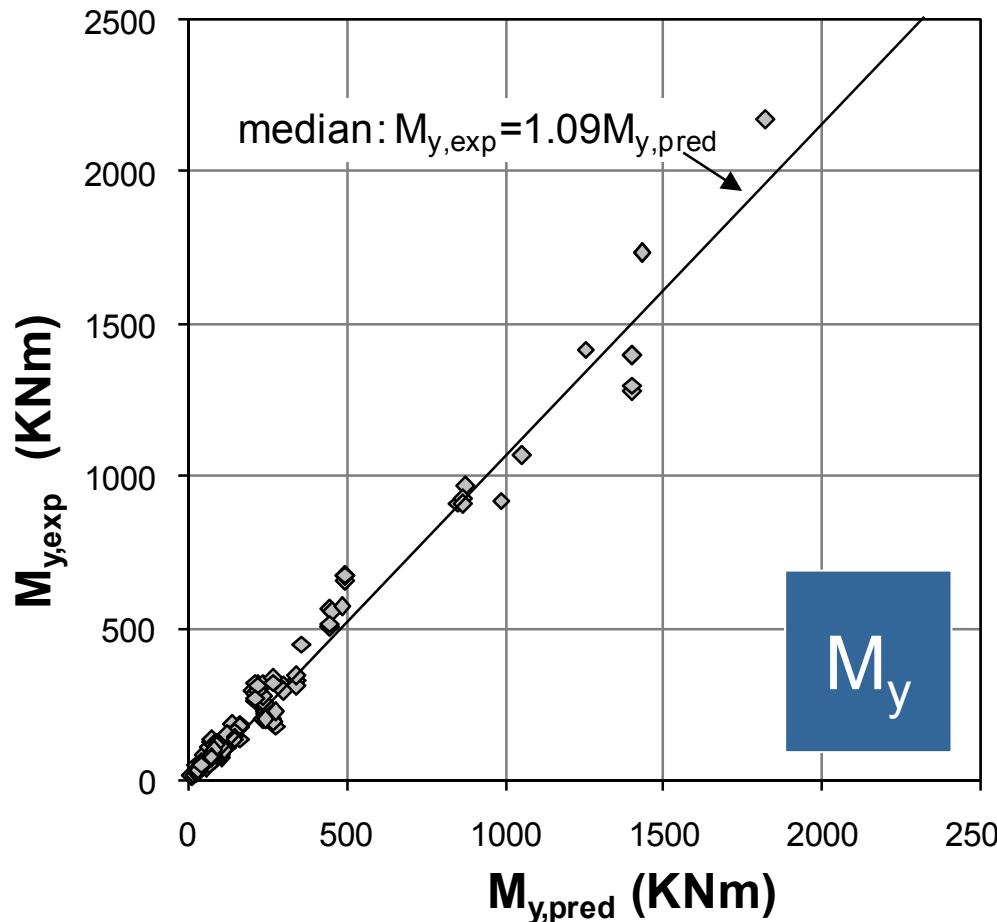
b, h : sides of section;

R : radius at section corner

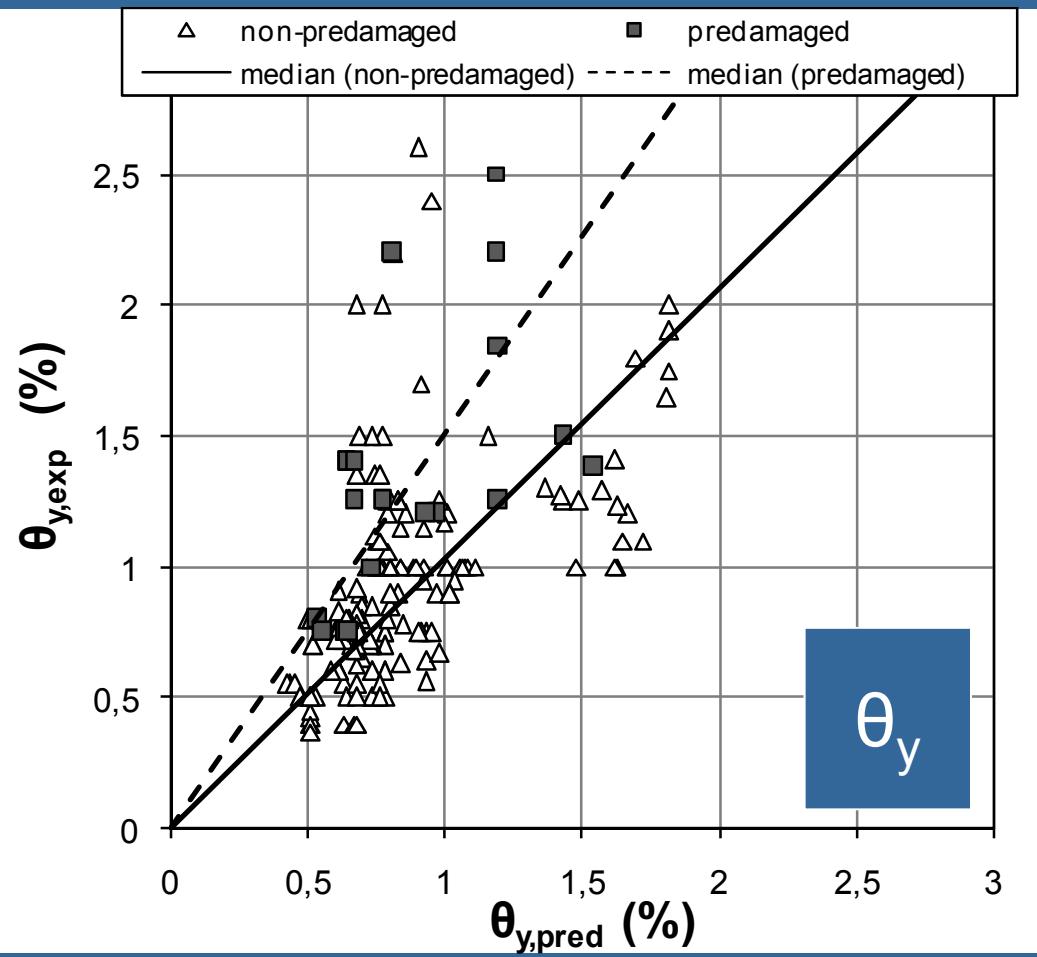


Test-model comparison - yield properties for FRP-wrapping of rectangular columns with continuous bars per EN 1998-3:2005

no.tests: 203 (pre-damaged or not)
median=1.09, CoV=19.4%



no.tests: 159 (no pre-damage)
median=1.03, CoV=37.3%



EI_{eff} (no pre-damage), no.tests: 159, median=1.02, CoV=30%

Yield properties for FRP-wrapping of the plastic hinge and continuous bars – Biskinis & Fardis 2007, 2013

- **Yield moment, M_y :**

- Strength of FRP-confined concrete f_{cc} , instead of f_c , in section-analysis for φ_y , M_y , including the calculation of the concrete Elastic Modulus, E_c ;
 - E_c may be estimated per ***fib*MC2010**, using f_{cc} instead of f_c :

$$E_c = 10000(f_{cc}(\text{MPa}))^{1/3}$$

- f_{cc} from (widely used) Lam & Teng 2003:

$f_{u,f}$: effective strength of FRP: $f_{u,f} = E_f(k_{\text{eff}}\varepsilon_{u,f})$

E_f : Elastic Modulus of FRP,

$\varepsilon_{u,f}$: FRP failure strain,

k_{eff} : FRP effectiveness factor, equal to 0.6 per Lam & Teng

(~same results using other FRP-confinement models: Teng et al 2009, Samaan et al 1998, Bisby et al 2005, Ilki et al 2008, Wang et al 2012)

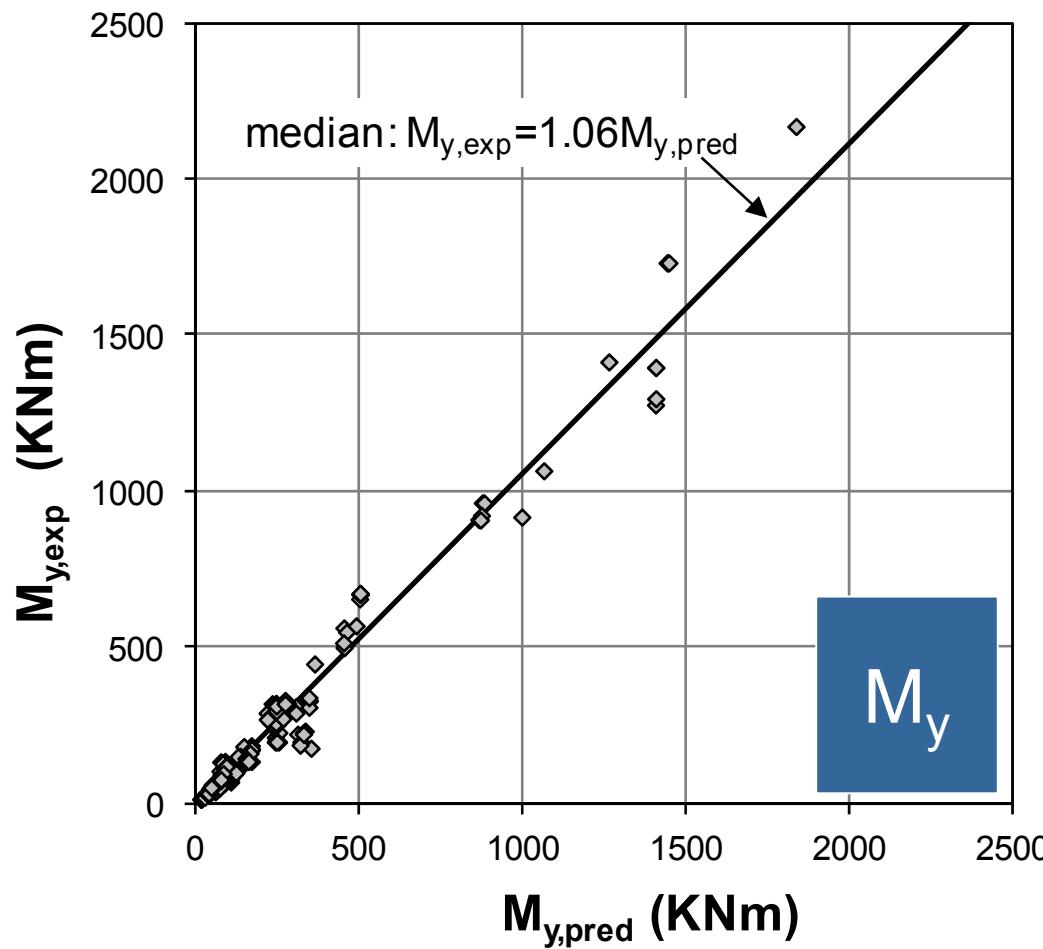
$$\frac{f_{cc}}{f_c} = 1 + 3.3 \left(\frac{b_x}{b_y} \right)^2 \frac{a_f \rho_f f_{u,f}}{f_c}$$

- **Chord rotation at yielding, θ_y , effective stiffness, EI_{eff} :**

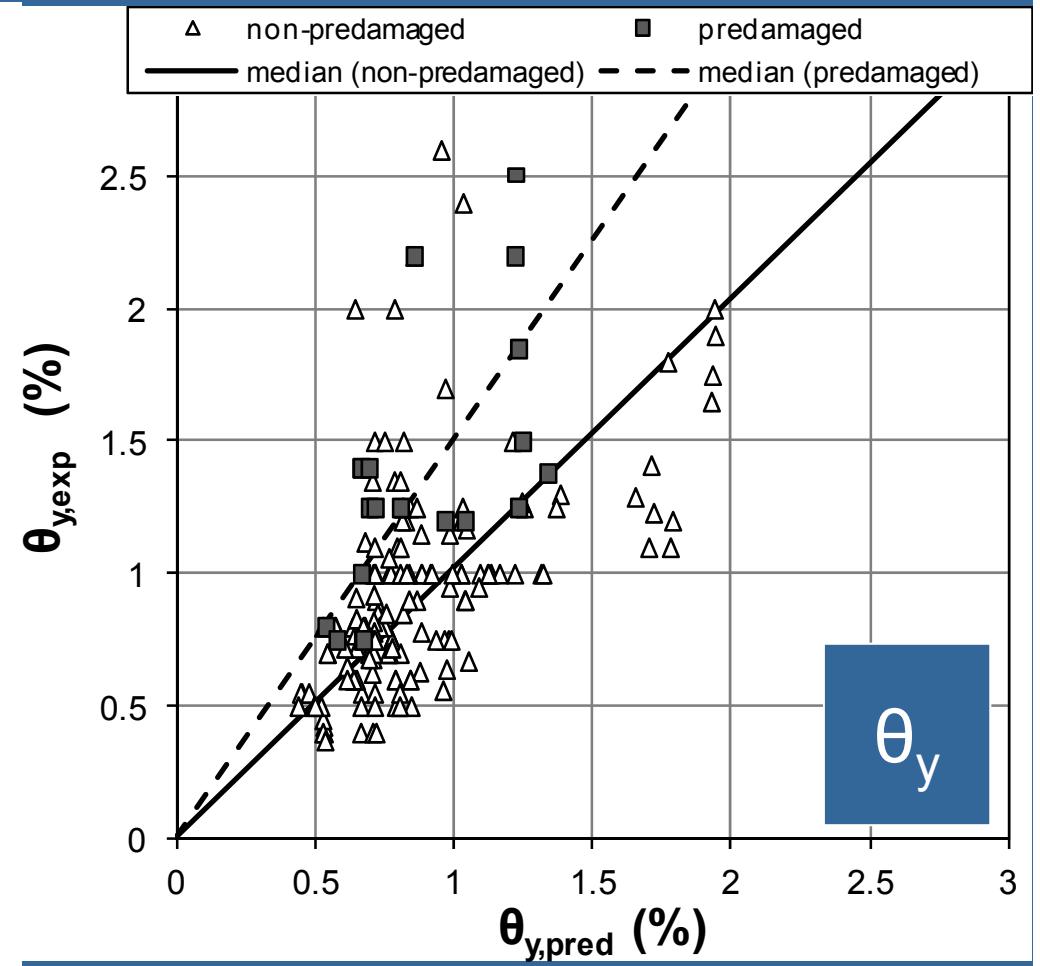
- Yield curvature φ_y calculated with f_{cc} (as above) & multiplied times correction factor 1.06
 - $EI_{\text{eff}} = M_y L_s / 3\theta_y$

Test-model comparison - yield properties for FRP-wrapping of rect. columns with continuous bars per Biskinis & Fardis 2007, 2013

no.tests: 203 (pre-damaged or not)
median=1.06, CoV=19.3%



no.tests: 159 (no pre-damage)
median=1.01, CoV=37.3%



EI_{eff} (no pre-damage): no.tests: 159, median=0.99, CoV=30.4%

Extension of empirical ultimate plastic chord rotation of rect. columns with continuous bars & FRP-wrapping - Biskinis & Fardis 2007, 2013

- Pre-damaged or not:

$$\theta_u^{pl} = 0.0185 \cdot \left(1 - 0.52a_{cy}\right) \left(1 + \frac{a_{sl}}{1.6}\right) (0.25)^v \left(\frac{\max(0.01, \omega')}{\max(0.01, \omega)}\right)^{0.3} f_c^{0.2} \left(\frac{L_s}{h}\right)^{0.35} 25^{\left[\frac{apf_u}{f_c} + \left(\frac{apf_u}{f_c}\right)_{f,eff}\right]} 1.275^{100\rho_d}$$

with:

$$\left(\frac{apf_u}{f_c}\right)_{f,eff} = a_f c_f \min\left[0.4; \frac{\rho_f f_{u,f}}{f_c}\right] \left(1 - 0.5 \min\left[0.4; \frac{\rho_f f_{u,f}}{f_c}\right]\right)$$

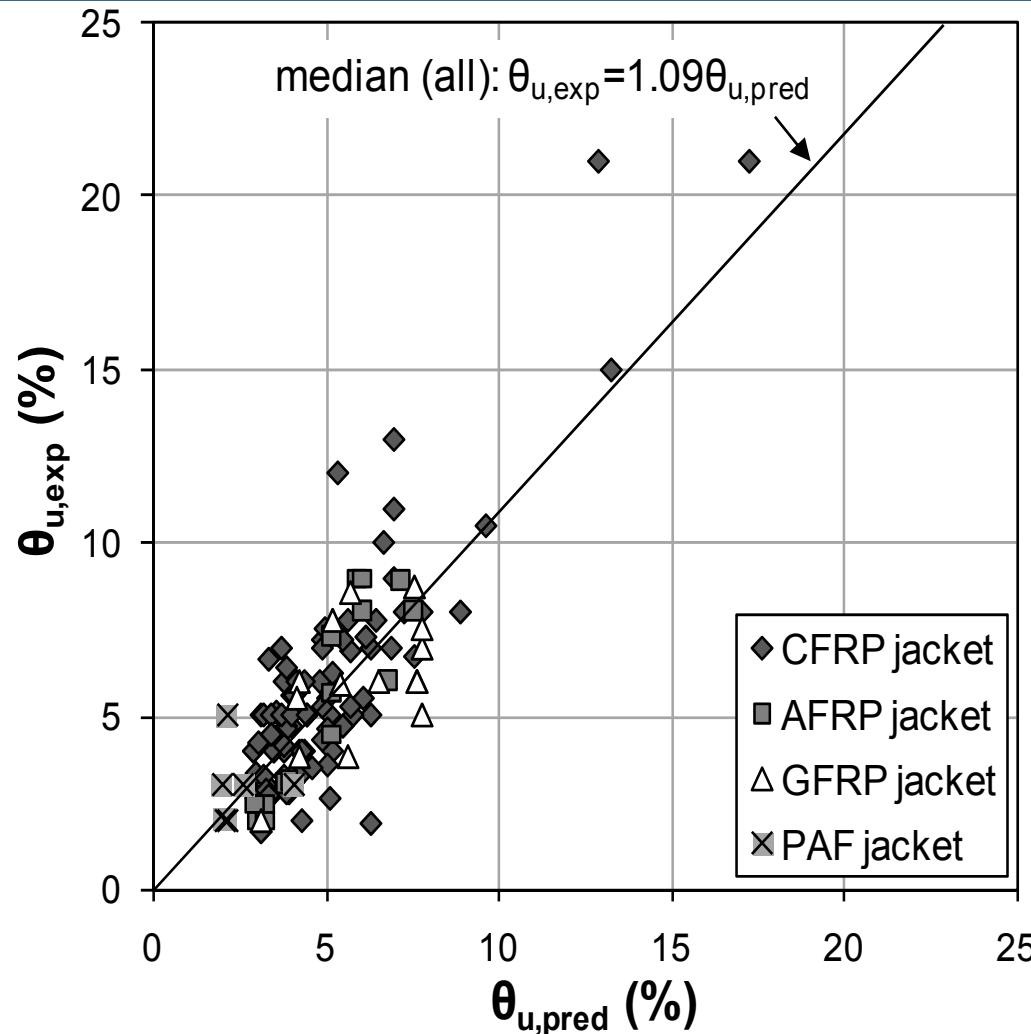
$c_f=1.8$ for CFRP or polyacetal fiber (PAF) sheets,

$c_f=0.8$ for GFRP or AFRP.

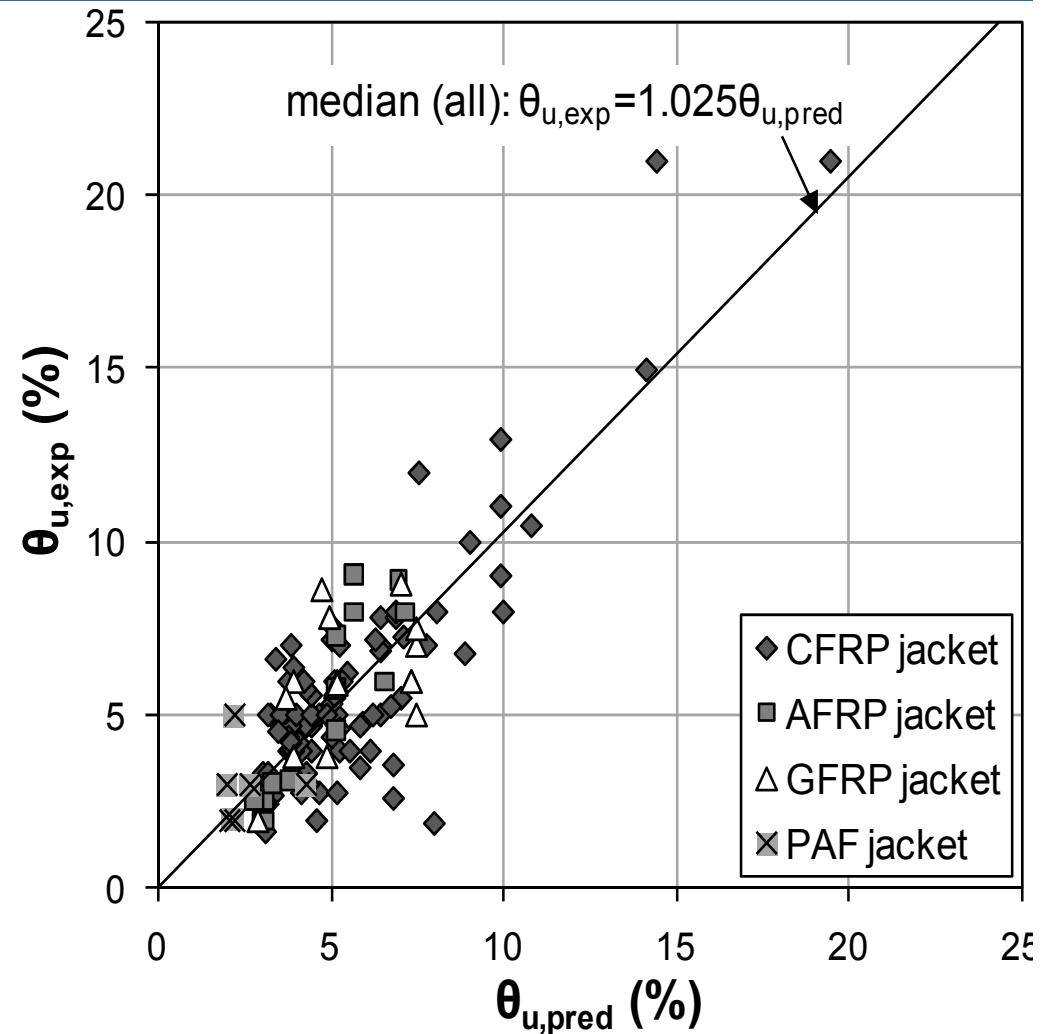
$f_{u,f}=E_f(k_{eff}\varepsilon_{u,f})$: effective FRP strength

Test-model comparison – ultimate cyclic chord rotation for FRP-wrapping of rect. columns with continuous bars: no. tests 128 (pre-damaged or not)
EN 1998-3:2005 v Biskinis & Fardis 2007, 2013

EN 1998-3:2005
median=1.09, CoV=30.6%



Biskinis & Fardis 2013
median=1.025, CoV=30.4%



Alternative ultimate cyclic chord rotation for continuous bars & FRP-wrapping – per GCSI (KANEPE)

$$\frac{f_{cc}}{f_c} = 1.125 + 1.25\alpha\omega_{wd}$$

ω_{wd} : FRP volumetric ratio

α : FRP-confinement effectiveness factor

CFRP (adopted here also for

AFRP and PAF): $\varepsilon_{cu} = 0.0035(f_{cc}/f_c)^2$

GFRP:

$$\varepsilon_{cu} = 0.007(f_{cc}/f_c)^2$$

FRP effective strength: $f_{f,e} = f_{u,f}\psi$

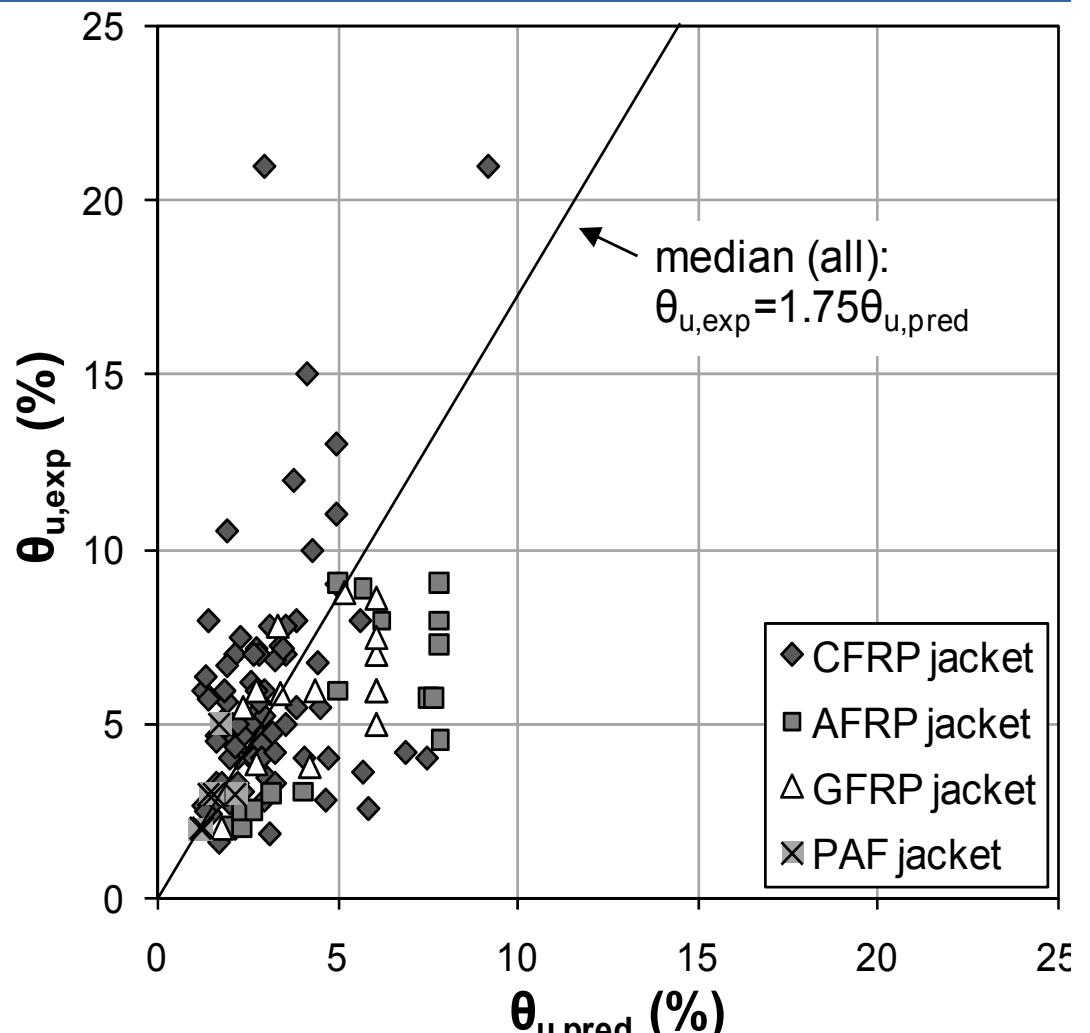
ψ : reduction factor for number of layers (k);

$\psi=1$ for $k<4$

$$\psi = k^{-1/4} \text{ for } k \geq 4$$

$$\mu_\theta = \frac{\theta_u}{\theta_y} \cong \mu_\delta = (\mu_\varphi + 2)/3$$

Test-to-predicted ultimate cyclic chord rotation, θ_u – no.tests: 128, median=1.75, CoV=54.5%



FRP-wrapped rectangular members with ribbed bars lap-spliced over length l_o in the plastic hinge: EN 1998-3:2005 & other options

- **EN 1998-3:2005:**

1. Both bars in pair of lapped compression bars count as compression steel.
2. For the yield properties (M_y , ϕ_y , θ_y), the stress f_s of tension bars is:

$f_s = f_y (l_o / l_{oy,min})$, if $l_o < l_{oy,min} = (0.2f_y / \sqrt{f_c})d_b$ (f_y , f_c in MPa), $l_{oy,min}$: one-third shorter than without FRP.

3. Ultimate chord rotation

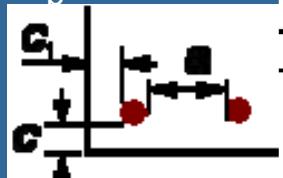
$$\theta_u = \theta_y + \theta_{pl,u} (l_o / l_{ou,min}), \text{ if } l_o < l_{ou,min} = d_b f_y / [(1.05 + 14.5 \alpha_l \rho_f f_{f,e}) \sqrt{f_c}]$$

- f_c : MPa, $\rho_f = 2t_f/b_w$: FRP ratio // loading, $f_{f,e}$: effect. FRP strength (MPa),
- $\alpha_l = \alpha_f (4/n_{tot})$ (n_{tot} : total lap-spliced bars; only the 4 corner bars restrained).

- **Or, Eligehausen & Lettow 2007 for fibMC2010:**

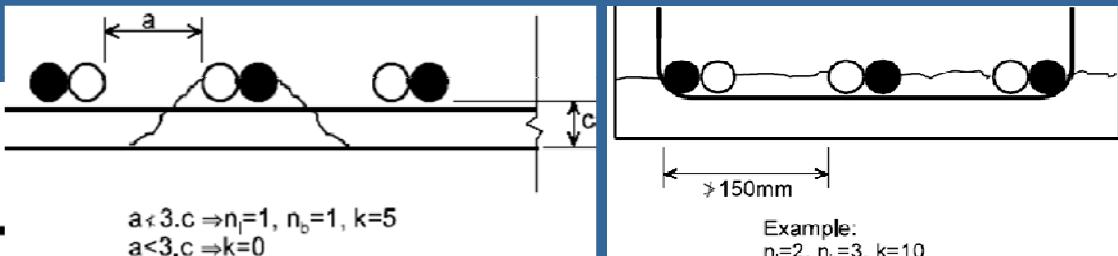
$$f_s = 51.2 \left(\frac{l_b}{d_b} \right)^{0.55} \left(\frac{f_c}{20} \right)^{0.25} \left(\frac{20}{\max(d_b, 20mm)} \right)^{0.2} \left[\left(\frac{c_d}{d_b} \right)^{1/3} \left(\frac{c_{max}}{c_d} \right)^{0.1} + k K_{tr} \right] \leq f_y, \quad k K_{tr} = \frac{1}{n_b d_b} \left(\frac{k_s n_l A_{sh}}{s_h} + \frac{k_f n_f t_f E_f}{E_s} \right)$$

- $c_d = \min[a/2; c_1; c] \geq d_b$, $c_d \leq 3d_b$



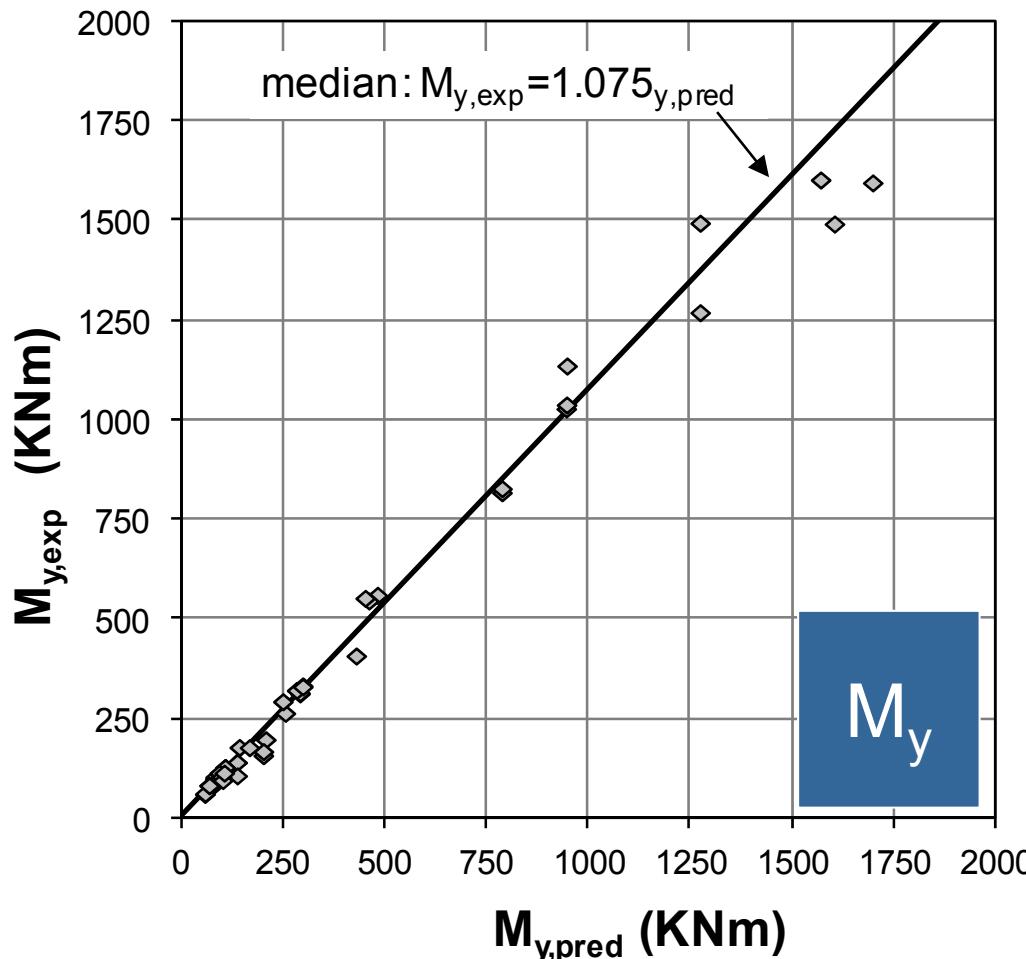
- $c_{max} = \max[a/2; c_1; c] \leq 5d_b$

- f_y, f_c in MPa

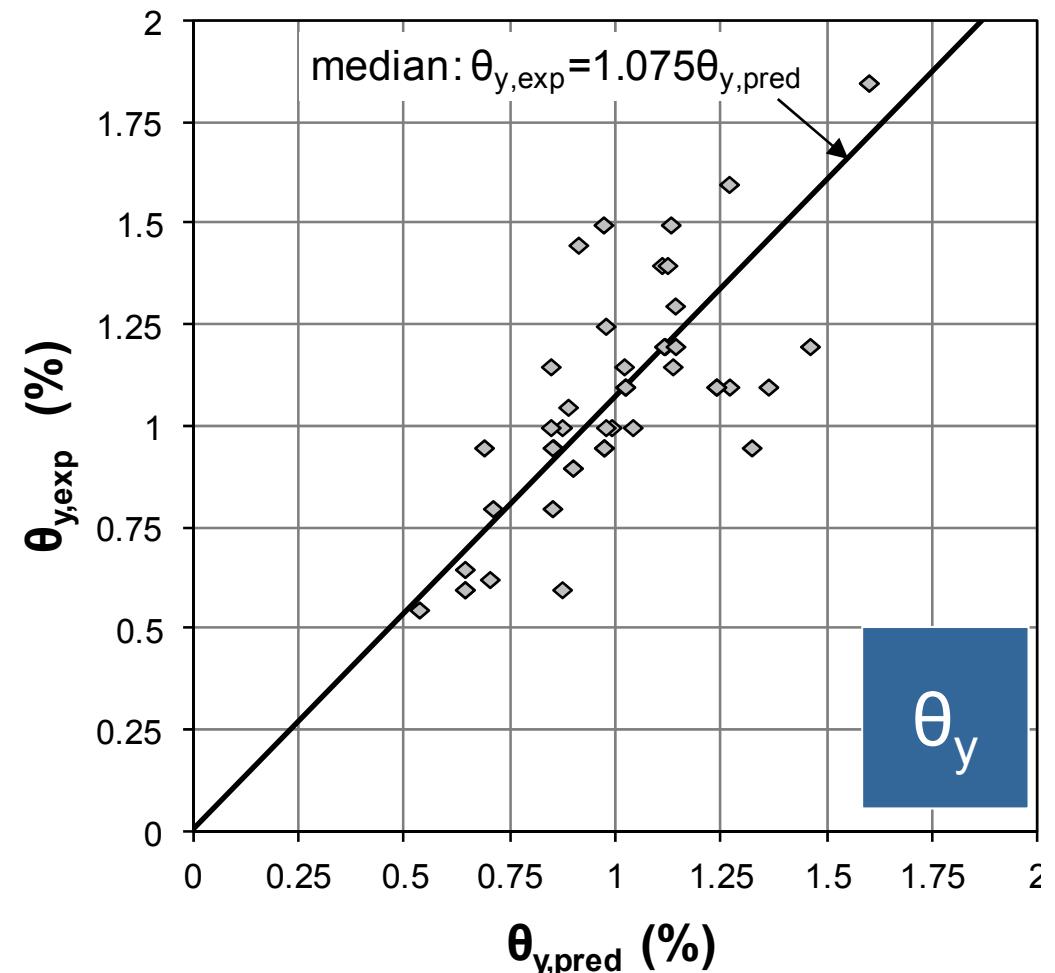


Test-model comparison - Yield properties for FRP-wrapping of rect. columns with bars lap-spliced over length l_o per EN1998-3:2005, no.tests 45

M_y : median=1.075, CoV=10.5%

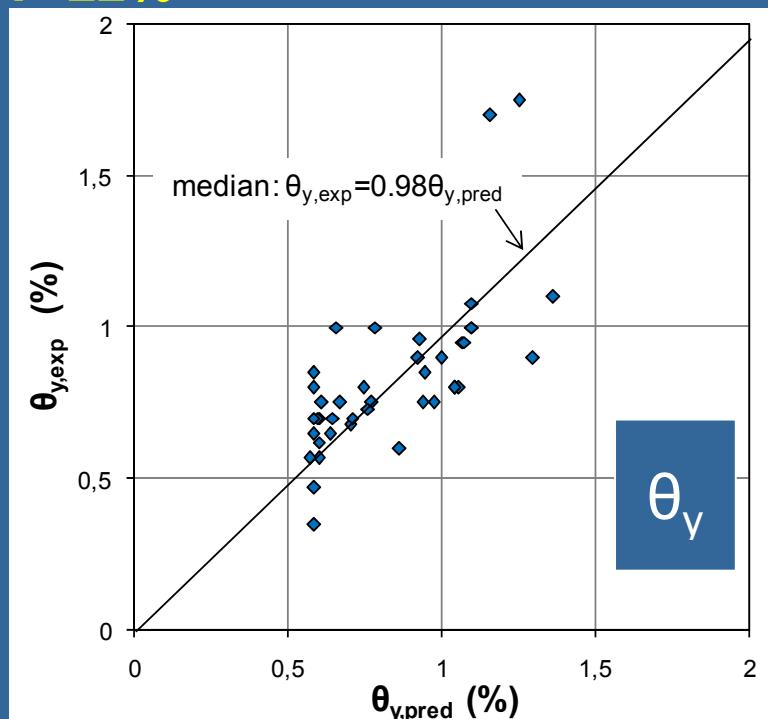


θ_y : median=1.075, CoV=17.7%

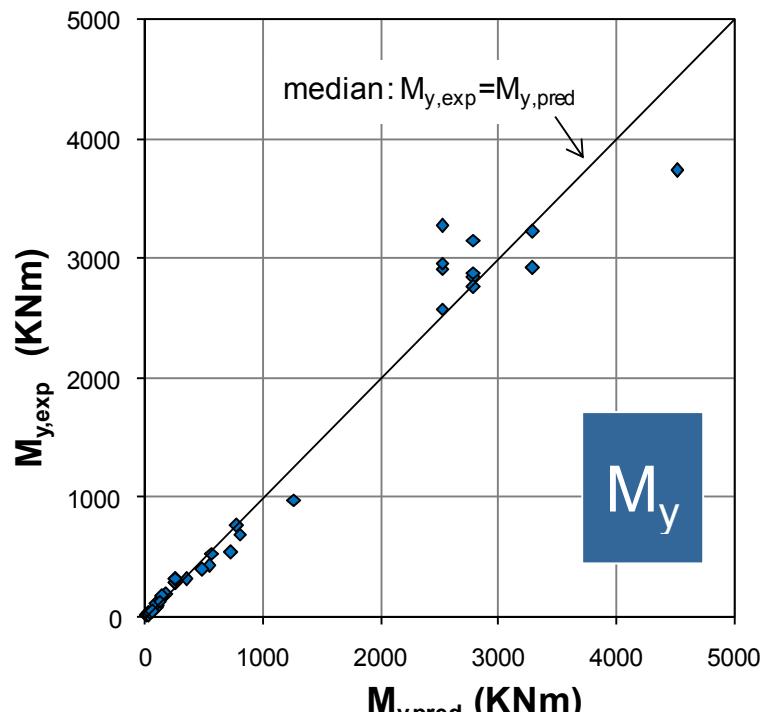


Test-model comparison - yield properties of FRP-wrapped circular columns with lap-spliced bars per EN1998-3:2005 (Biskinis & Fardis 2007, 2013)

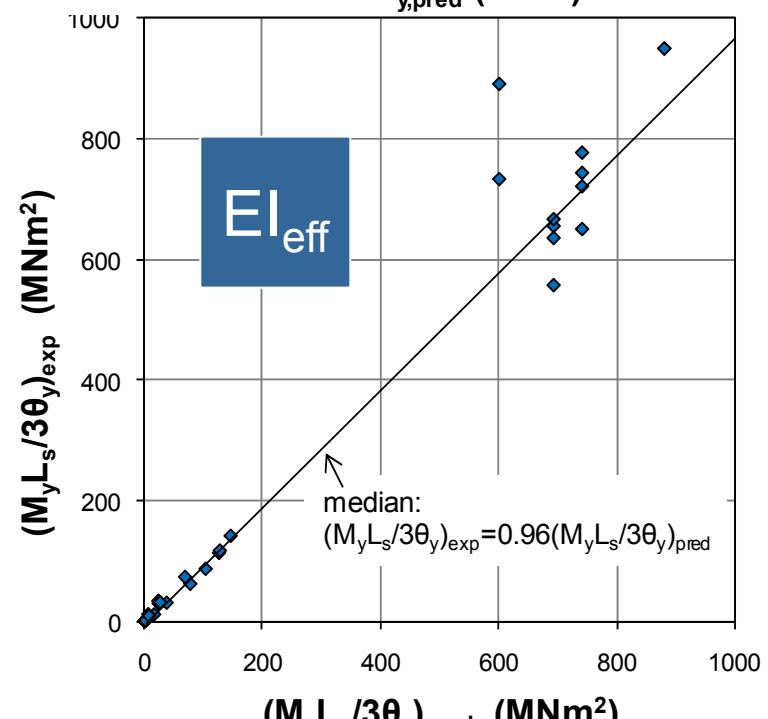
no.tests: 42
median=0.98
CoV=22%



no.tests: 42
median=1.00
CoV=15.2%



no.tests: 42
median=0.96
CoV=22.2%



Extension of empirical ultimate plastic chord rotation for rectangular members with lap-spliced bars & FRP-wrapping – Biskinis & Fardis 2007, 2013

- Required lapping for no adverse effect of lap-splice on ultimate deformation

$$l_{ou,min} = \frac{d_b f_y}{\left(1.05 + 14.5 \left(\frac{2}{n_{tension}} \right)^2 \left(\frac{apf_u}{f_c} \right)_{f,eff} \right) \sqrt{f_c}}$$

with: $\left(\frac{apf_u}{f_c} \right)_{f,eff} = a_f c_f \min \left[0.4; \frac{\rho_f f_{u,f}}{f_c} \right] \left(1 - 0.5 \min \left[0.4; \frac{\rho_f f_{u,f}}{f_c} \right] \right)$

$$f_{u,f} = E_f (k_{eff} \varepsilon_{u,f})$$

$n_{tension}$: number of lapped bars on tension side of section

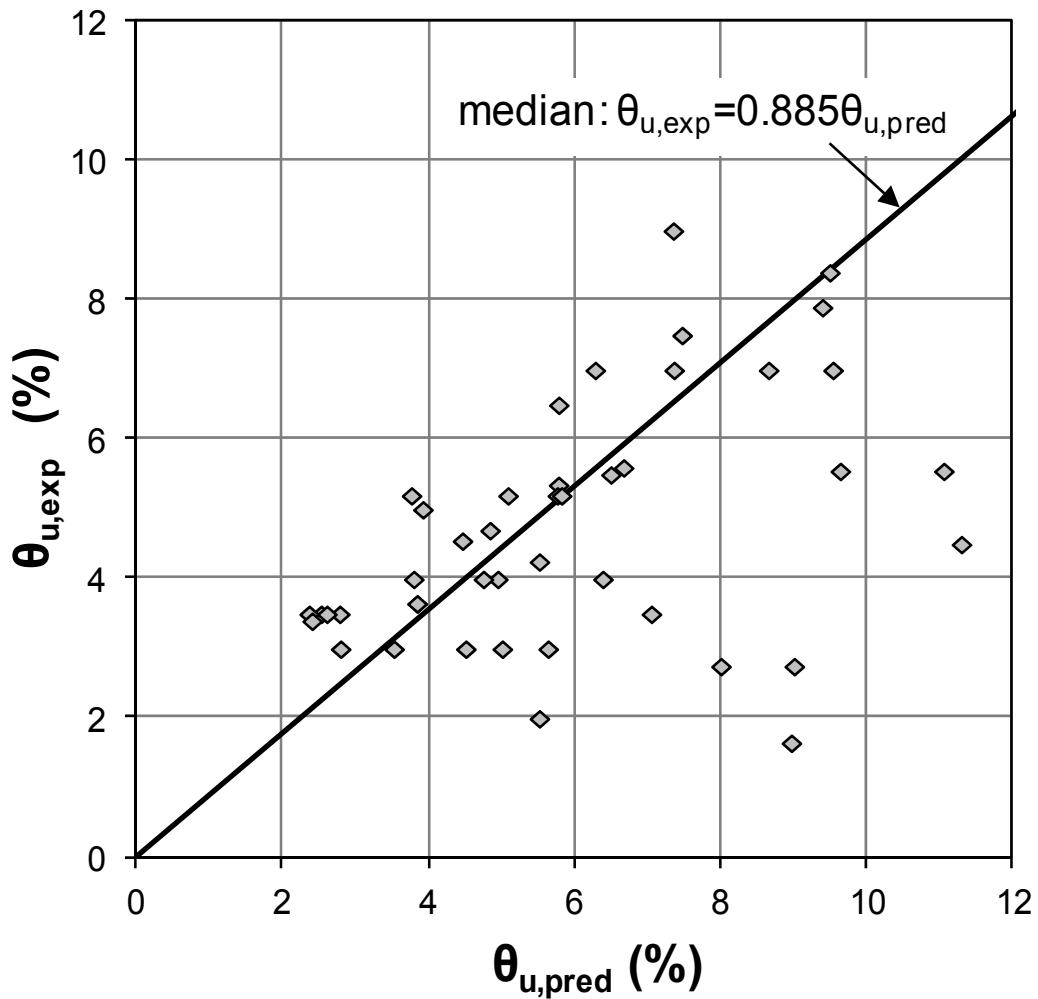
- If $l_o < l_{ou,min}$, θ_u^{pl} is reduced as:
- $(apf_u/f_c)_{f,eff}$ neglected in θ_u^{pl} if $2/n_{tension} \leq 0.5$

$$\theta_{u,lap}^{pl} = \min \left(1, \frac{l_o}{l_{ou,min}} \right) \theta_u^{pl}$$

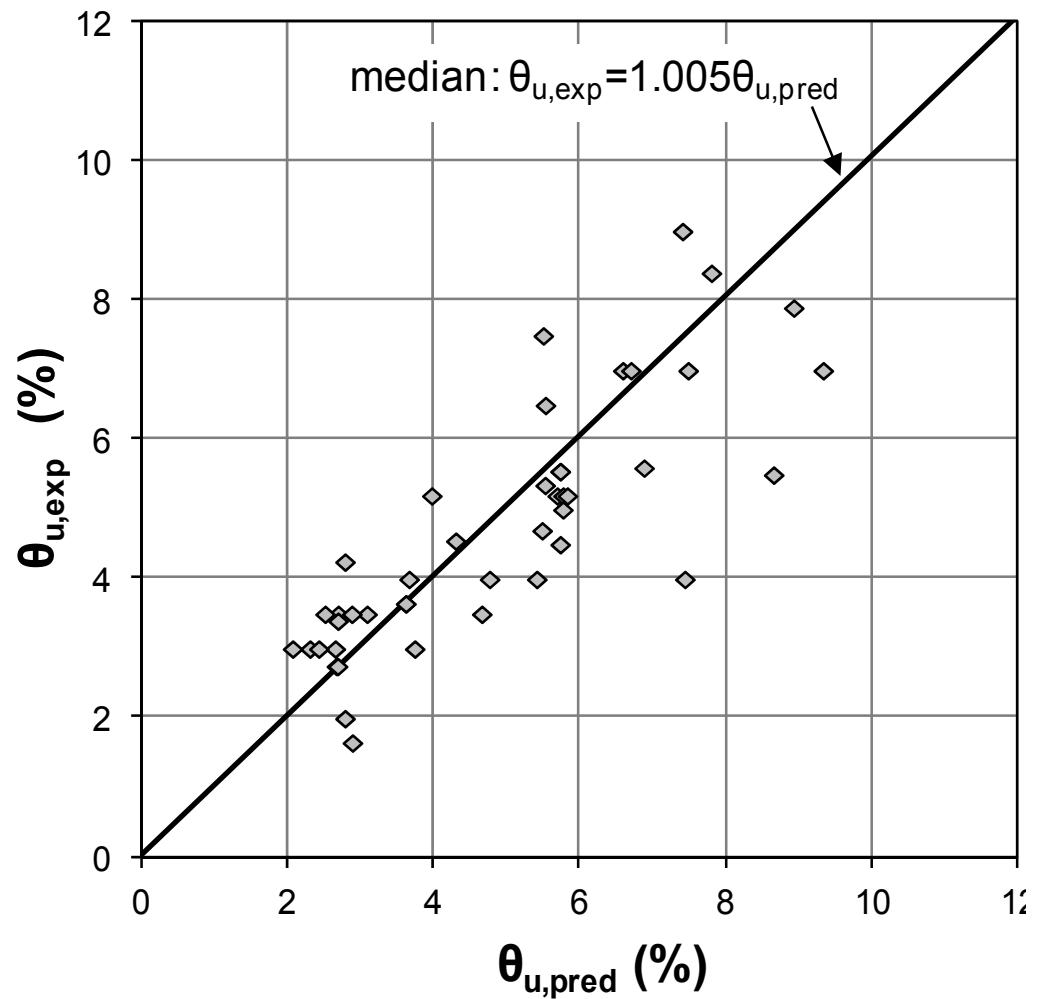
Test-model comparison – ultimate cyclic chord rotation of FRP-wrapped of rectangular columns w/ lap-spliced bars, no.tests 44 (pre-damaged or not)

EN1998-3:2005 v Biskinis & Fardis 2007, 2013

EN 1998-3:2005
median=0.89, CoV=36.3%



Biskinis & Fardis 2013
median=1.005, CoV=23.2%



Extension of ultimate plastic chord rotation model with curvatures and plastic hinge length to circular columns with lap-spliced bars & FRP-wrapping – Biskinis & Fardis 2013

$$\theta_u = \theta_y + (\varphi_u - \varphi_y) L_{pl} \left(1 - 0.5 L_{pl} / L_s \right) + a_{sl} \Delta \theta_{u,slip}$$

$$\Delta \theta_{u,slip} = 5.5 d_{bL} \varphi_u$$

L_{pl} , f_{cc} , ε_{cu} as in members (FRP-wrapped or not) with continuous bars

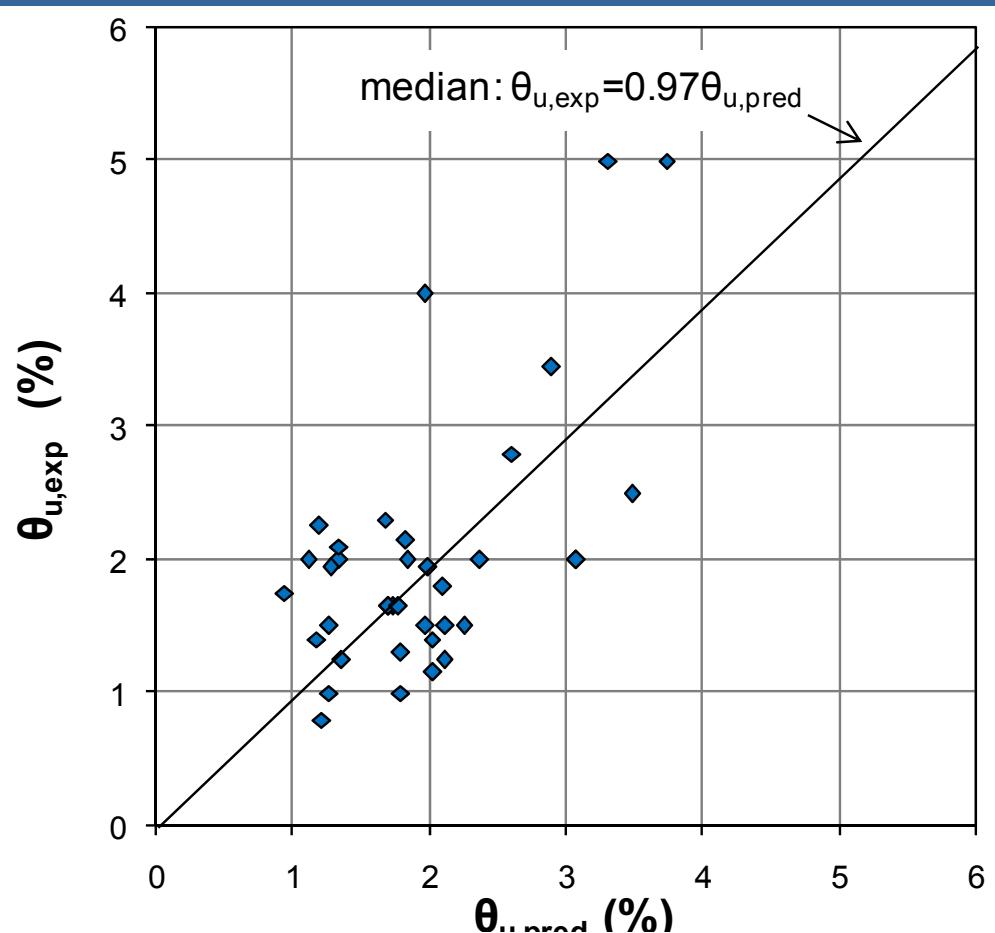
Steel strain at ultimate deformation of member depends on lap length, if $l_o < l_{ou,min}$:

$$\varepsilon_{su,l} = \left[1.2 \frac{l_o}{l_{ou,min}} - 0.2 \right] \varepsilon_{su} \geq \frac{l_o}{l_{oy,min}} \frac{f_y}{E_s}$$

with:

$$l_{ou,min} = \frac{d_{bL} f_y}{\left(5/6 + 6a\rho_{sh} f_{yh} / f_c \right) \sqrt{f_c}}$$

Test-to-prediction ratio, no.tests: 37
median=0.97, CoV=38.6%

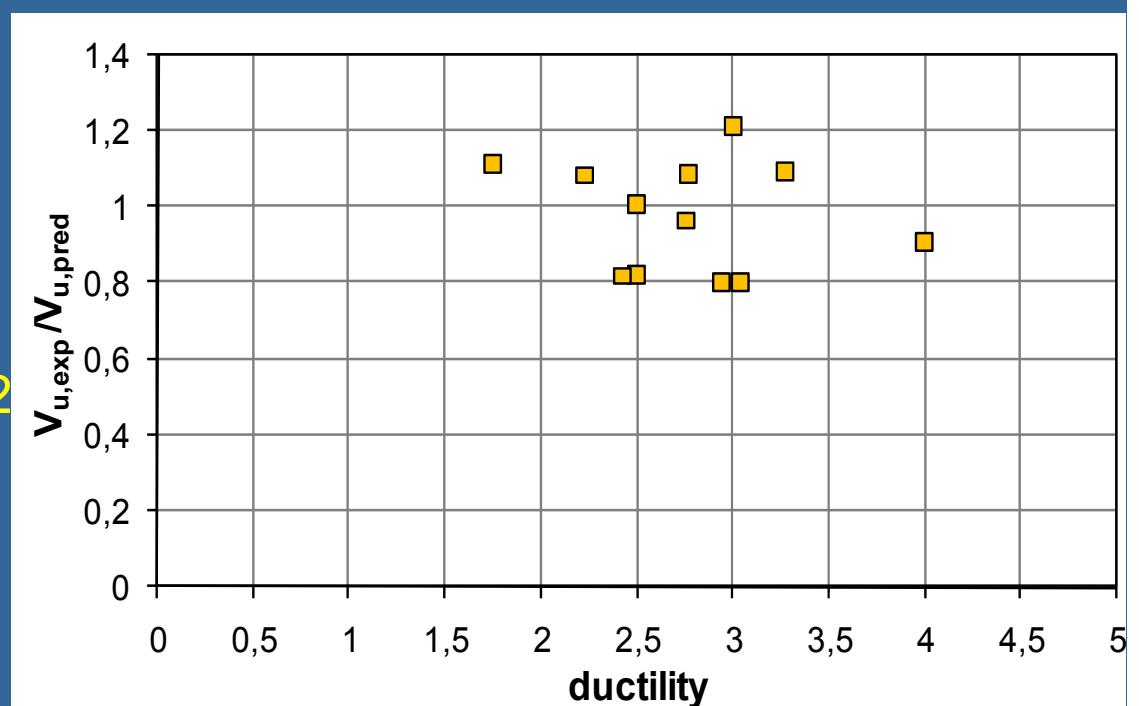


Cyclic shear resistance of FRP-wrapped rectangular members as controlled by diagonal tension, per EN1998-3:2005

$$V_R = \frac{h-x}{2L_s} \min(N, 0.55 A_c f_c) + (1 - 0.05 \min(5, \mu_\theta^{pl})) \left[0.16 \max(0.5, 100 \rho_{tot}) \left(1 - 0.16 \min\left(5, \frac{L_s}{h}\right) \right) \sqrt{f_c} A_c + V_w \right] + V_f$$

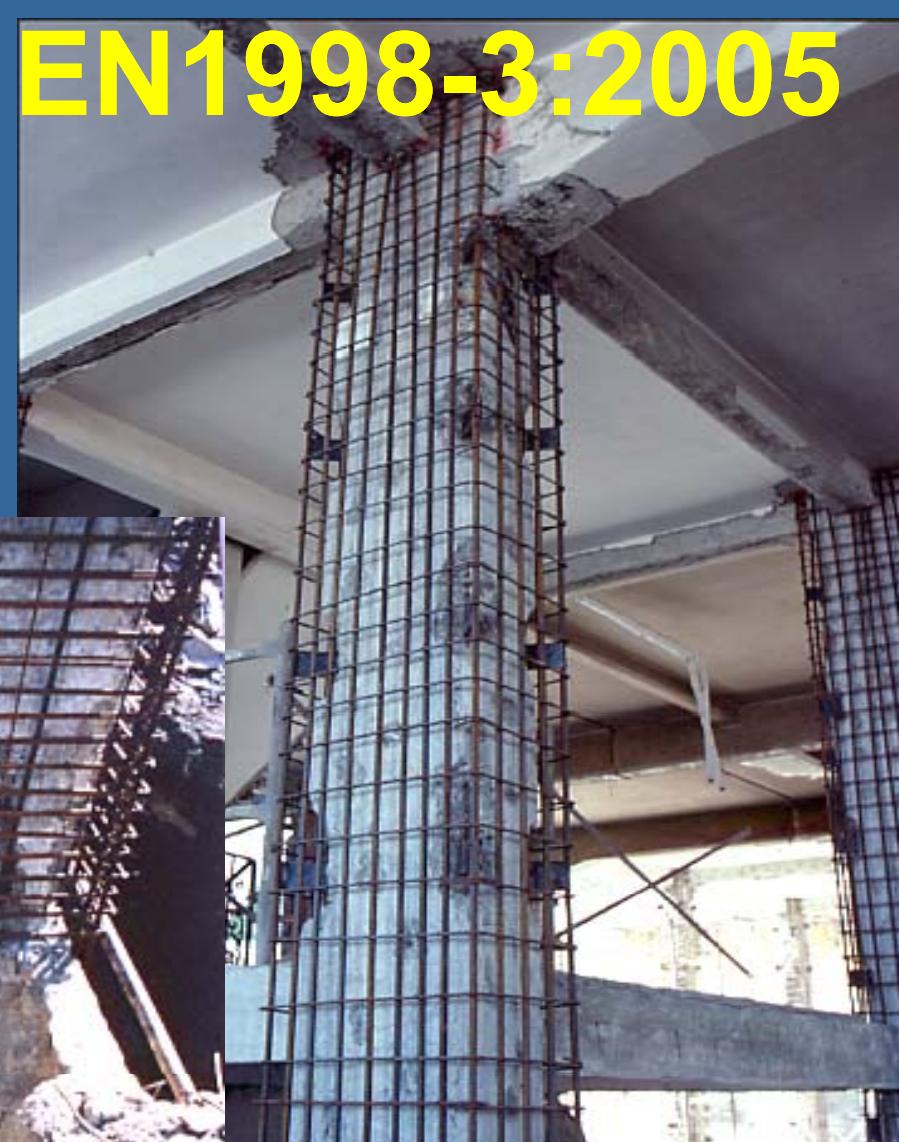
- $V_f = \min(\varepsilon_{u,f} E_{u,f}, f_{u,f}) \rho_f b_w z / 2$: FRP-contribution to cyclic shear resistance:
 - ρ_f : FRP ratio, $\rho_f = 2t_f/b_w$;
 - $f_{u,f}$: FRP tensile strength.
- $V_w = \rho_w b_w z f_{yw}$: contribution of web steel (b_w : web width, z : int. lever arm; ρ_w : steel ratio)
- ρ_{tot} : total longitudinal steel ratio
- h : section depth
- x : depth of compression zone
- $A_c = b_w d$

Test-to-prediction ratio vs μ , no. tests 12
median=0.99, CoV=14.8%:



- In a FRP-retrofitted member the shear resistance as controlled by diagonal tension cannot exceed the shear resistance of old member as controlled by web crushing

RC jackets per EN1998-3:2005



Concrete Jackets (continued/anchored in joint; w/ or w/o lap splices in old member)

Calculation assumptions:

- Full composite action of jacket & old concrete assumed (jacketed member: "monolithic"), even for minimal shear connection at interface (roughened interface, steel dowels epoxied into old concrete: useful but not essential);
- f_c of "monolithic member" = that of the jacket (avoid large differences in old & new f_c)
- Axial load considered to act on full, composite section;
- Longitudinal reinforcement of jacketed column: mainly that of the jacket. Vertical bars of old column considered at actual location between tension & compression bars of composite member (~ "web" longitudinal reinforcement), with its own f_y ;
- Only the transverse reinforcement of the jacket is considered for confinement;
- For shear resistance, the old transverse reinforcement taken into account only in walls, if anchored in the (new) boundary elements
- The detailing & any lap-splicing of jacket reinforcement are taken into account.

Then:

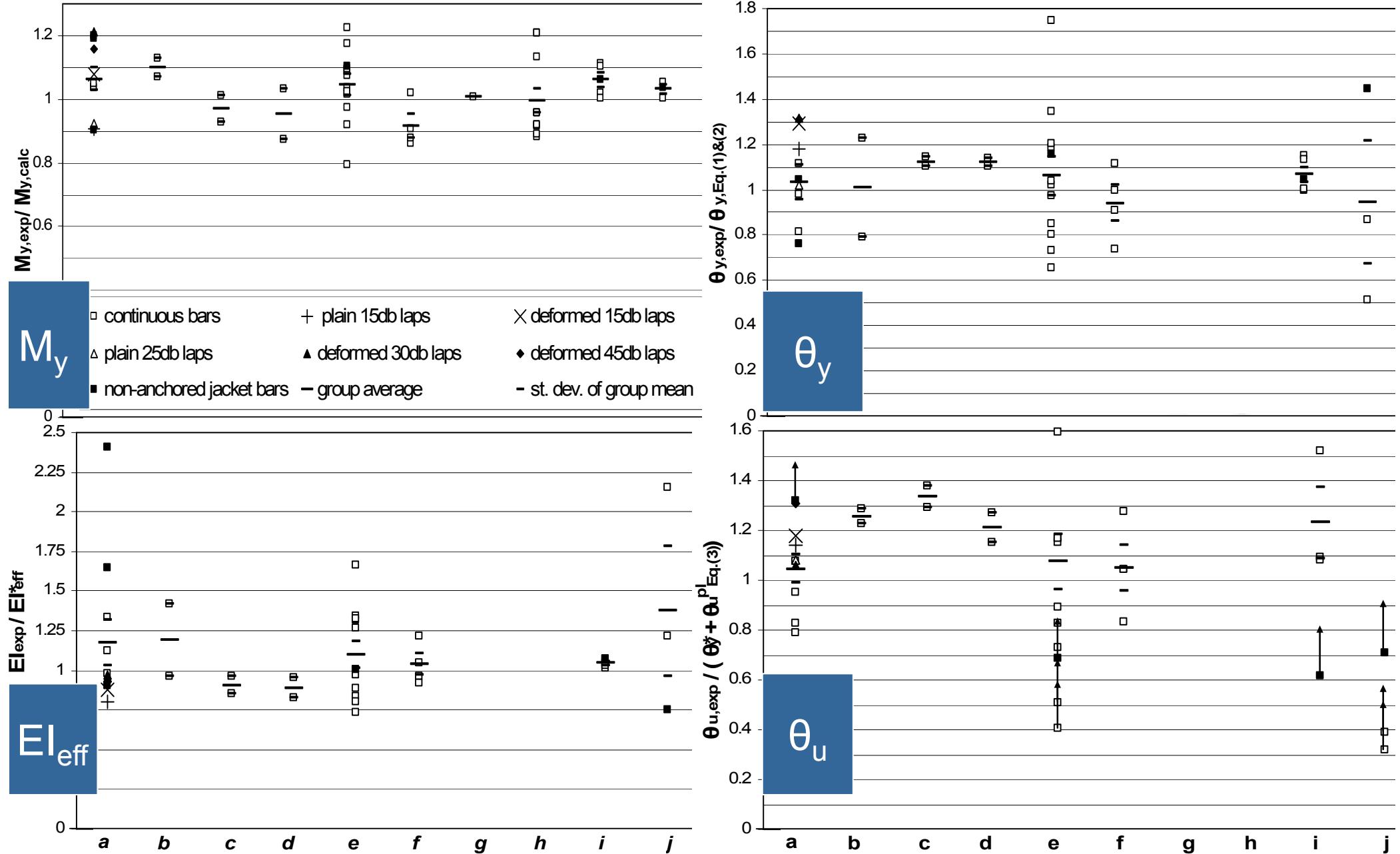
- ✓ M_R & M_y of jacketed member: ~100% of
- ✓ θ_y of jacketed member for pre-yield (elastic) stiffness: ~105% of
- ✓ Shear resistance of jacketed member: ~100% of
- ✓ Flexure-controlled ultimate deformation θ_u : ~100% of

those of "monolithic member" calculated w/ assumptions above.

If jacket bars are not continued/anchored in the joint:

The jacket is considered only to confine fully the old column section.

54 jacketed members w/ or w/o lap splices: test-to-calculated as monolithic



a: no treatment, **b:** no treatment, predamage, **c:** welded U-bars, **d:** dowels, **e:** roughened, **f:** roughened / predamage, **g:** U-bars+ roughened, **h:** U-bars+roughened / predamage, **i:** roughened+dowels, **j:** roughened+dowels / predamage

Steel jackets per EN1998-3:2005



Steel Jackets (not continued/anchored in joint)

Jacket stops before the joint (several mm gap to joint face)

- Flexural resistance, pre-yield (elastic) stiffness & flexure-controlled ultimate deformation of RC member is not enhanced by jacket (flexural deformation capacity ~same as in old member inside jacket, no benefit from confinement);
- 50% of shear resistance of steel jacket, $V_j = A_j f_{yj} h$, can be relied upon for shear resistance of retrofitted member (suppression of shear failure before or after flexural yielding);
- Lap-splice clamping via friction mechanism at jacket-member interface, if jacket extends to ~1.5 times splice length and is bolt-anchored to member at end of splice region & ~1/3 its height from the joint face (anchor bolts at third-point of side)